



Introduction

- Surrey miniaturised spacecraft programme
 - Role of Micro and Nano technologies in this work
 - Role of ultra-small spacecraft in micro-nano-technology development

SNAP-1 mission (2000)
PalmSat development work
ChipSat development work





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Satellite Evolution

- Standard and small satellites:
 - Standard

- 500 kilograms or more;
- Mini-satellites
- Micro-satellites
- 100 to 1,000 kilogram;
- 10 to 100 kilograms;



- Ultra small satellites: •
 - Nano-satellites
 - Pico-satellites

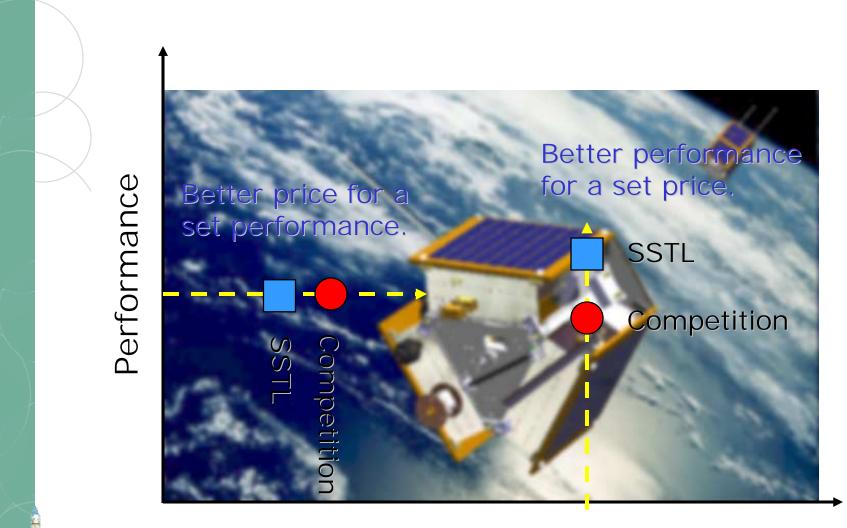


- 1 to 10 kilograms;
- <1 kilogram;
- Femto-satellites <0.1 kilogram.





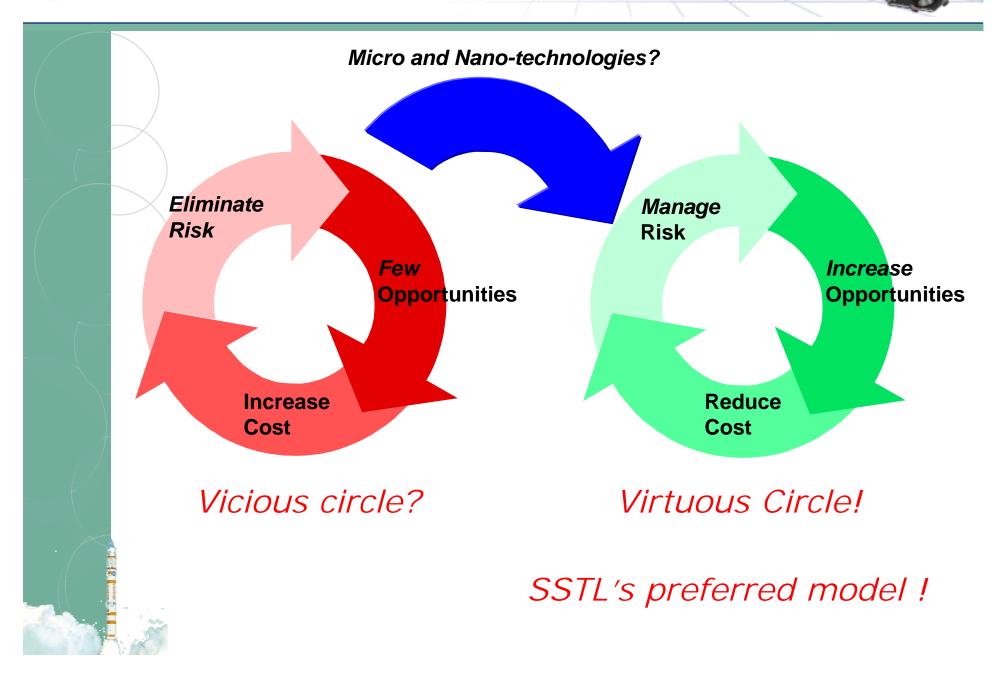
What SSTL aims to offer



M€



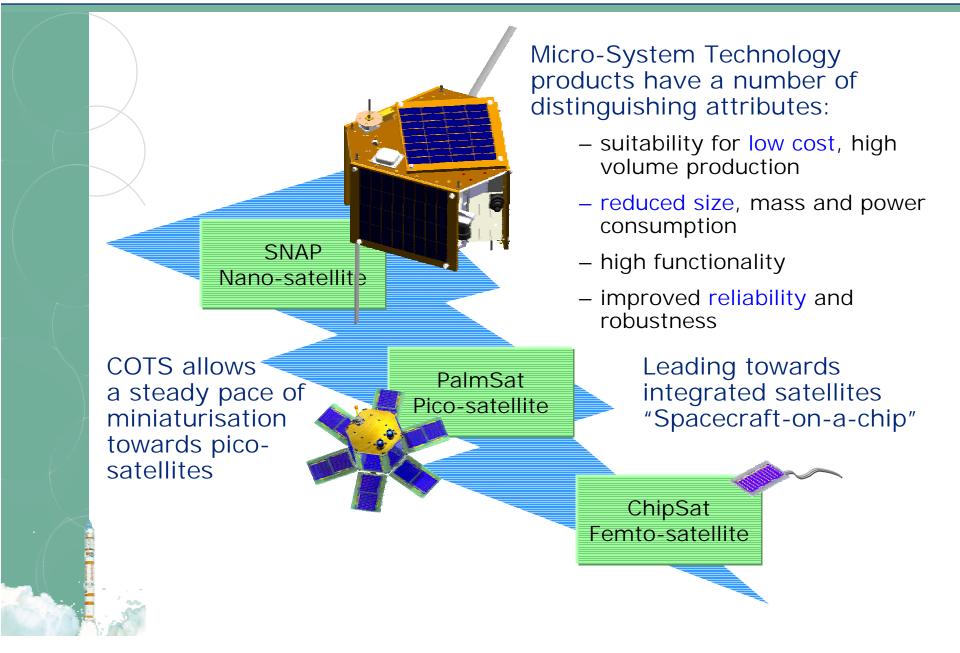


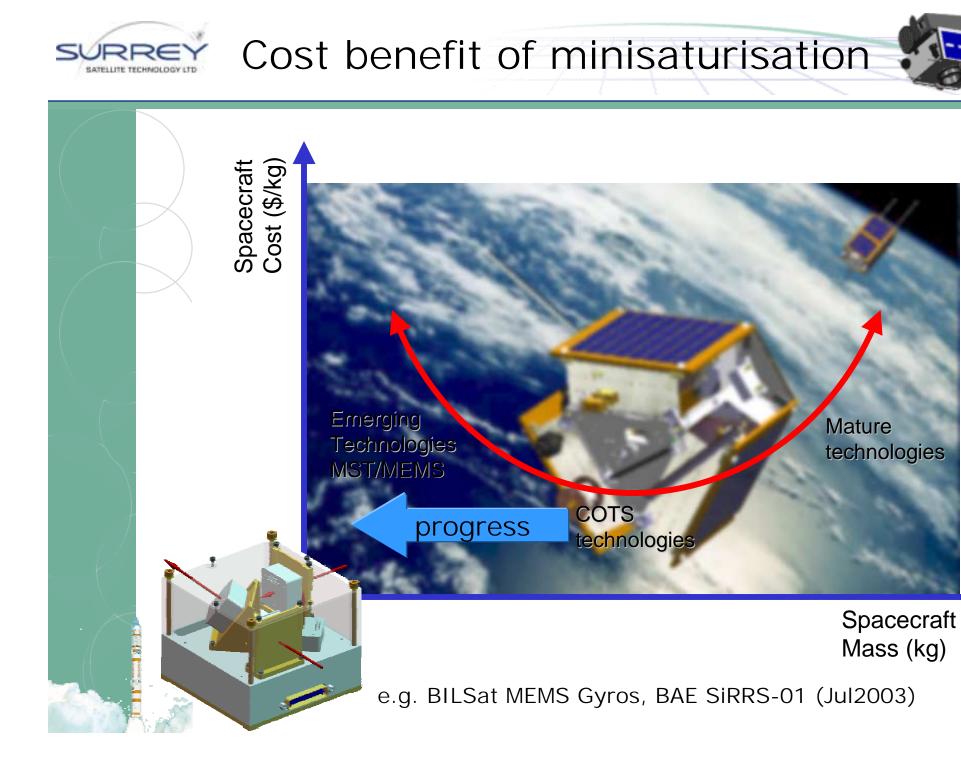




SSTL miniature spacecraft roadmap









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Applications of miniature spacecraft

- At present to serve as a platform for testing miniature technologies and techniques.
 - E.g. Micro Systems Technology validation and space qualification
- Educational applications as cost effective tools to develop an experienced workforce
- Scientific applications agricultural, mineral, and water resources could be mapped and monitored continuously. Space weather applications.
 - Military applications swarms of such satellites, launched together on single, inexpensive rockets, could communicate with sensors on a battlefield and convey important surveillance and tactical information.

Disposable spacecraft Remote inspection



Nano/Pico-Satellites

- SNAP-1 is the UK's first nano-satellite.
 - Designed and constructed by SSC staff and students and SSTL engineers. Funded by SSTL as an R&D Project.
 - SNAP-1 design begun in earnest in October 1999.
 - Spacecraft delivered in May 2000.
 - Launched June 28th 2000 9 months design-to-orbit.



Jerome Salvignol, Project Manager Ed Stevens, AIT Manager



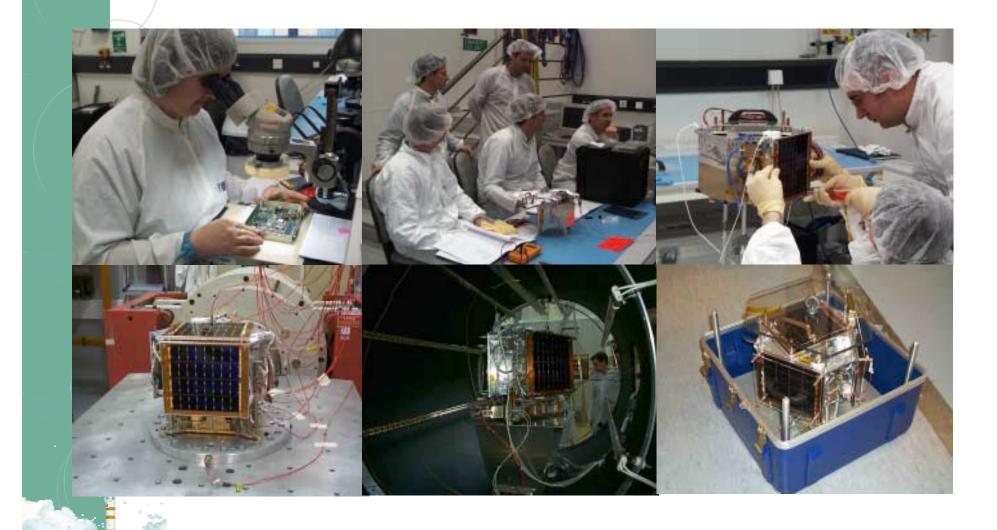
Dr Guy Richardson, SNAP-1's Chief Mechanical Engineer

Dr Craig Underwood, SNAP-1's Chief Architect and Co-Project Manager

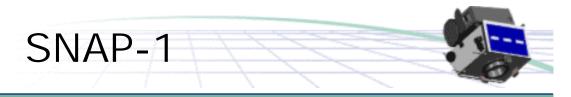




SNAP-1

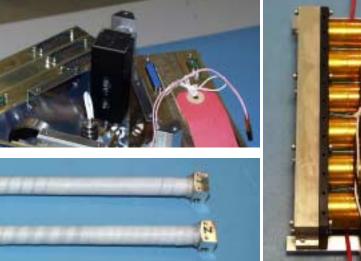


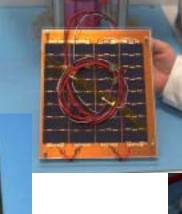




SNAP-1 Advanced Miniaturised Technologies

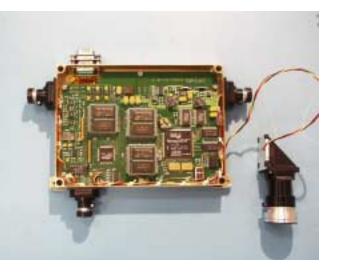














SNAP-1 Power System; VHF U/L Rx.; Spread-Spectrum Comms. Payload; S-Band D/L Tx.; GPS Antenna

SNAP-1





SNAP-1 key features

Structure

- 6.5 kg mass modular, COTS-based design.
- Power
 - 4 GaAs solar panels; advanced NiCd battery.

Data Handling

 3.3V/1.5V 220 MHz SA-1100 32 bit RISC; 4 Mbyte DEDDEC EDAC RAM; FPGA implemented sub-systems.

• RF/TTC

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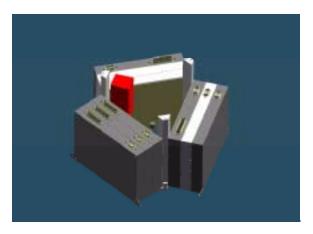
 VHF U/L at 9600 bps FSK; S-Band D/L at 38.4 kbps BPSK or 76.8 kbps QPSK; internal CAN bus.

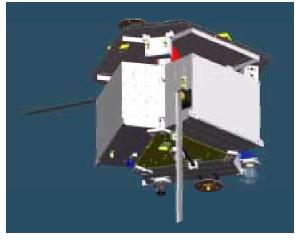
ADCS/GNC

- 3-axis stabilisation with orbit control via a modified 12 channel "ORION" GPS receiver, a compact 3-axis fluxgate magnetometer, 3 miniature magnetorquer rods and a single miniature pitch-axis momentum wheel.
- Butane Cold-Gas Thruster (50 mN; ~3.5 m/s DV)
- UHF Inter-Satellite Link Receiver Payload
- VHF Spread Spectrum Communication Payload

Payloads

 Machine Vision System Payload - four "5th Generation" CMOS video cameras for remote inspection and Earth imaging.

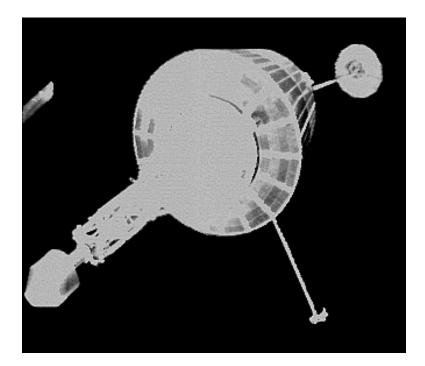


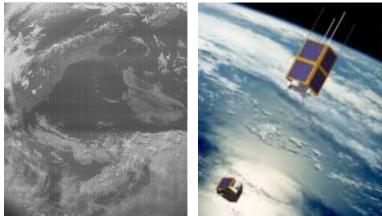




SNAP-1 achievements

- Rapid mission: 9 months from design to orbit;
- Low mission cost: < €1.6m.
- First nanosatellite to image other space vehicles – remote inspection capability confirmed.
- First nanosatellite to demonstrate three-axis attitude control;
- First nanosatellite with onboard propulsion to demonstrate orbital control and manoeuvring;
- SNAP's suitability for agile nanosatellite missions is confirmed.







Recent work

- Platform and technologies
 - ADCS
 - Inter-satellite link work
 - Fuel tank upgrade



Fuel tank



SNAP upgrade



PalmSat (preliminary specs)

- <1 kg mass integrated, COTS-based design.
- Power 18 GaAs/Ge dual-junction solar cells; advanced 4 cell NiCd battery; ~2W power;
- OBC/Bus Controller 3.3V 20 MHz PIC Micro-Controller; EEPROM memory.
- RF/TTC/ISL Low power 2.4 GHz SS ISL; Amateur Band(s) U/L & D/L;
- ADCS/GNC Passive magnetic.
- CMOS Camera Payload One VGA CMOS video camera for remote inspection and Earth imaging.
- Options Advanced Triple Junction Solar Cells; GPS Positioning; Active Magnetic ACS; MEMS gyro INU; Optical Attitude Determination; MEMS Micro-Thruster.

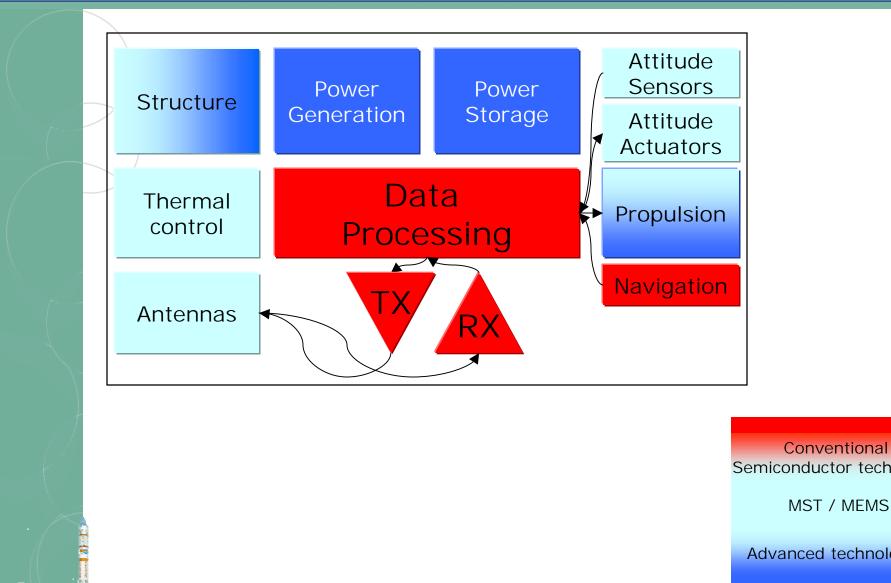




- Most satellites are constructed from physically separate subsystems, each of which is composed of some combination of circuit boards and components.
- Through the use of semiconductor fabrication techniques, all the components required for a state-of-the-art satellite can be implemented on silicon chips.
- Space engineers can use MST / MEMS to make femto-satellites in the form of:
 - layered silicon wafers containing sensors, computing power, communications systems, mechanical devices and even micro rocket thrusters.
- Unique properties enable silicon to become most of the mass of a satellite it can simultaneously function as structure, heat transfer system, radiation shield, optic, and semiconductor substrate
 - A long-term prospect is an entirely integrated satellite that contains all the subsystems of a fully functional satellite condensed into one module.



Spacecraft elements



Semiconductor technology

MST / MEMS

Advanced technologies



Enabling Technologies

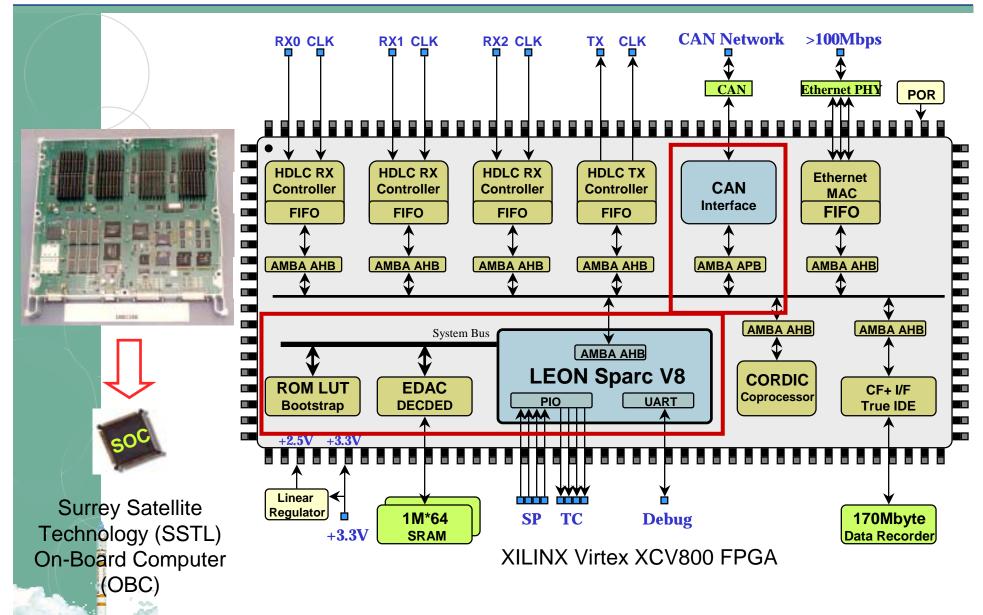
Technologies

- Microelectronics System-on-a-Chip + low energy consumption, evolvable hardware
- MEMS extremely small sensors, optical communication components, power supplies
- Accurate Position Sensing and Actuation
- Advanced Imaging Sensors
- Formation Flying Techniques
- Advanced Computers, Reconfigurable Computing, Remote Programmability
- Distributed System Design Methodologies
- Cluster Management Algorithms
- Artificial Intelligence
- Challenges
 - Independent tracking
 - End-of-Life Disposal
 - Drag!



Satellite-on-a-chip research

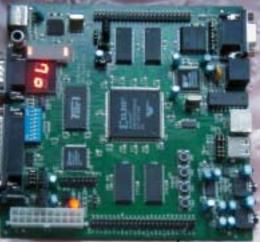






Experimental set-up

- Processor LEON 2-1.0.2a VHDL IP core (SPARC V8) (ESA)
 - Working Frequency 25 MHz
 - UART Baud Rate 38,400 bps
 - Internal S-Record Boot Loader
- On-Board Network (Node 1) HurriCANe VHDL IP core (ESA)
 - Baud Rate: 312,500 bps
- EDAC EDAC VHDL IP core (SSTL)
 - Double-bit correcting Quasi-Cyclic (16,8) shortened EDAC code
- Prototyping Board XESS XSV800 (Xilinx Virtex XCV800 FPGA)
 - Up to 100 MHz programmable oscillator
 - 16M Bits SRAM (two banks 512K x 16)
 - 16M Bits flash RAM
- On-Board Application Program S-Record
 - 160K Bytes (CCSDS_SC Software Package)





Conclusions

- Emerging micro and nano-technologies will revolutionise some space missions
 - Through technology infusion into existing systems
 - Through spacecraft miniaturisation
- Small, inexpensive satellites, can play a key role in validating complex micro and nano-technologies rapidly for use in space.
 - E.g. SNAP spacecraft can carry 1kg/10W-pk payload
 - E.g. PalmSat can carry 100g payload

Spacecraft-on-a-chip mi

Thank-You!