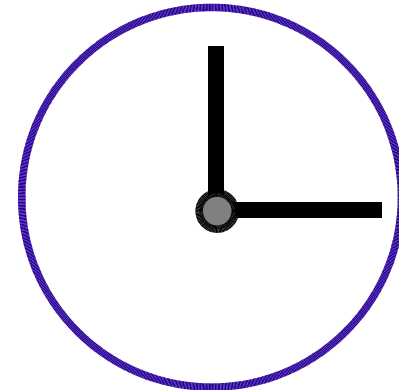
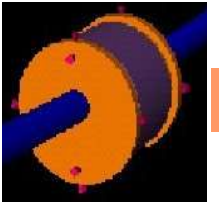
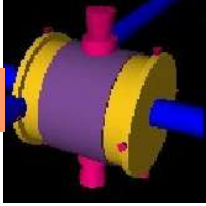


H – 15 min

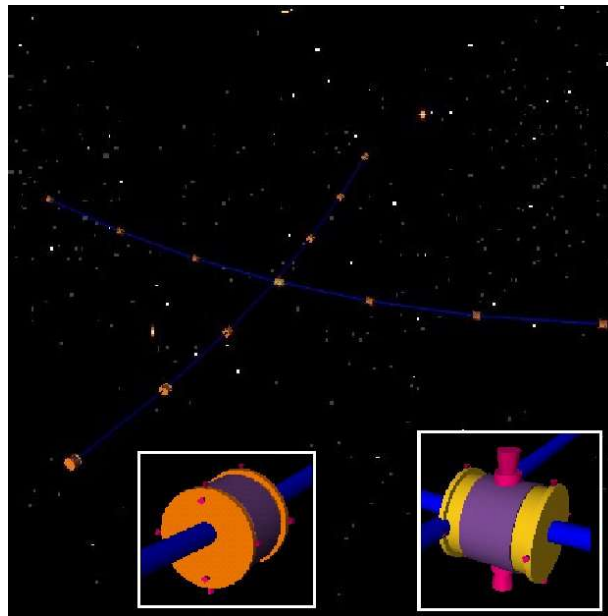


THIS PRESENTATION AIMS TO BE SHORT AND SWEET!

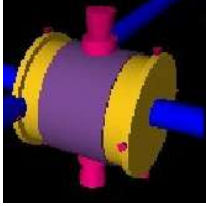


MICROCELLS for METASTRUCTURES

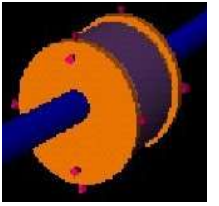
An long-term application of MNT to distributed space structures



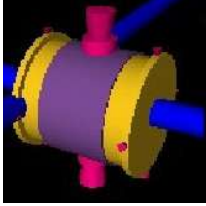
Arnaud Lecuyot (SSTL), S.E. Hobbs (Cranfield CoA), I. Honstvet (EADS Astrium)



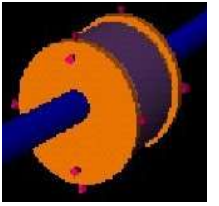
Warning & Acknowledgments



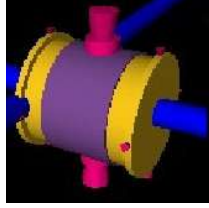
- *Acknowledgements*
 - *EADS Astrium*
 - *EPRSC*
- *Warning*
 - *Doctoral work*
 - *Long-term, exploratory*
 - *Interest in process as much as product*



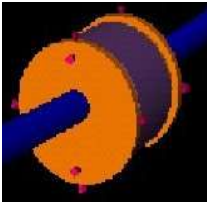
OUTLINE



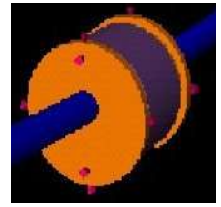
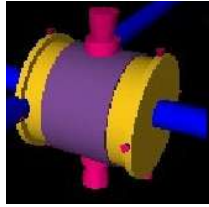
- Introduction: Context & Concept rationale
- Study Logic: Research flow, Methods & Tools
- Analysis: Mission & System analysis
- Analysis: System and subsystem design
- Discussion: Metastructures in context
- Conclusion



Introduction: MNT System Eng.



- Long-term (20 years) Systems Engineering to:
 - Fully realise the potential of MNT in space
 - Provide a long-term framework to space planners
 - Identify Technological points and requirements
- Systems Engineering Logic Applied
 - From Mission analysis to Requirement Flowdown
 - To include programmatics & “in context” analysis
 - Focus on distributed systems & structures



Introduction: Concept Rationale

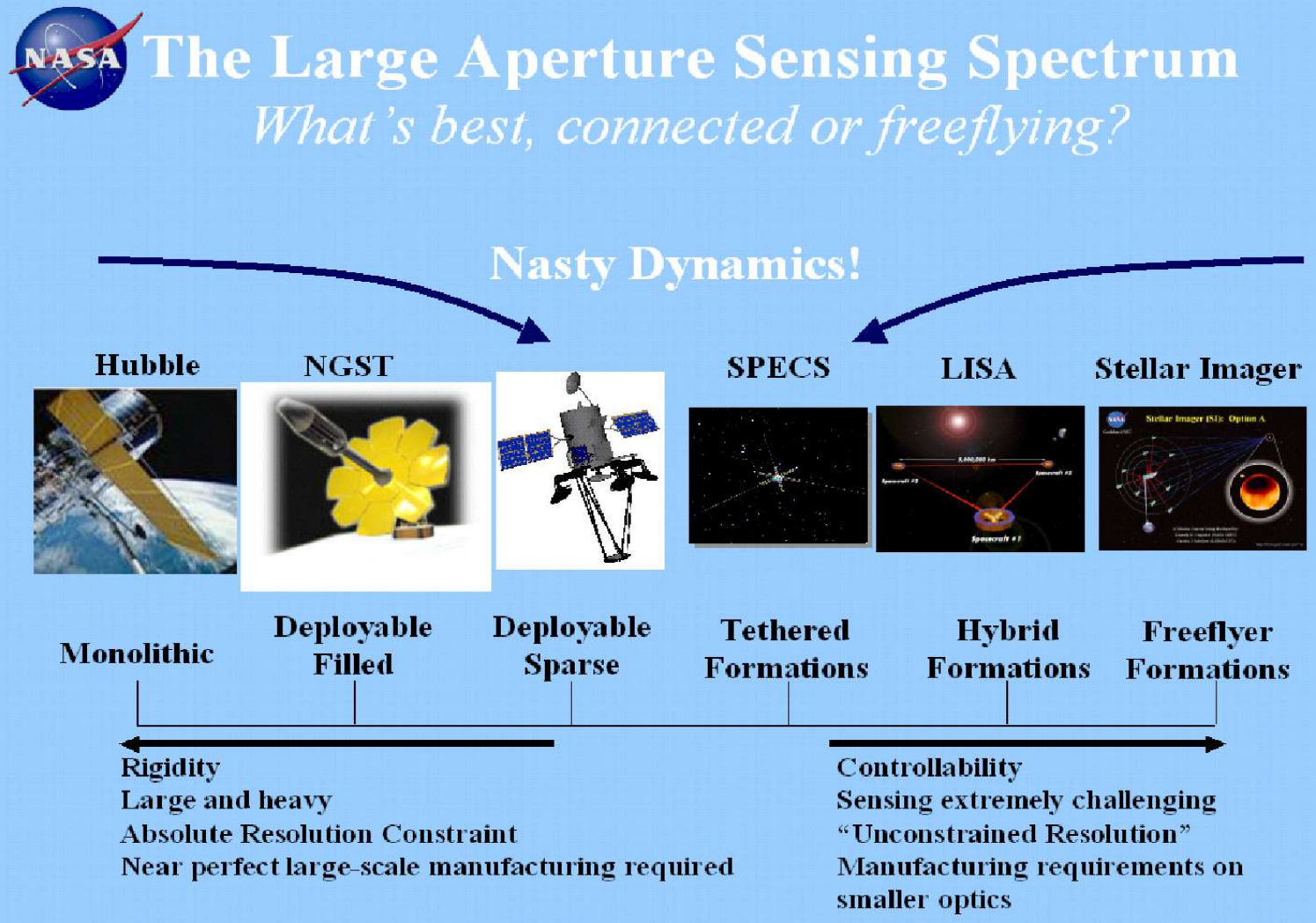
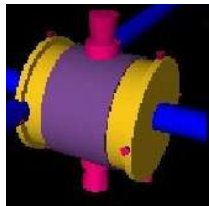
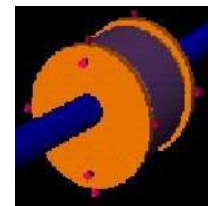


Image source NASA GSFC

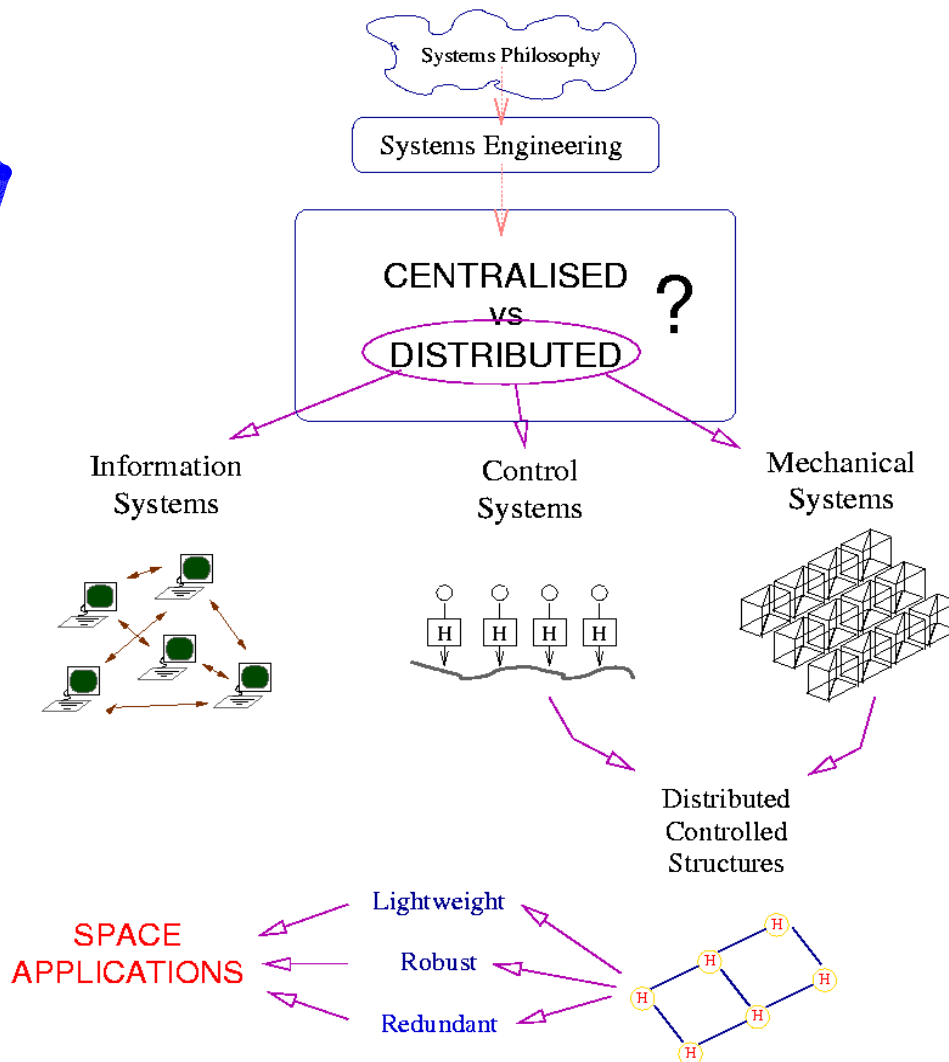
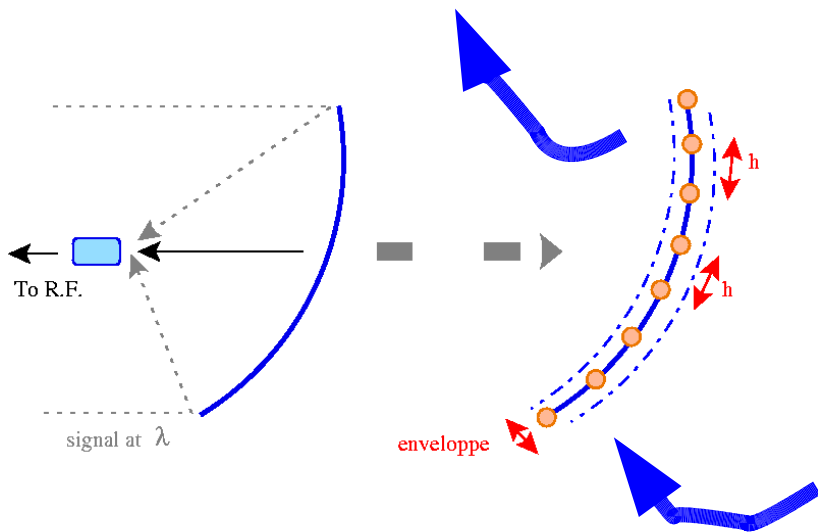


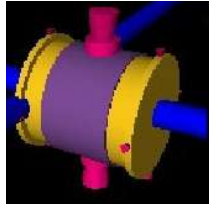
Introduction: MNT Attributes



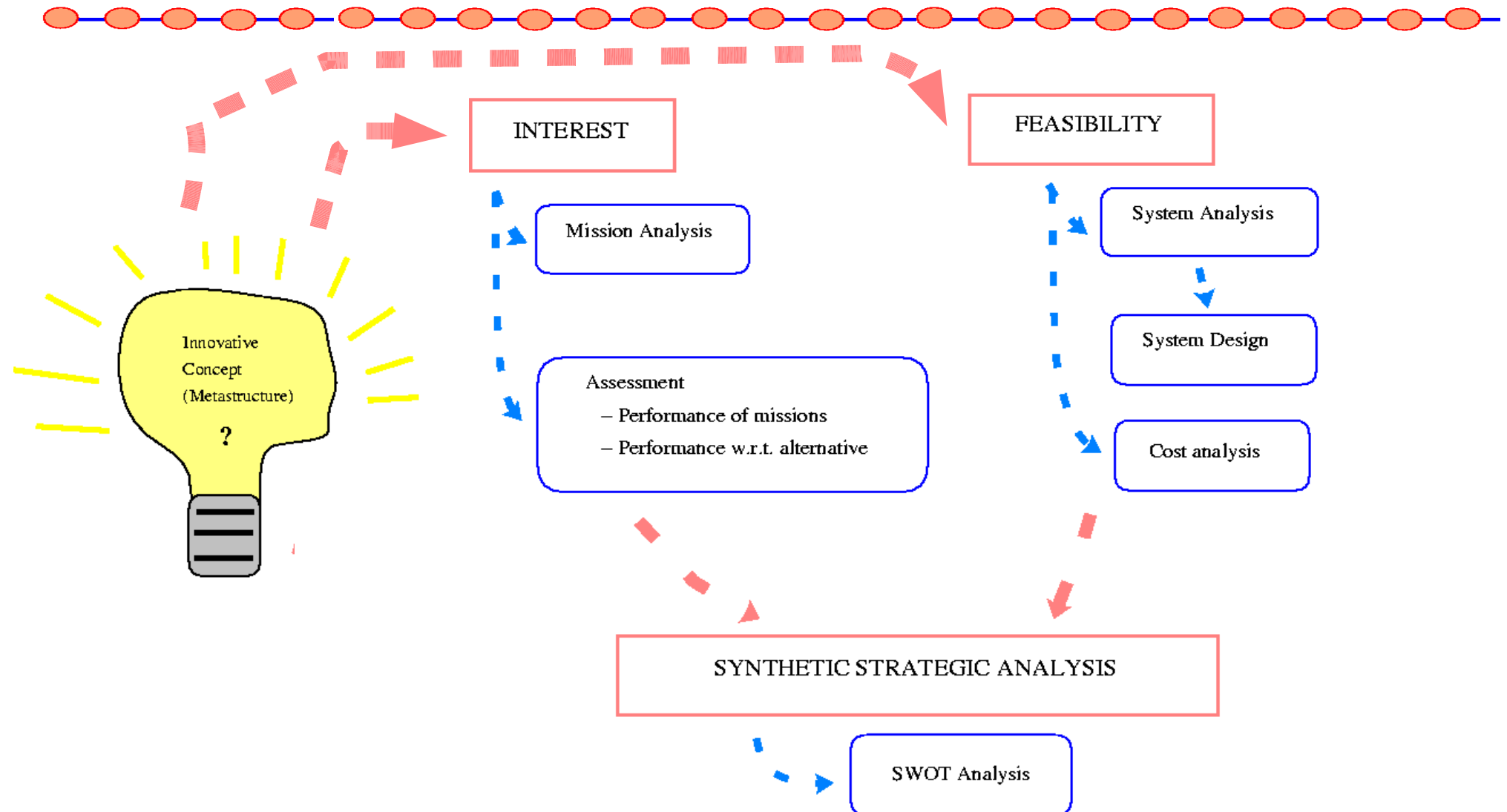
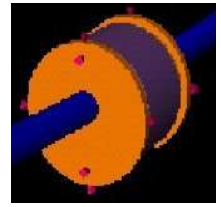
- System-level Distribution
- Subsystem-level Integration
- Multi-Domain Concurrence

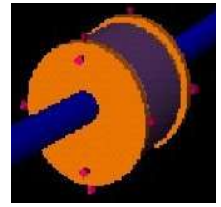
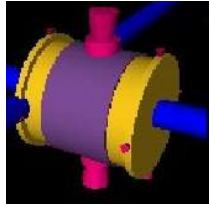
Node Mass	Link Section	System Mass
0.5	8	600
1	10	1300
5	15	6500



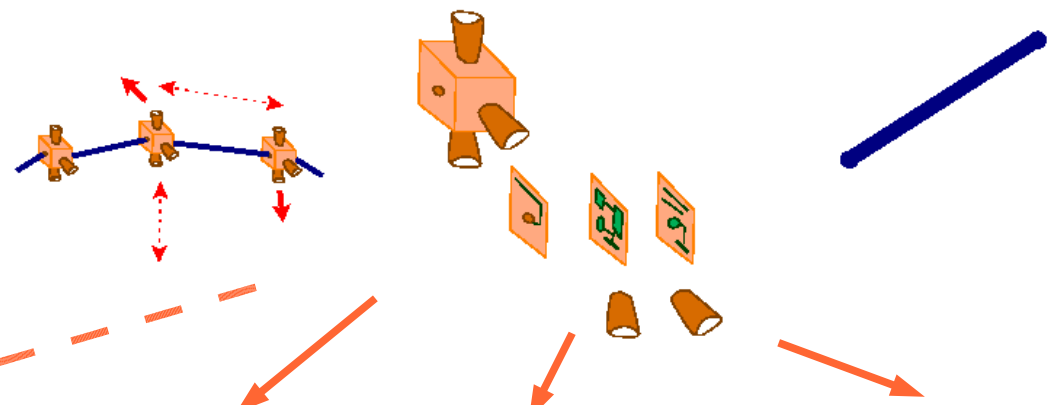
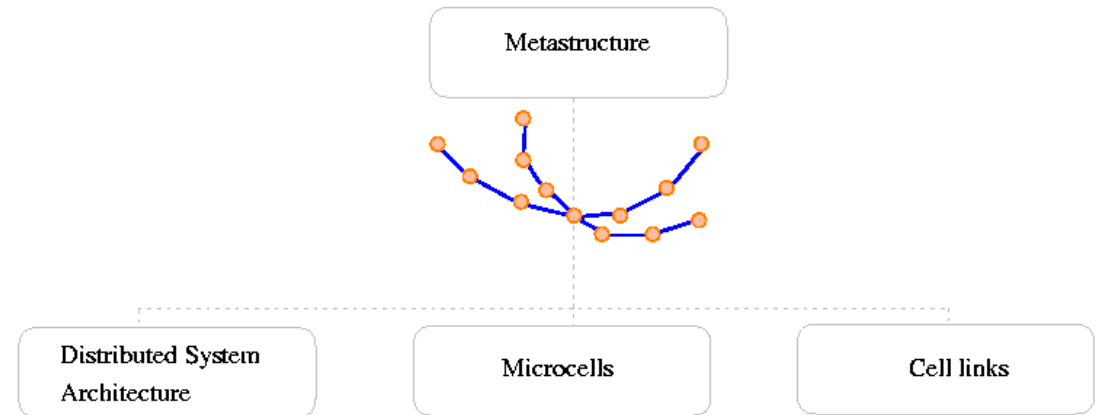
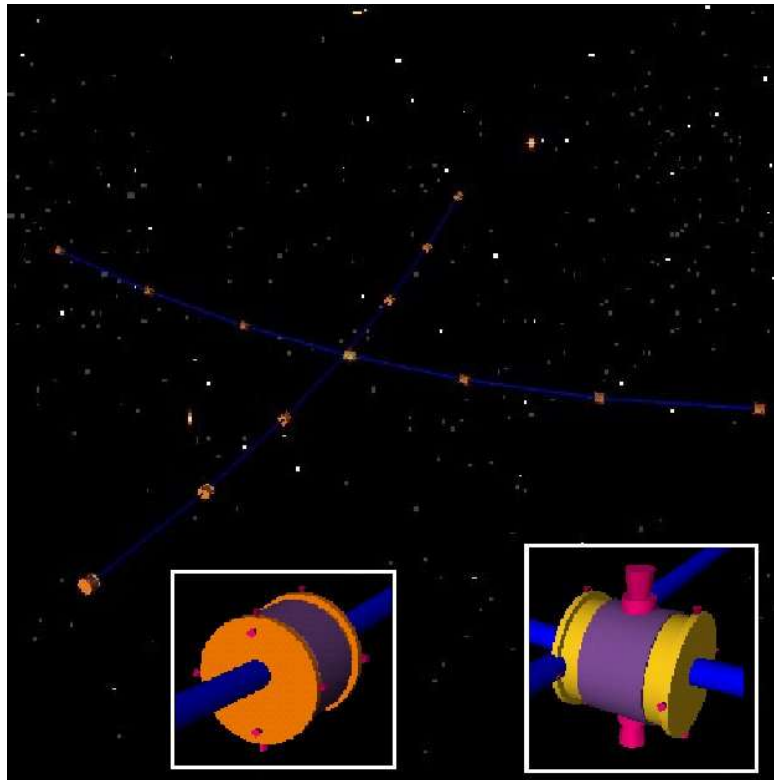


Study Logic – Methods & Tools





Mission & System Overview



Reference Design (Test)

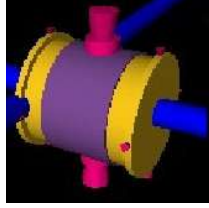
Communications

EO& Science

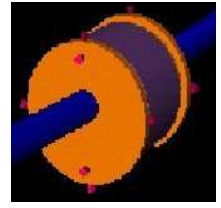
Infrastructure

Interferometer (cm)

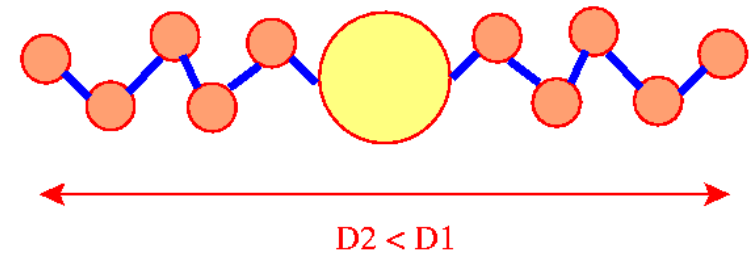
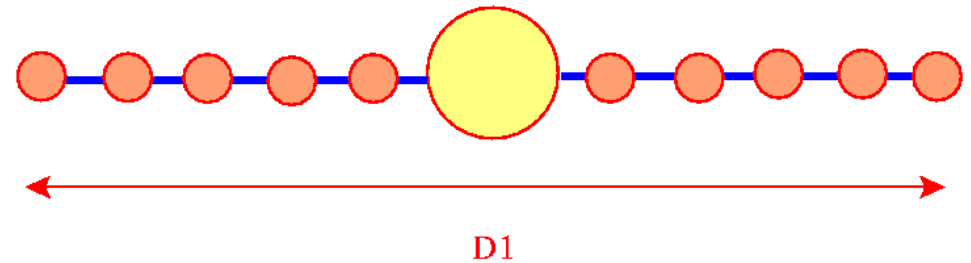
Solar Concentrator

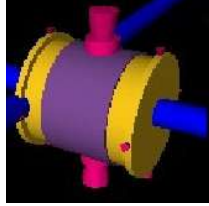


Mission – Cm Interferometer

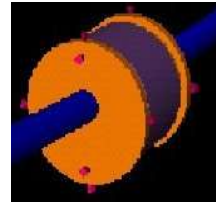


- For long λ RadioAstro
- In L1/L2 (& Earth)
- 32 x Mills-Cross array
- 5 cm accuracy
- 1 to 10 km Baseline D
- “Accordion” configuration

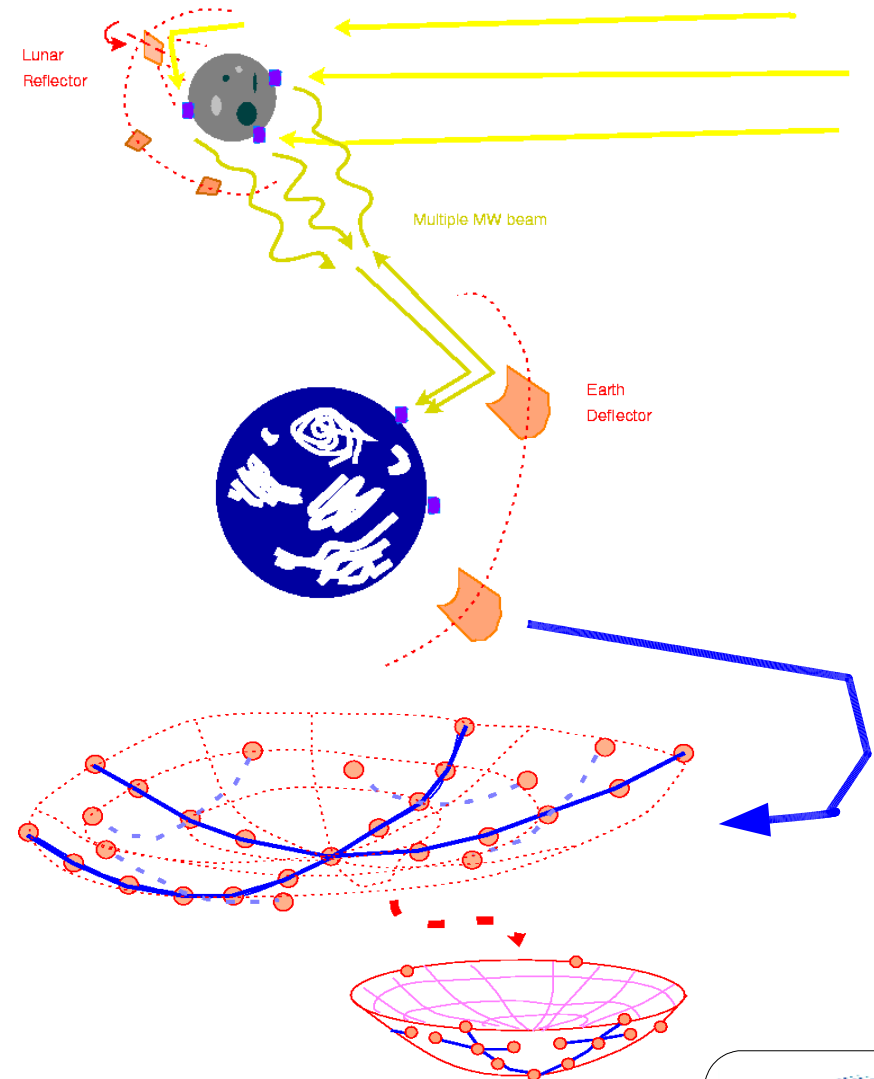


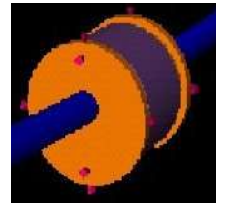
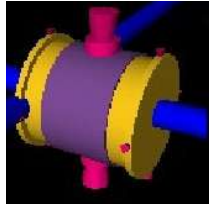


Mission – Solar Concentrator



- For Earth/Moon Lunar Power System
- In ~6000 km Orbit
- 1 km x 1 km
- Parabolic Configuration
- 1 arcmin accuracy
- Lifetime 5 years



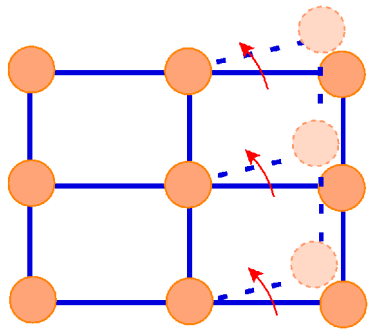


System Trade-offs

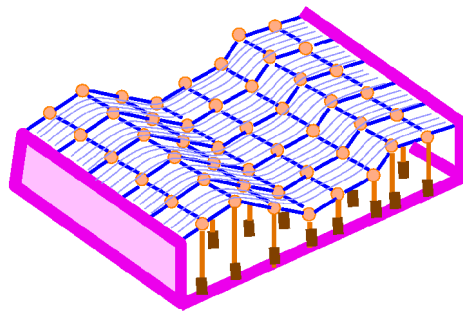


- Link Configuration

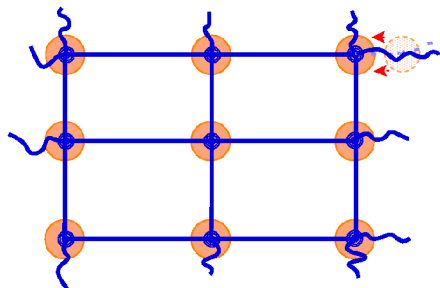
- Microcell patterns



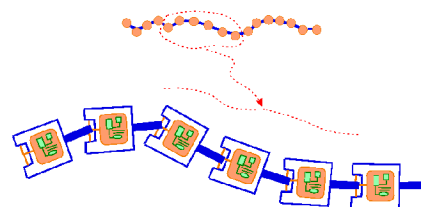
Rigid Net



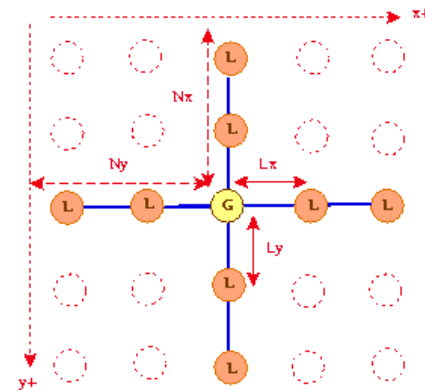
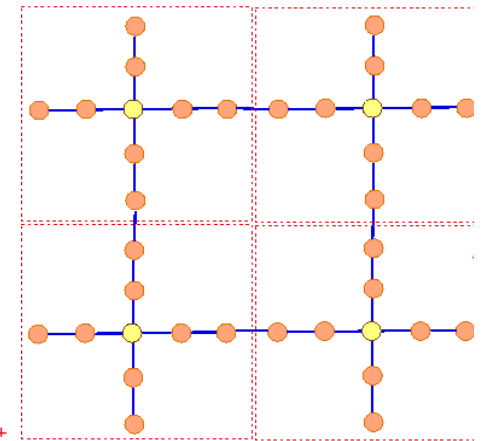
"Tent"

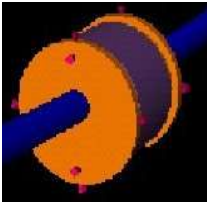
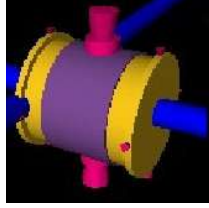


"Winch Net"



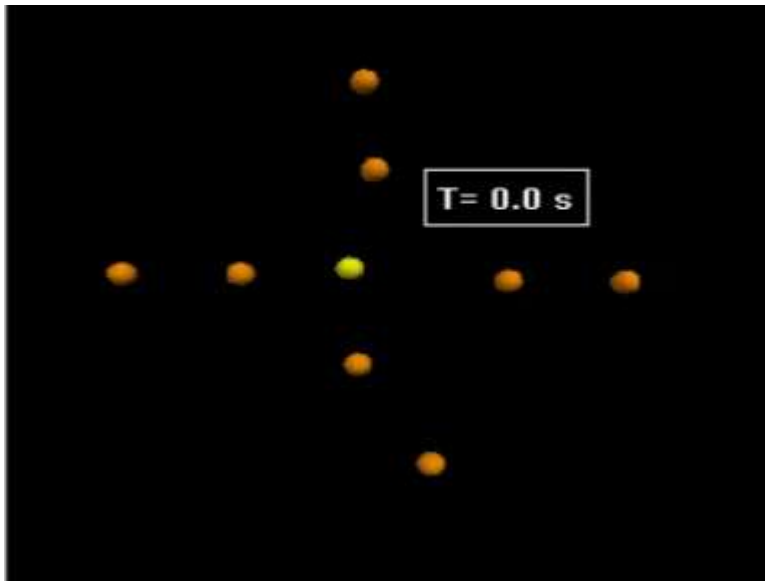
"Snake"



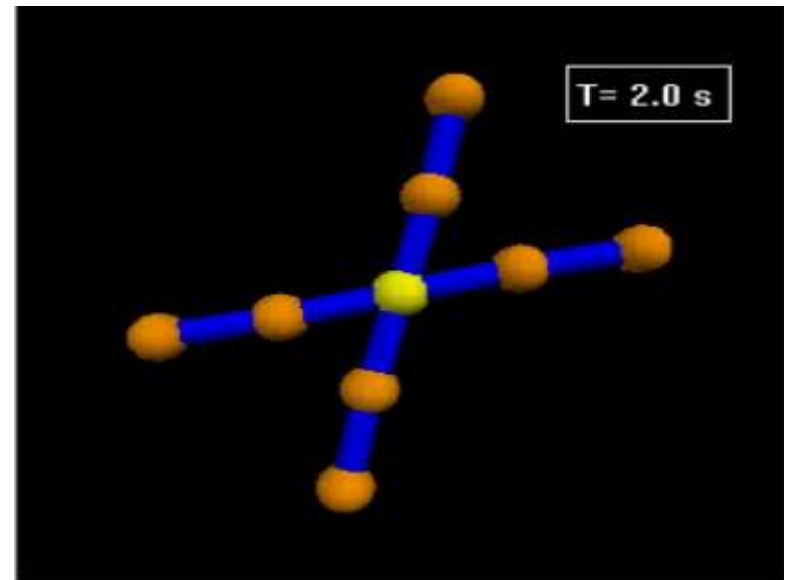


System Design - Simulations

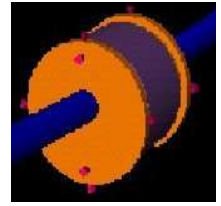
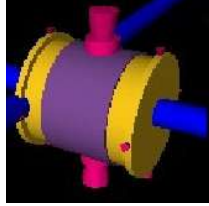
Reference metastructure



Unlinked structure (1000 km)



Basic controlled single pattern

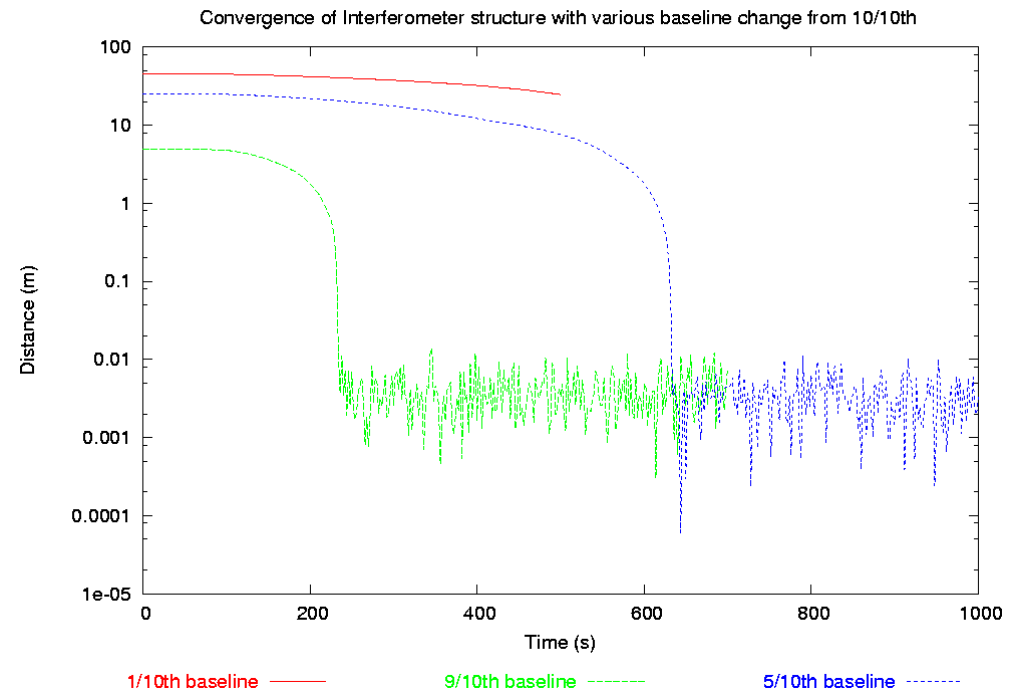


System Design - Simulations

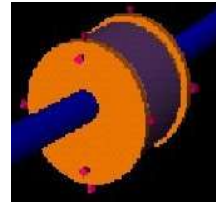
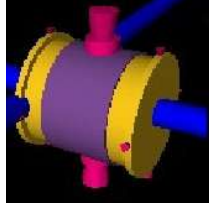
Interferometer



L1 Simulation

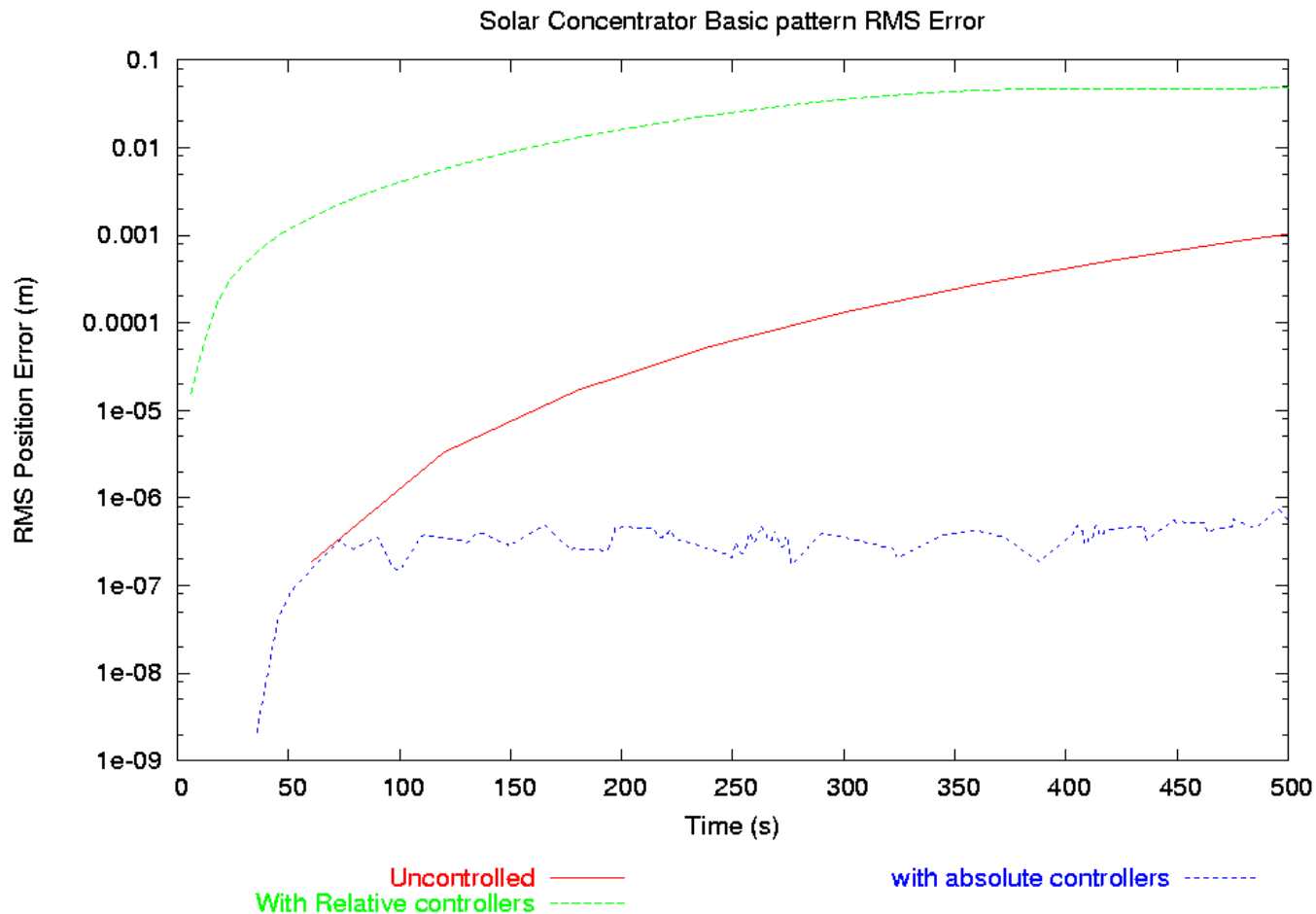


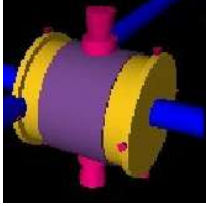
Baseline Changes



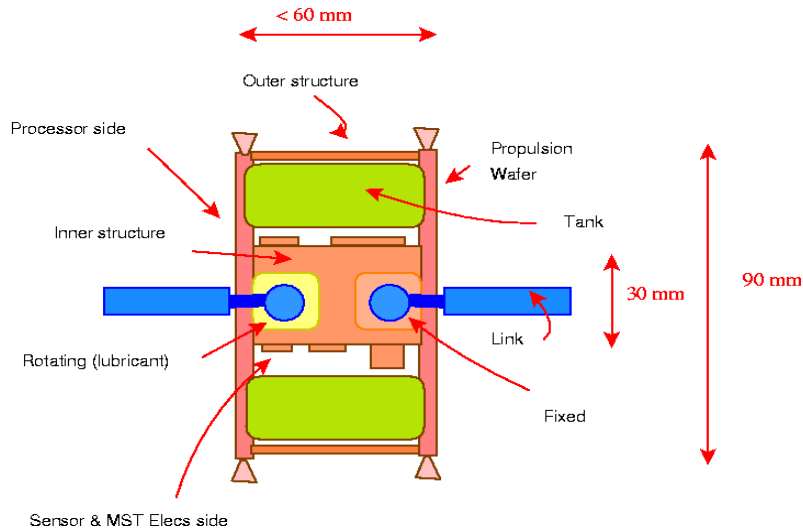
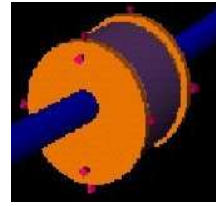
System Design - Simulations

Solar Concentrator

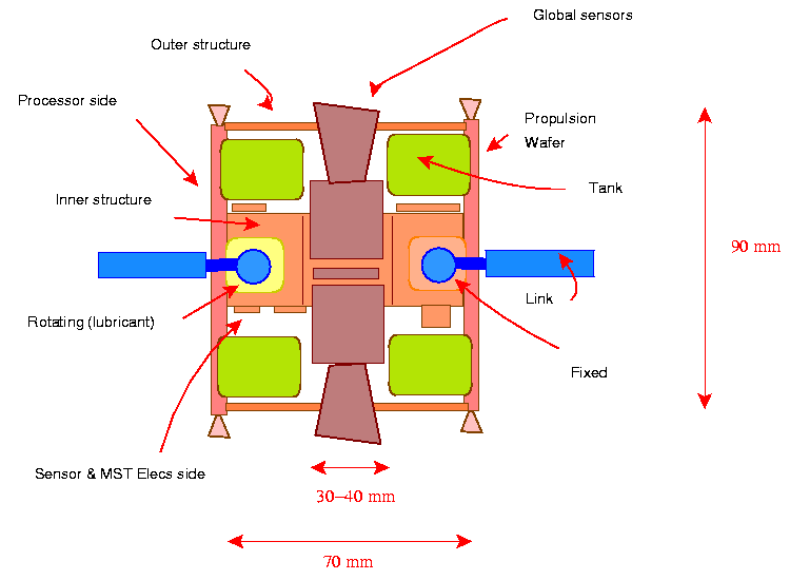




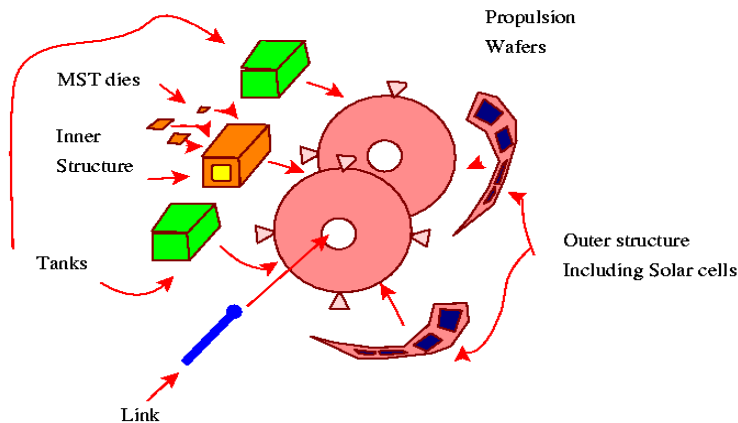
System Design – Cell layout

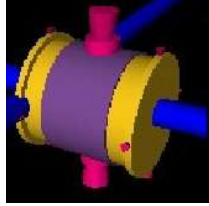


Local Cell

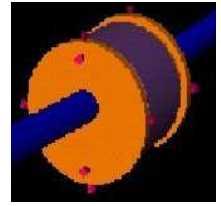


Global Cell

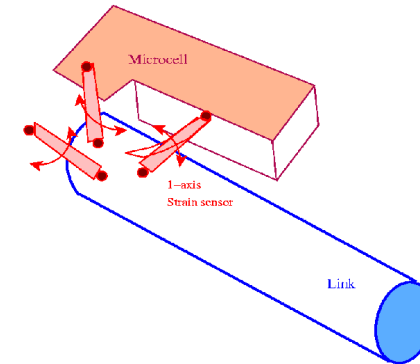




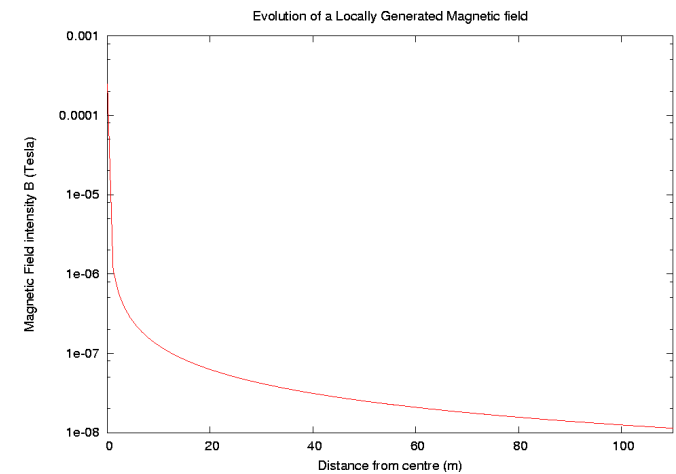
System Design – Cell systems



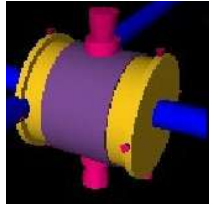
- Mass 340/380 g (L/G)
- Power 800/1100 mW
- Simple sensor Tech
- Micropropulsion (350 Ns)
- DH in Monolithic Wafers
- Similar to simpler “WaferSat”



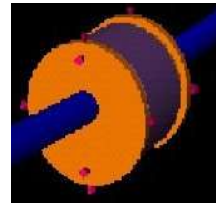
Link inclination strain sensing



Position Local Magnetic sensing



Discussion – Baseline & Issues



- Baseline & costing

- Interferometer

- L1 perf 0.02 m
- 100 kg, 80 MEuro
- Long links (>10 m)

- Solar Concentrator

- Earth perf 0.014 m
- 2 tons, 170 MEuro
- Short lifetime

- NRE 340 MEuro

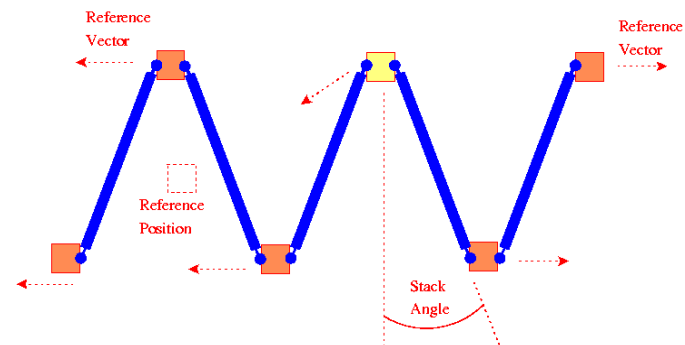
- Issues

- Lifetime of cells

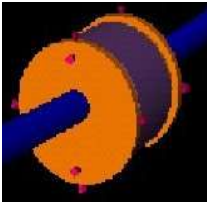
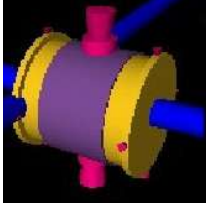
- Better Controller
- Lock-up links

- Launch & Stowing

- Problem TBD



Discussion - Alternatives



- Alternatives

- Deployables
- Inflatables/Active
- Formation Flying

- SWOT Analysis

- Strengths
 - Robust Arch, shape change
- Weaknesses
 - Dynamics, Low Maturity
- Opportunities
 - Applications, MNT use
- Threats
 - FF perf, High NRE cost

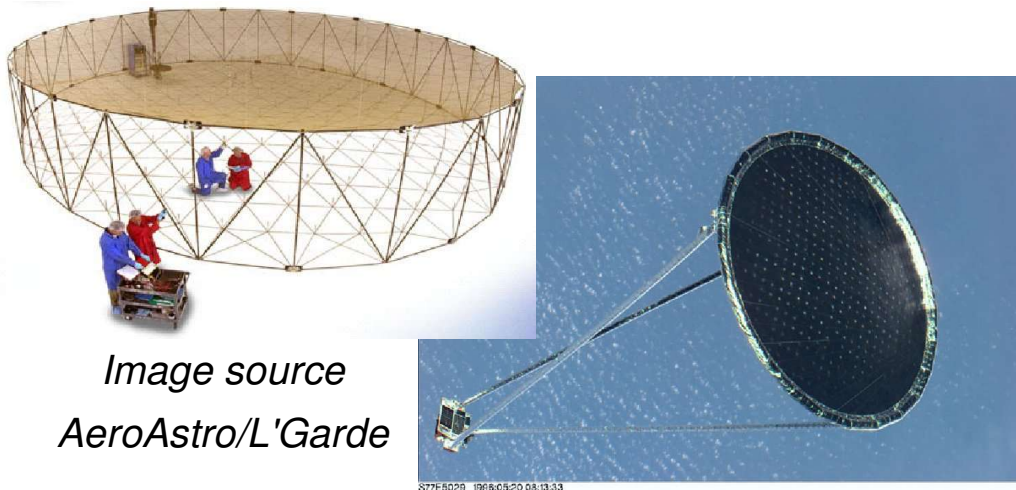
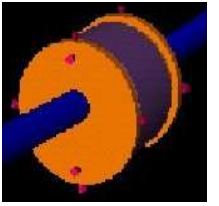
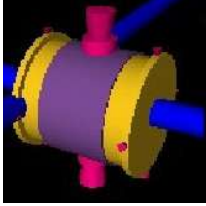


Image source
AeroAstro/L'Garde



Conclusions

- Concept on its own has potential interest
 - In special niche applications (with better control)

- As a concept to meet the large structure needs

- Exercise is useful for astronautics MNT R&D

Party Time!

- MNT maximum return dependant on:

- Distributed Systems Theory & control, Smart Structures
- Ultra-High Integration but not necessarily functionalities

- Comparison to Alternatives is essential

- Start with mission, applications, Reqrut analysis

