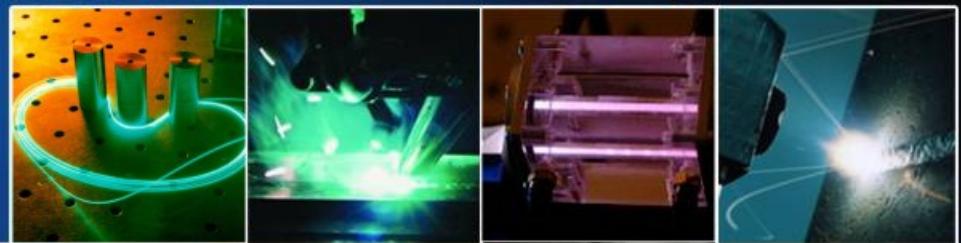


**PEOPLE MATTER**

*research  
development  
consulting*



LASER ZENTRUM HANNOVER e.V.

***Passively Q-Switched Nd:YAG Laser for  
Spaceborne Laser Altimetry –  
Bepi Colombo (Mercury) Altimeter***

**J. Neumann, S. Hahn, R. Huß,  
R. Wilhelm, M. Frede, D. Kracht**

# *Overview*

- BELA laser requirements & conceptual laser design
- Testbed setup for feasibility study
  - Laser performance
  - Temperature dependencies
- Optical & mechanical design of prototype models (PM)



LASER ZENTRUM HANNOVER e.V.

# *BELA - BepiColombo Laser Altimeter*

Solid-State Laser

Max Planck Institute for Solar System Research (D)

Receiver Optics/Detector

University of Bern (CH)

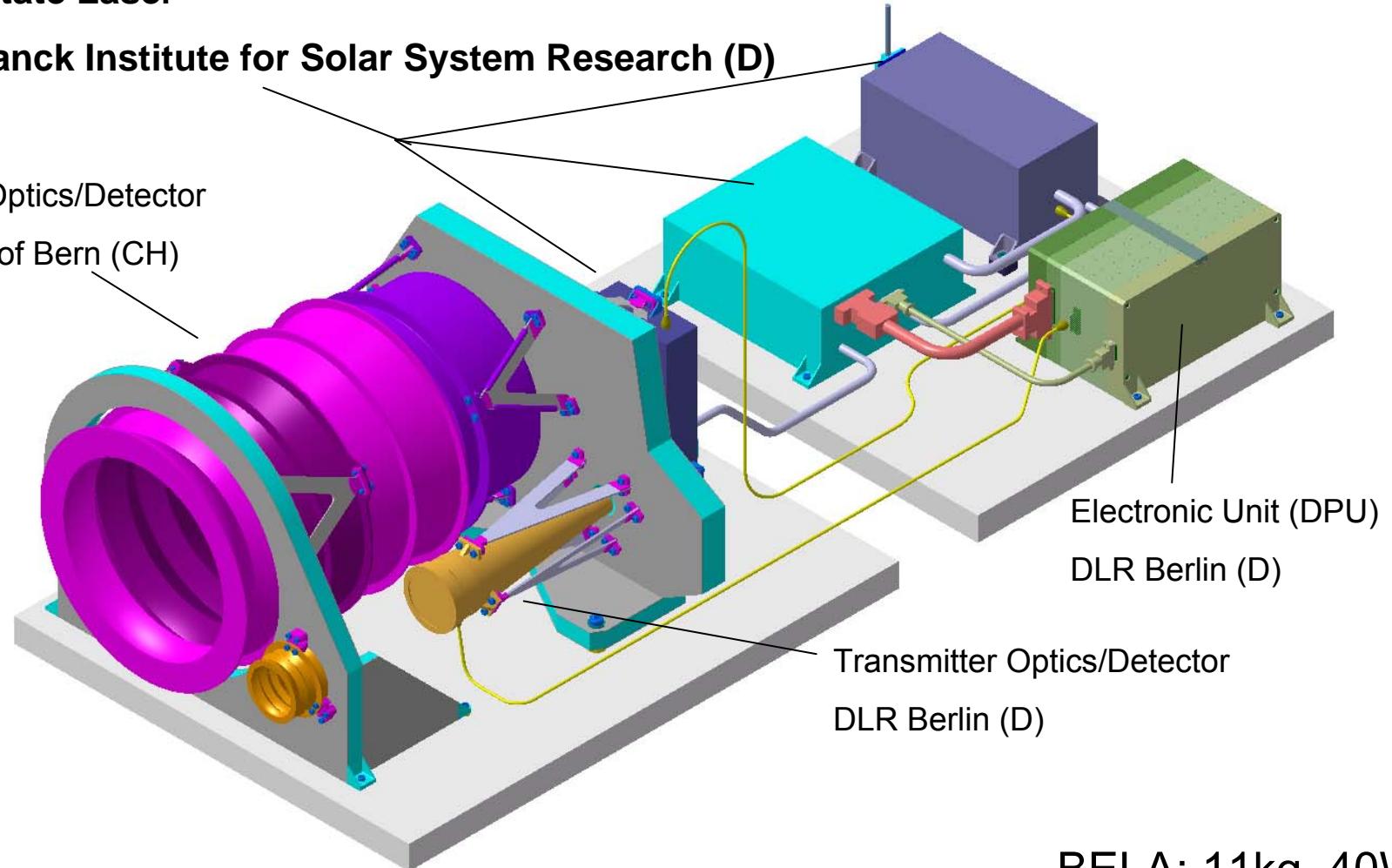
Electronic Unit (DPU)

DLR Berlin (D)

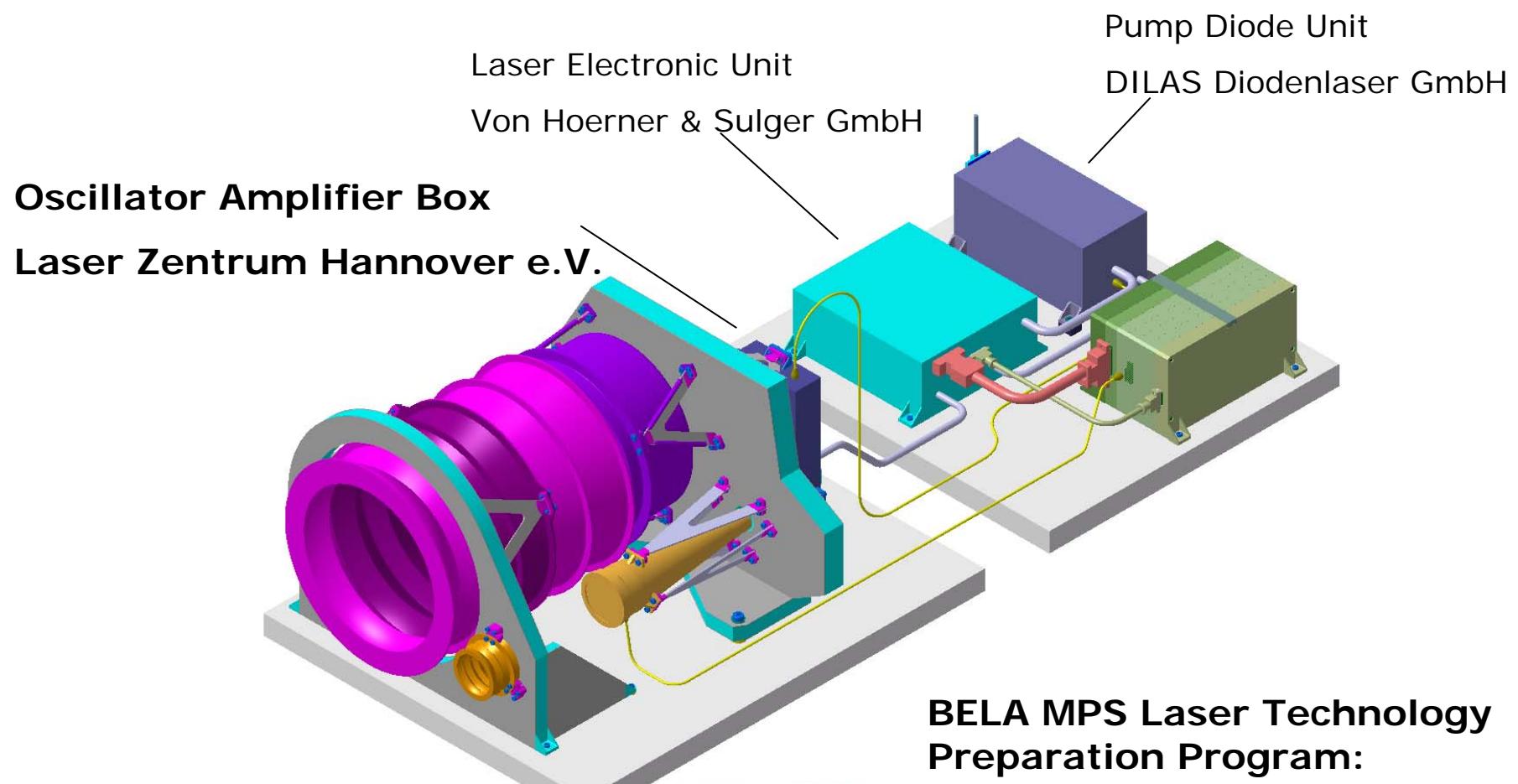
Transmitter Optics/Detector

DLR Berlin (D)

BELA: 11kg, 40W



# *BELA Diode Pumped Solid-State Laser*



**vH&S**

**LZH**

**EADS**

**DILAS**  
Diodenlaser GmbH

for

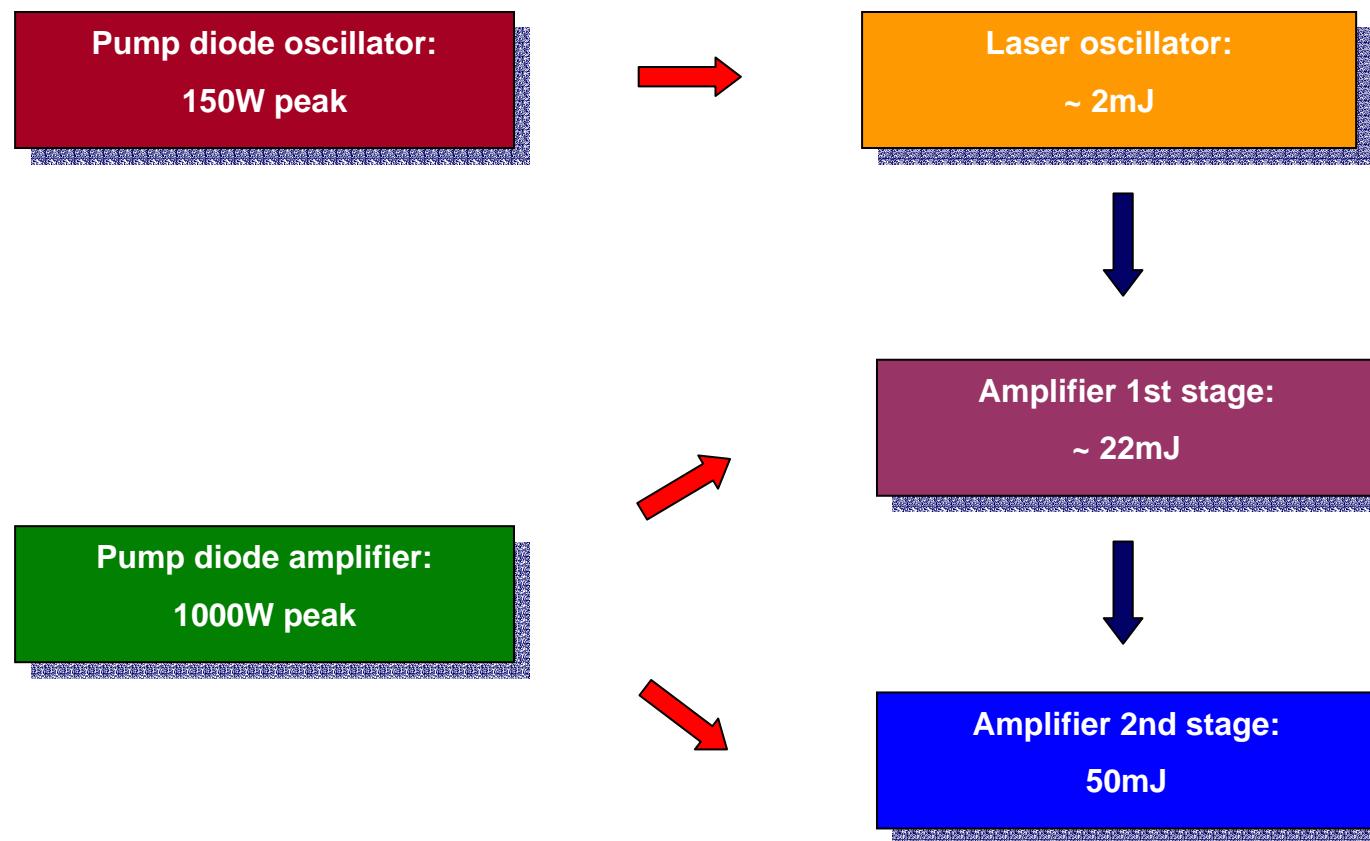
**MPS**

# *BELA Laser Requirements*

- Pulse energy >50mJ
- Pulse duration <10ns
- Beam quality  $M^2 < 1.6$
- Repetition rate 10Hz
- Wavelength 1064nm
- Radiation 100krad
- Op. temperature 20-45°C (laser head)  
18-33°C (pump diodes)



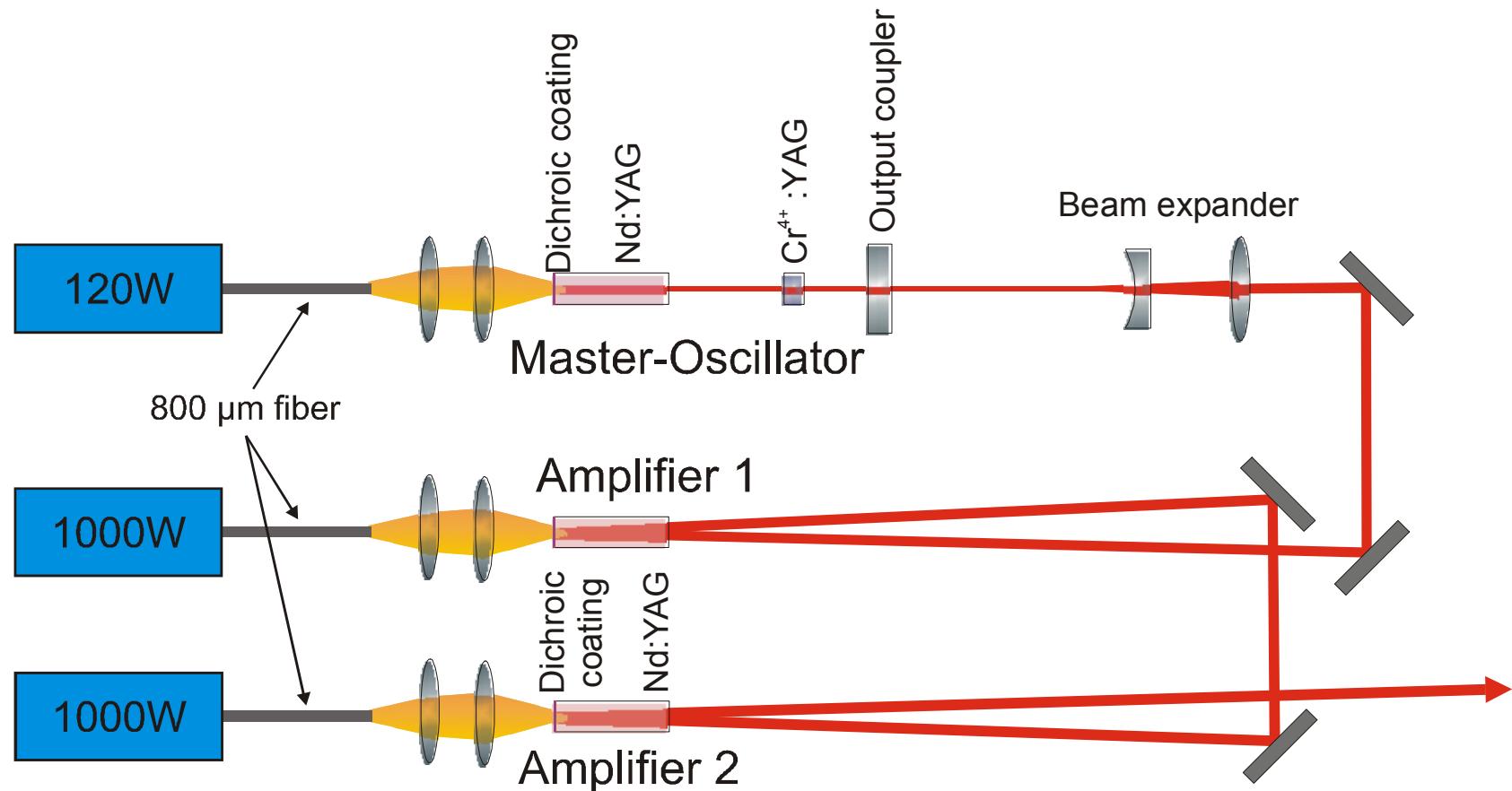
# *BELA Laser System Concept*



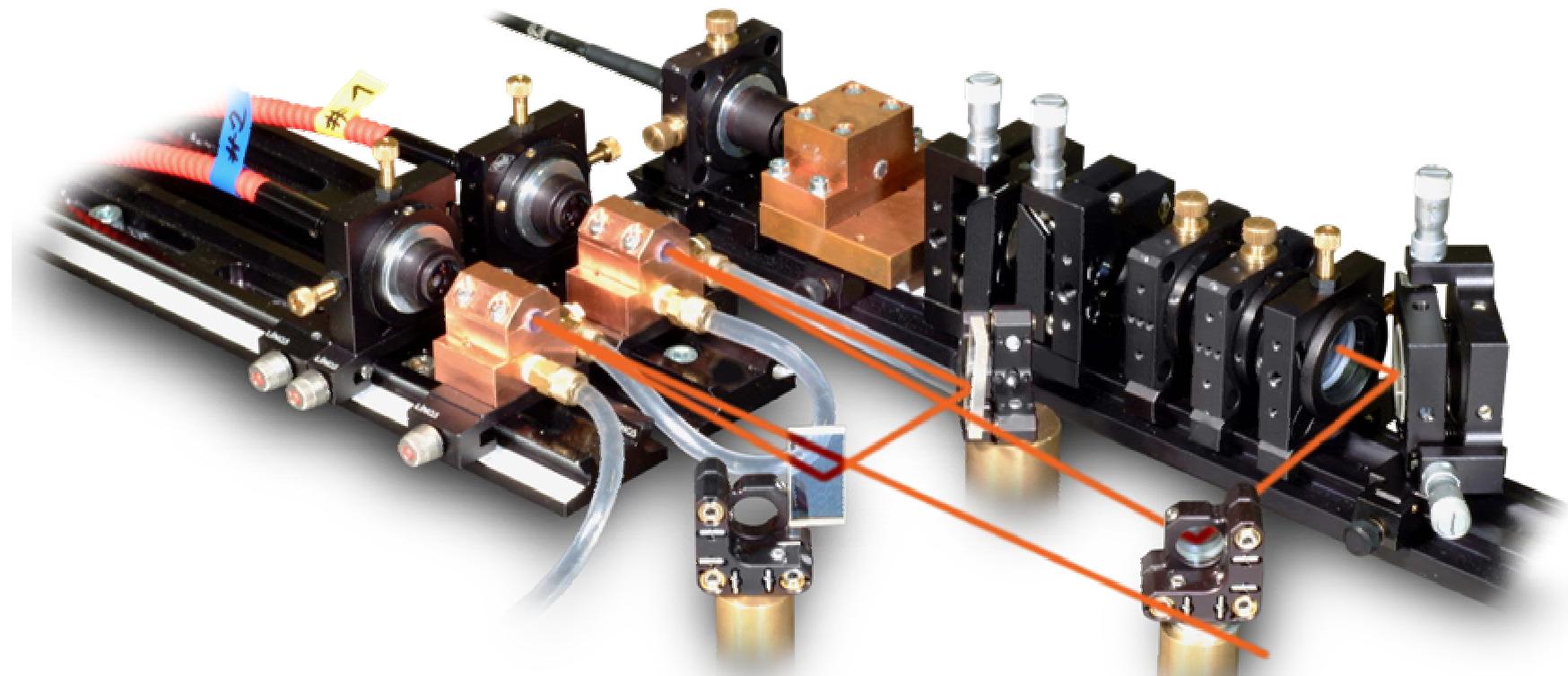
# *BELA Laser Design Concept*

- Fiber coupled pump diodes
  - Separation of pump source and laser head
- Longitudinal pumping scheme
  - Optimized overlap pump beam / laser mode
  - Higher efficiency
  - Long absorption path in laser crystal
- q-cw pumping
  - 200 $\mu$ s pump pulse duration as a trade off between efficiency and output energy
- Passive Q-switching with saturable absorber Cr<sup>4+</sup>:YAG
  - Simplicity
  - Low mass
  - Low power consumption
- 2-stage amplifier
  - Splitting of energy into 2 stages to avoid self-lasing

# *Master Oscillator Power Amplifier (MOPA)*

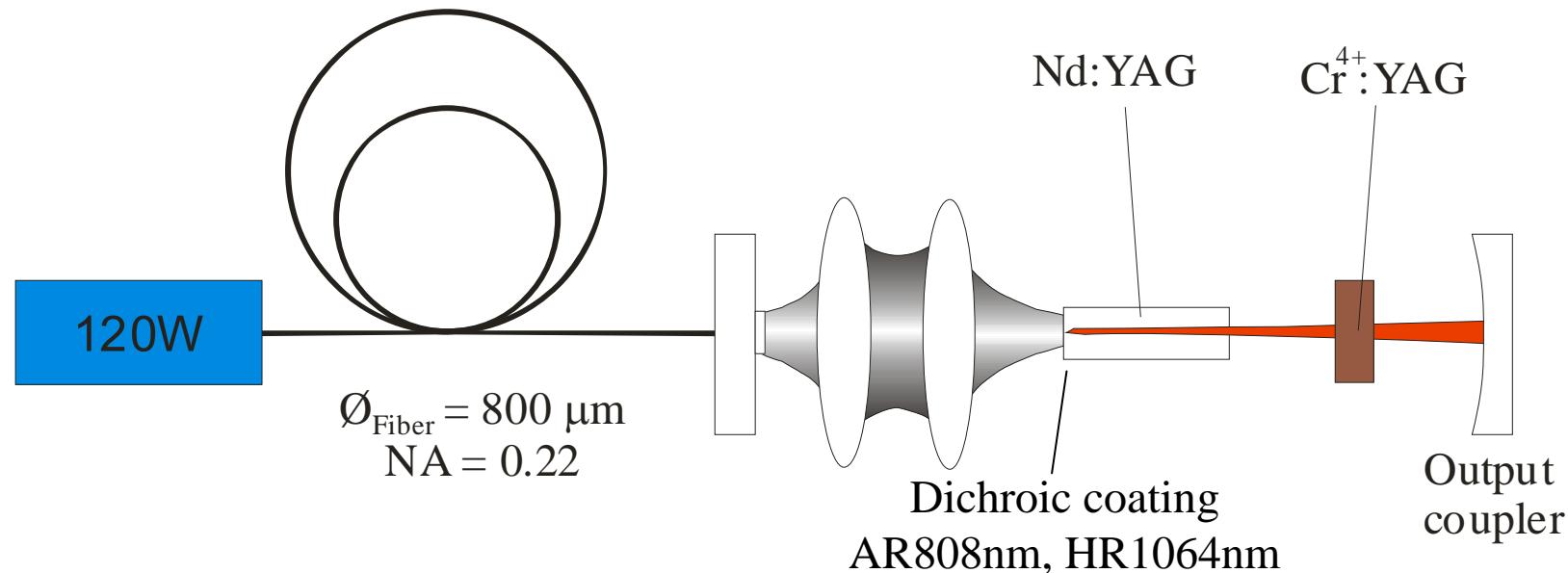


# *Miniaturized Testbed*



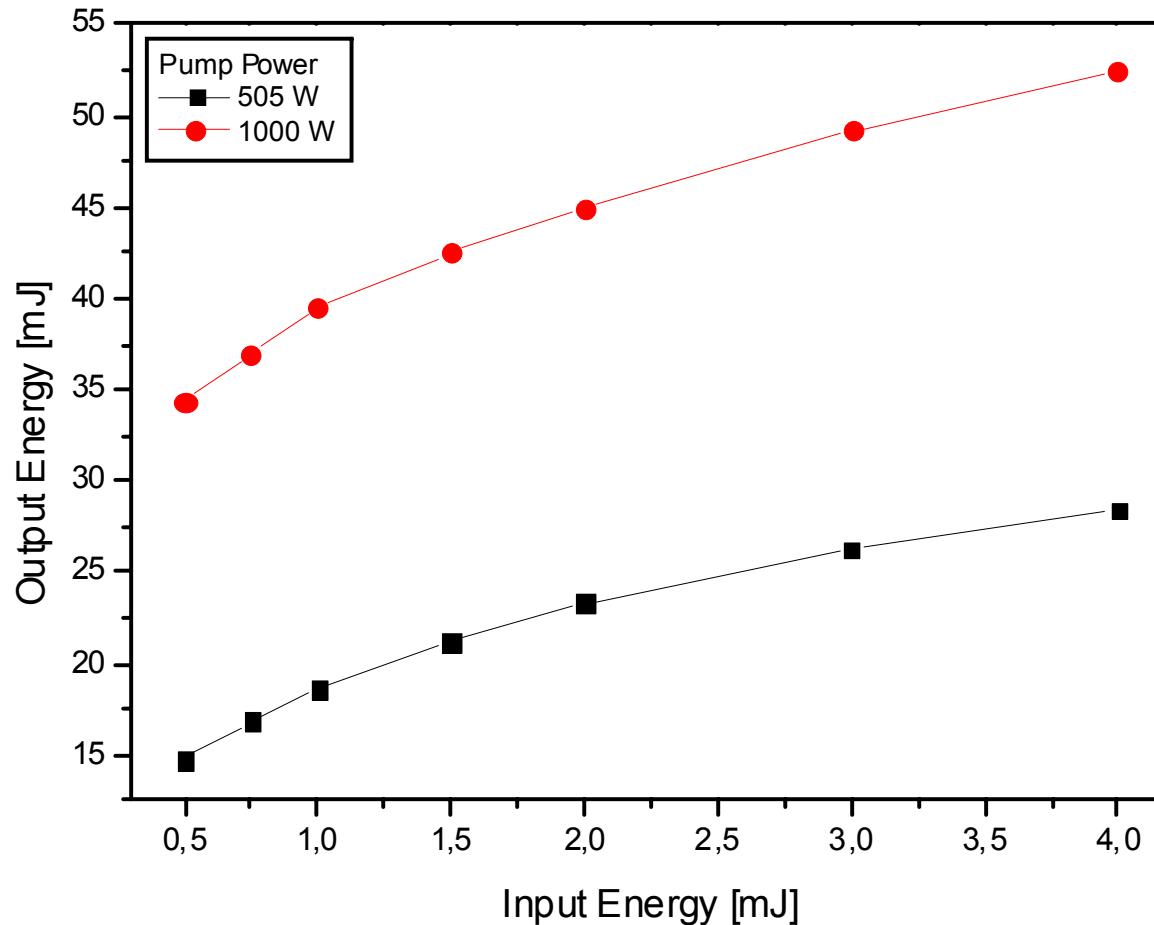
Mass of optical components ca. 40g

# *Laser Oscillator*



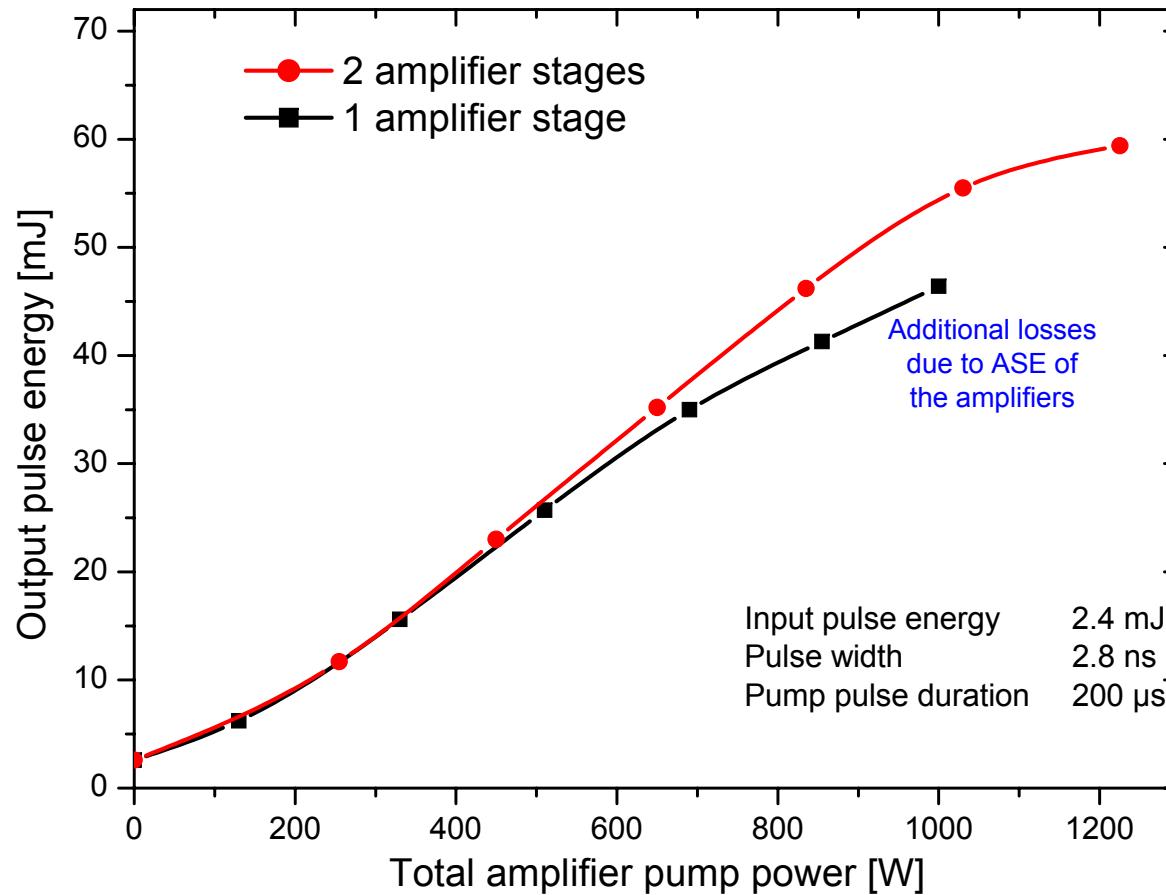
- Pulse energy: 2.4 mJ
- Pulse duration: 2.8 ns
- Beam quality:  $M^2=1.2$
- Peak-pump power: 104 W
- Pump duration: 200  $\mu\text{s}$
- Opt.-opt. efficiency: 12 %

# *Amplifier: 1st Stage*



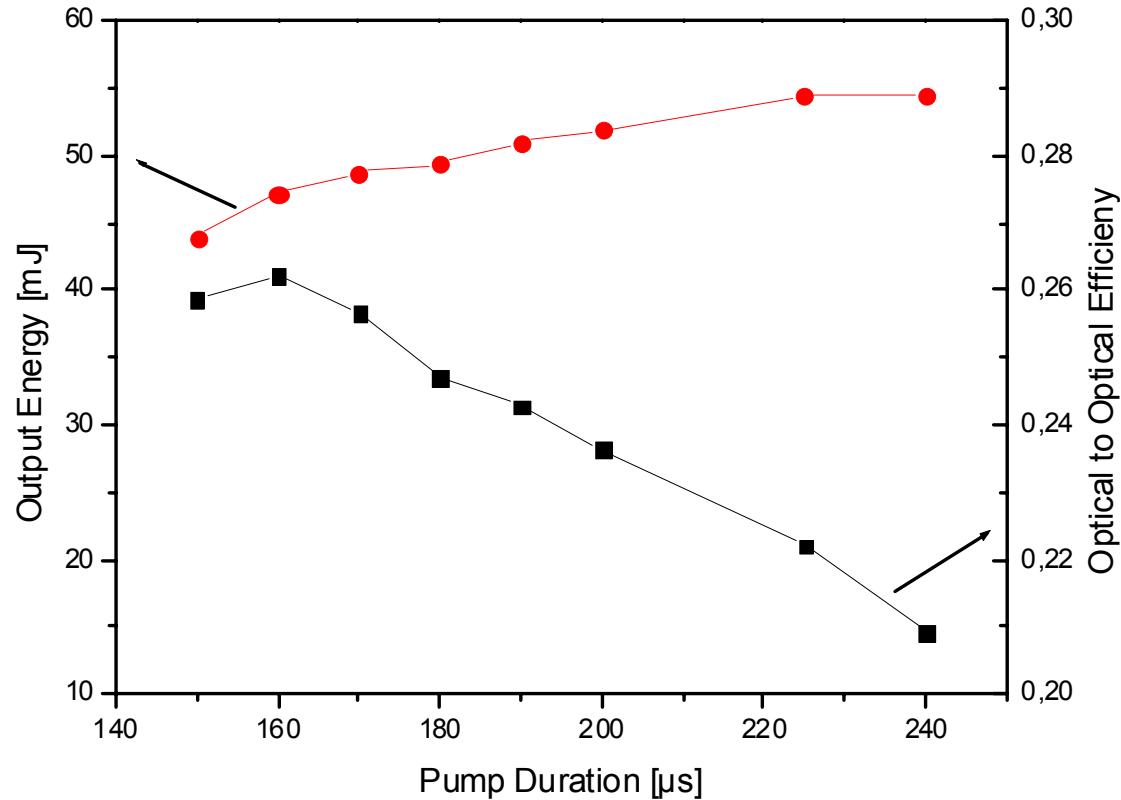
Saturated effective amplification @ > 2 mJ seed energy

# *Power Amplifier*



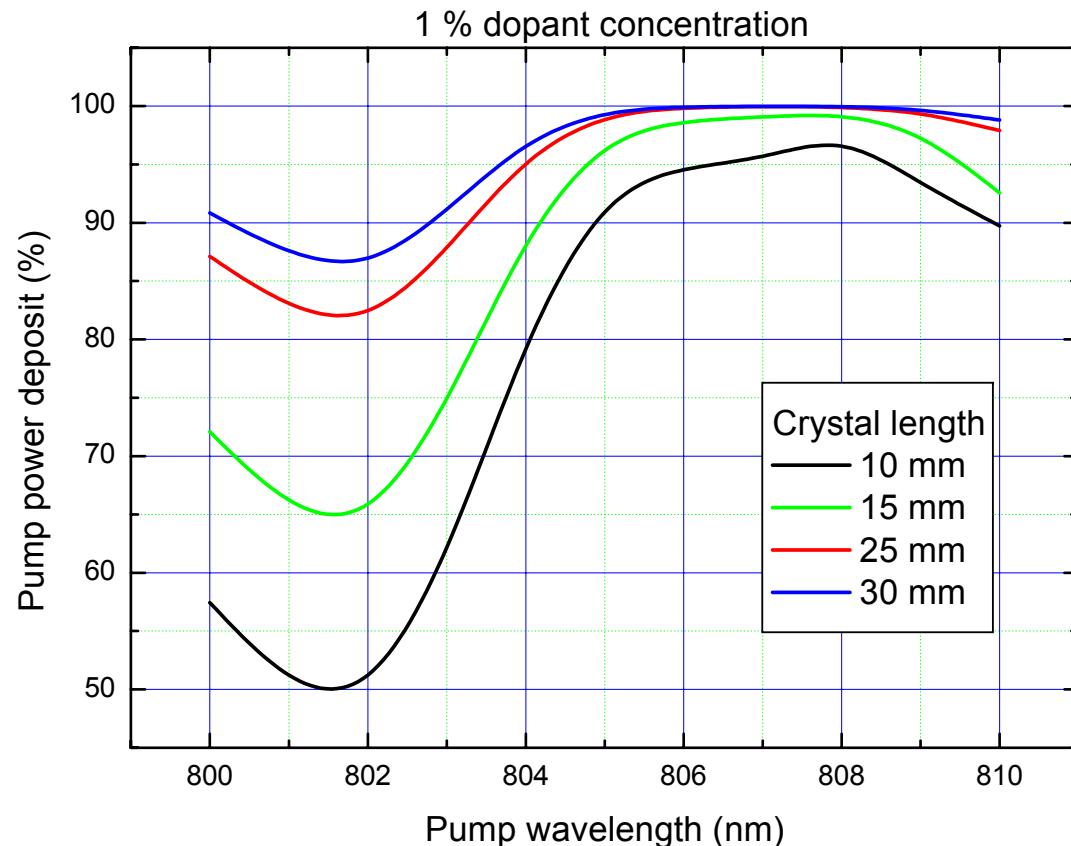
>50mJ @ 2 x 550 W, M<sup>2</sup><1.5

# *Optical-to-optical Laser Efficiency*

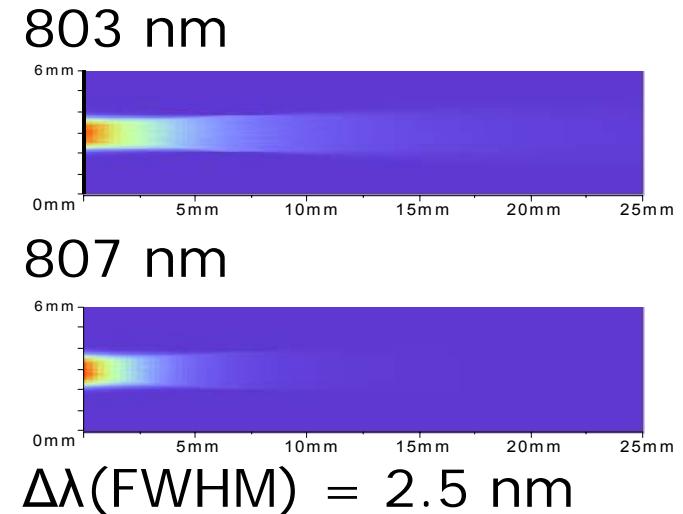


Optical-to-optical efficiency:  $\approx 23\% @ 200\mu\text{s}$

# Pump Light Absorption

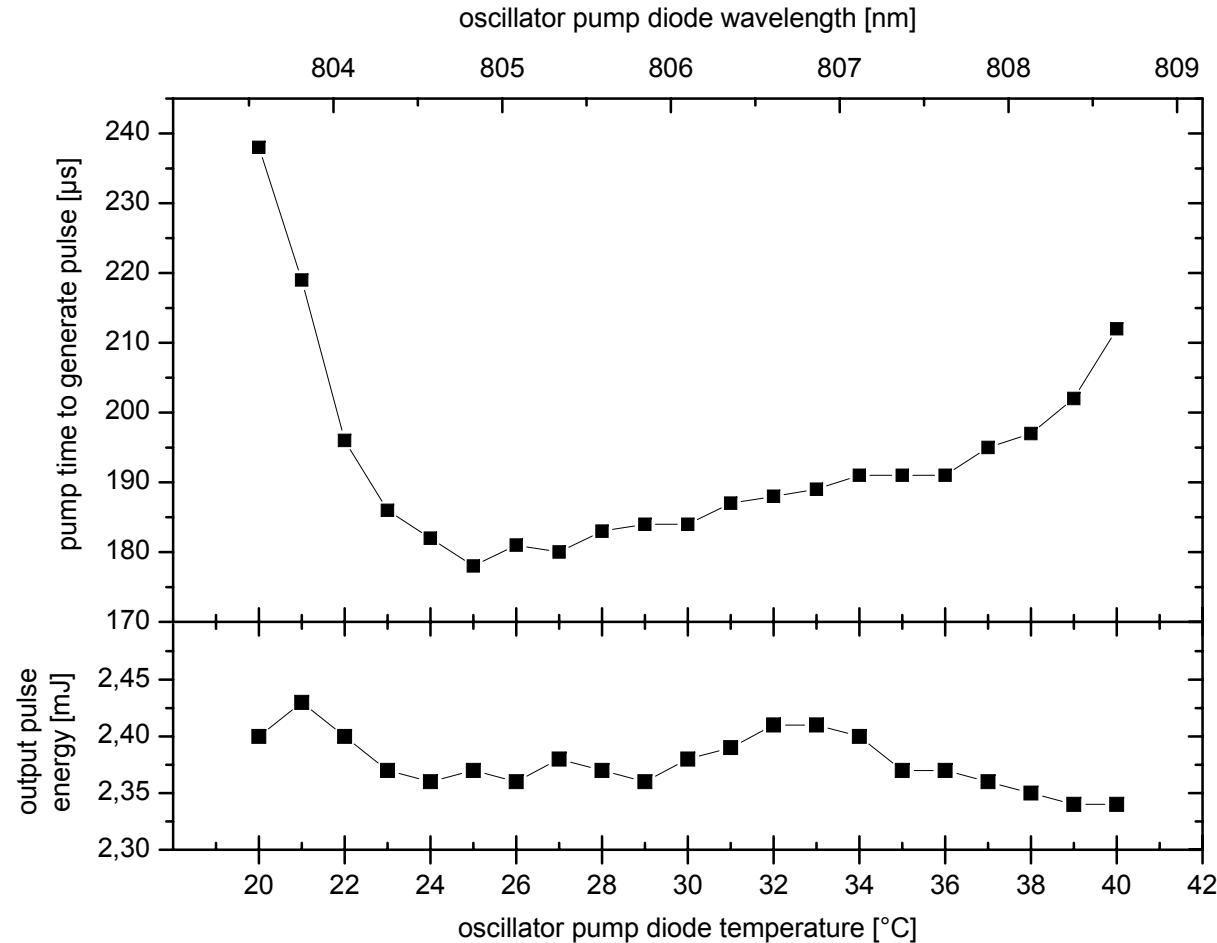


Pump diode shift: 0.25nm/K  
-> 3.75nm for  $\Delta T=15K$



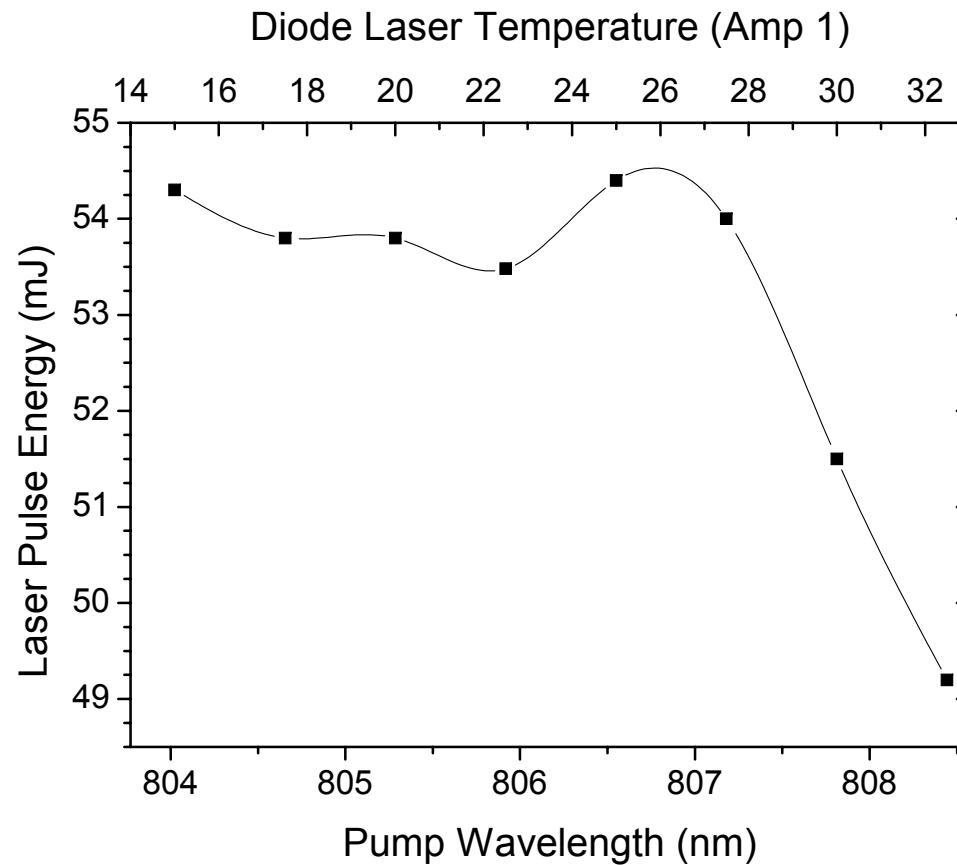
Absorption > 95% between 804nm and 809nm for 25mm crystal

# *Temperature Range: Oscillator Pump Diodes*



Temperature acceptance range:  $\Delta T > 15K$

# *Temperature Range: Amplifier Pump Diodes*



Temperature acceptance range:  $\Delta T > 15\text{K}$

## *Testbed Laser Performance*

✓ Pulse energy	54mJ
✓ Pump energy	245mJ
✓ Pump power	2x 550W + 100W
✓ Pump duration	200µs
✓ Beam quality	$M^2 < 1.5$
✓ Pulse duration	2.8ns
✓ Temperature acceptance range for pump diodes	$\Delta T > 15K$



## *Requirements for Flight Model*

- Mass (laser system) 4kg  
    Laser head <1.3kg
- Total ionizing radiation dose 100krad
- Vibration level (Sojuz) 26g<sub>rms</sub>
- Operating temperature 20-45°C
- Non-operating temperature -40-60°C
- Sealed pressurized box to avoid contamination of optics

**Mass of optical components approx. 40g**

 Most of the mass due to housing in sealed box

## *2 Prototype Models*

- Prototype I

Optical functionality for integration in BELA instrument prototype

- Prototype II

- Laser operation during thermal cycling
- Verification of thermal model
- Vibration tests of subcomponents

Modular concept: reversible screw joints / O-ring sealed -> easy replacement of non-adequate parts



LASER ZENTRUM HANNOVER e.V.

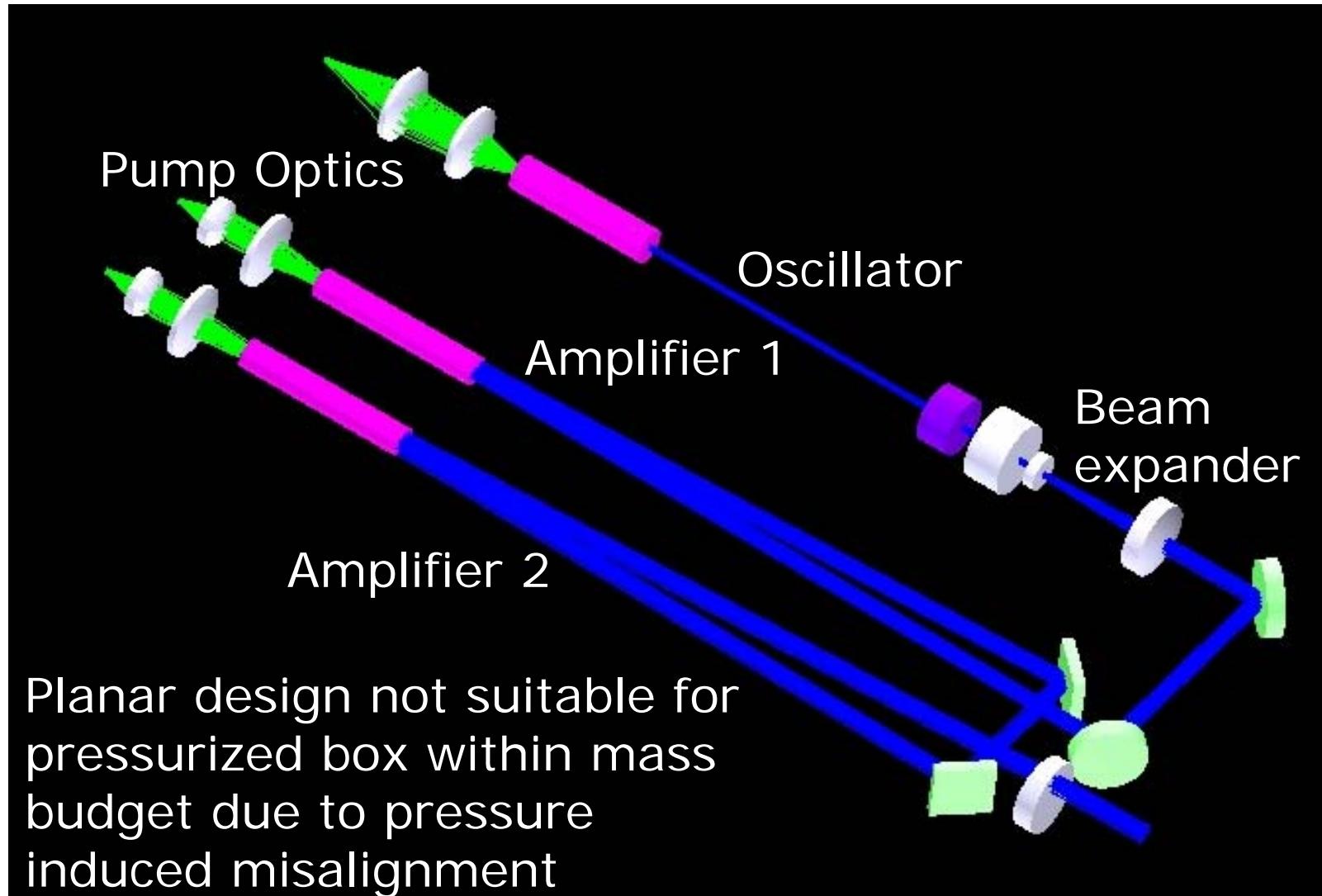
# *Prototype Model Optical Materials*

- Mirrors, lenses, windows fused silica
- Aspheric lenses in pump optics Co550
- Laser crystals Nd<sup>3+</sup>:YAG
- Passive Q-switch Cr<sup>4+</sup>:YAG

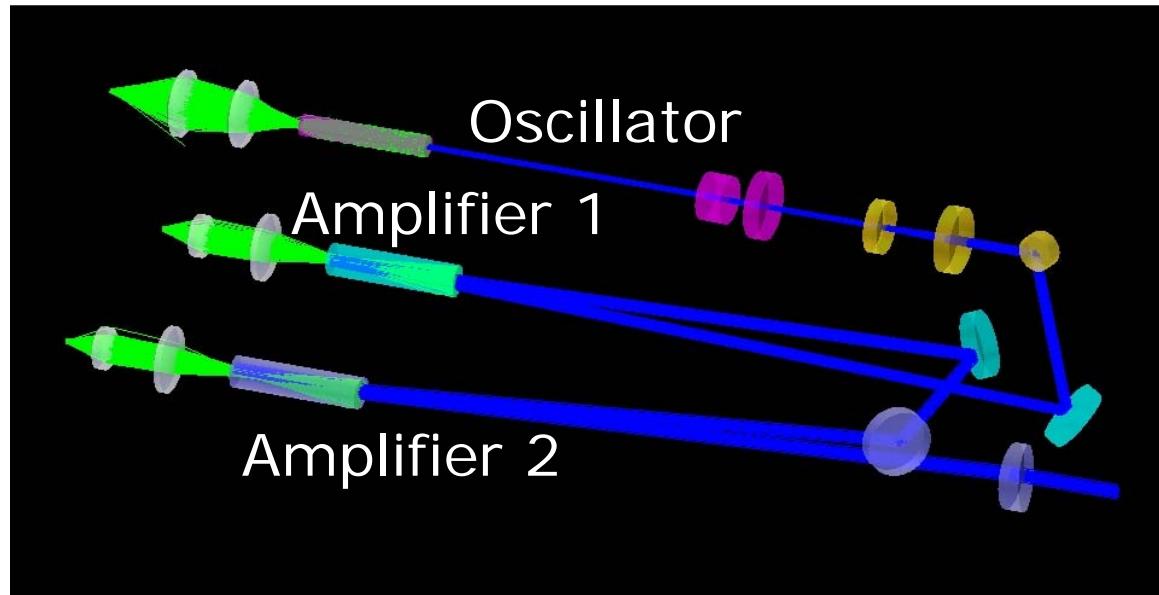


Materials already used for other space missions, i.e. can be made radiation hard

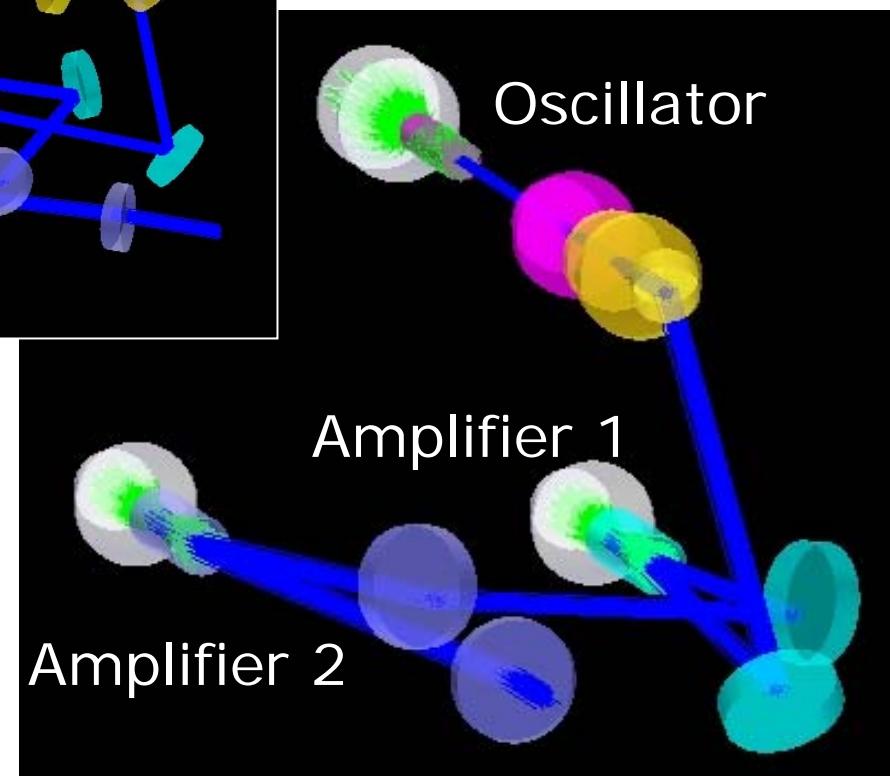
# *Planar Testbed*



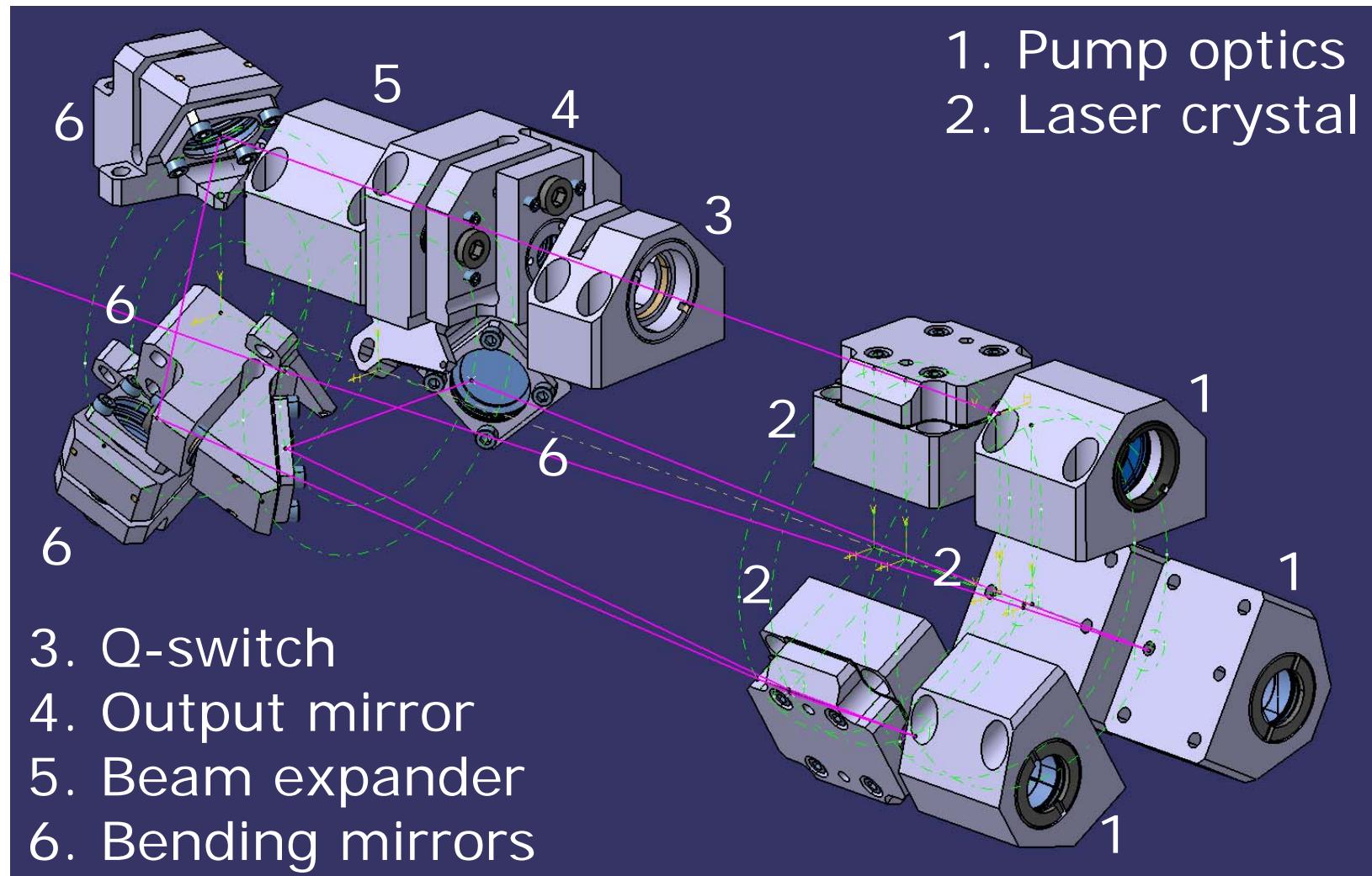
# *Optical Design for Prototype*



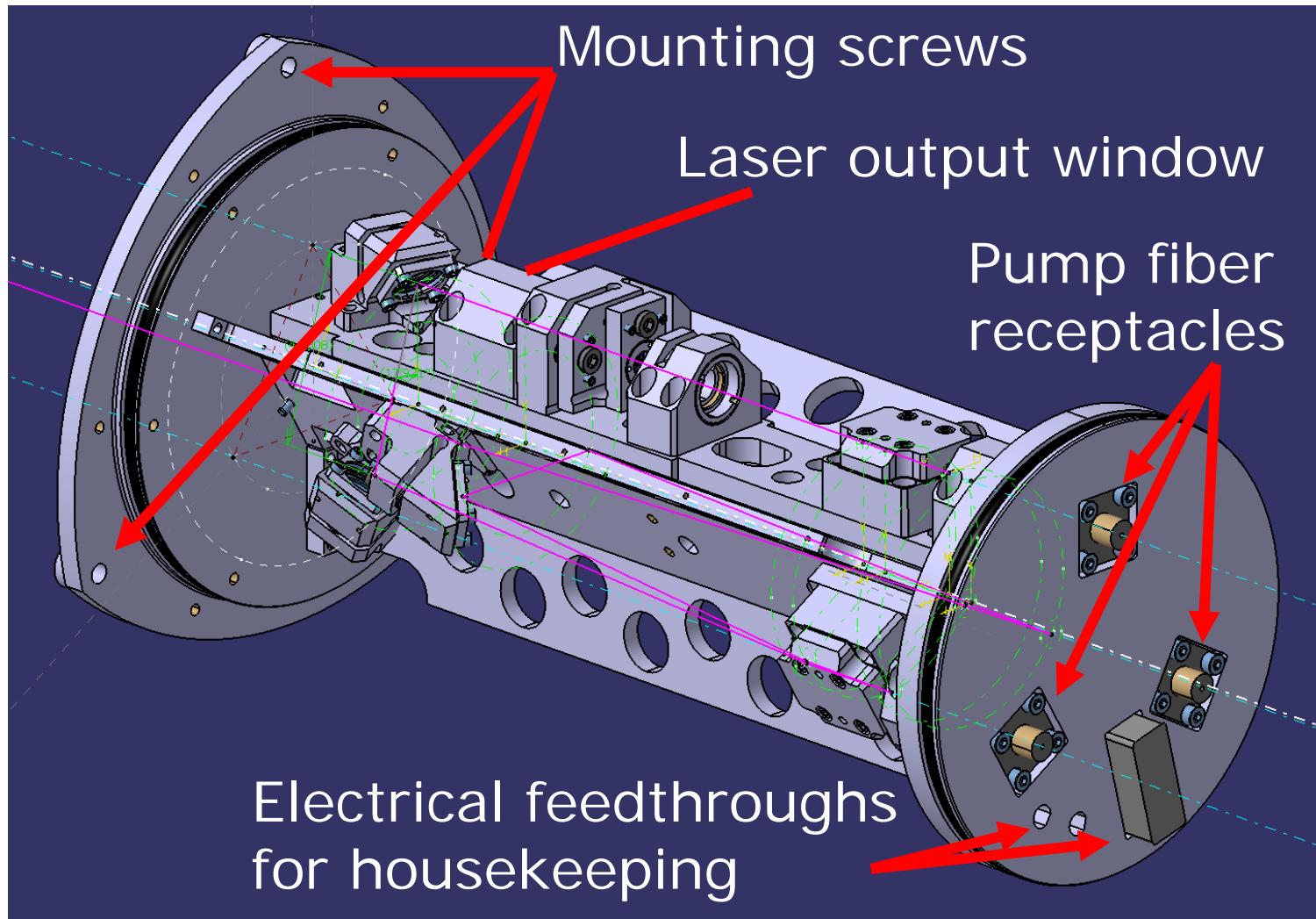
- 3D Optical Design (Zemax)
- Only reflection angles at bending mirrors have changed compared to testbed



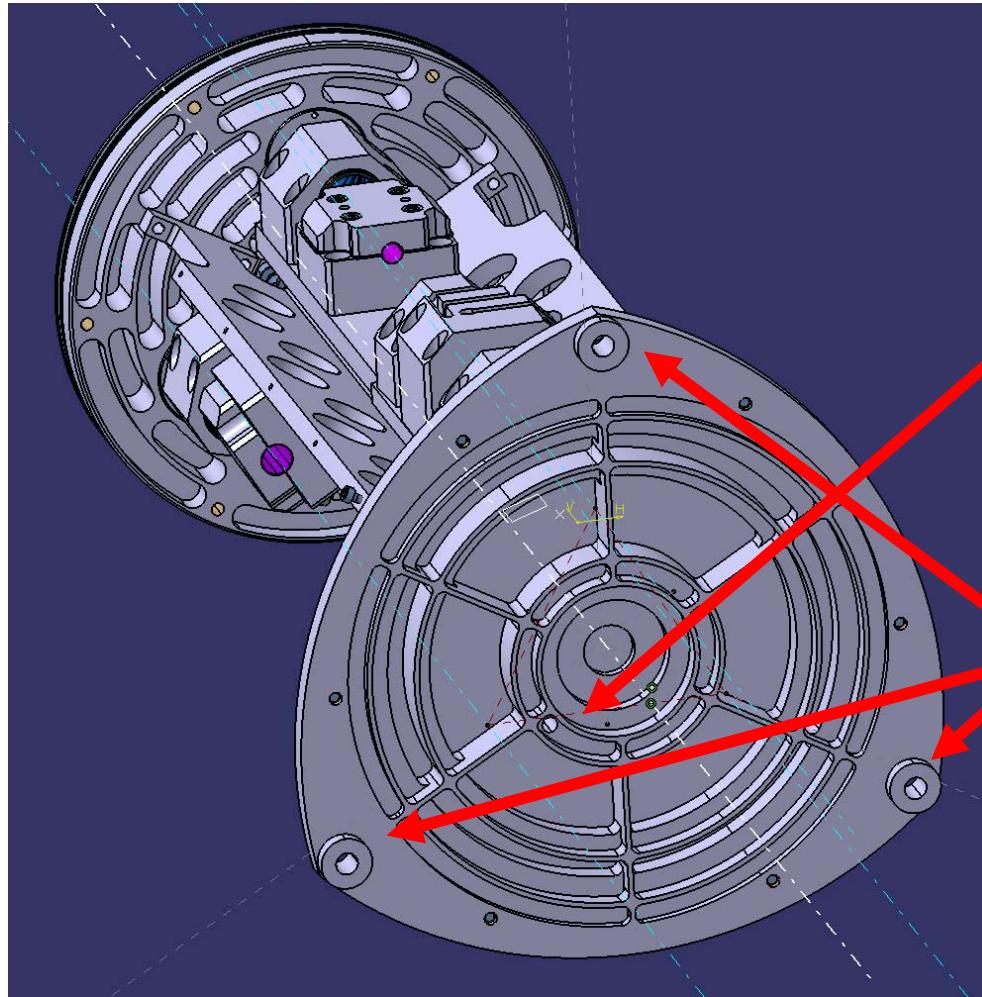
# *Mechanical Mounts for Optics*



# *Optical Bench*



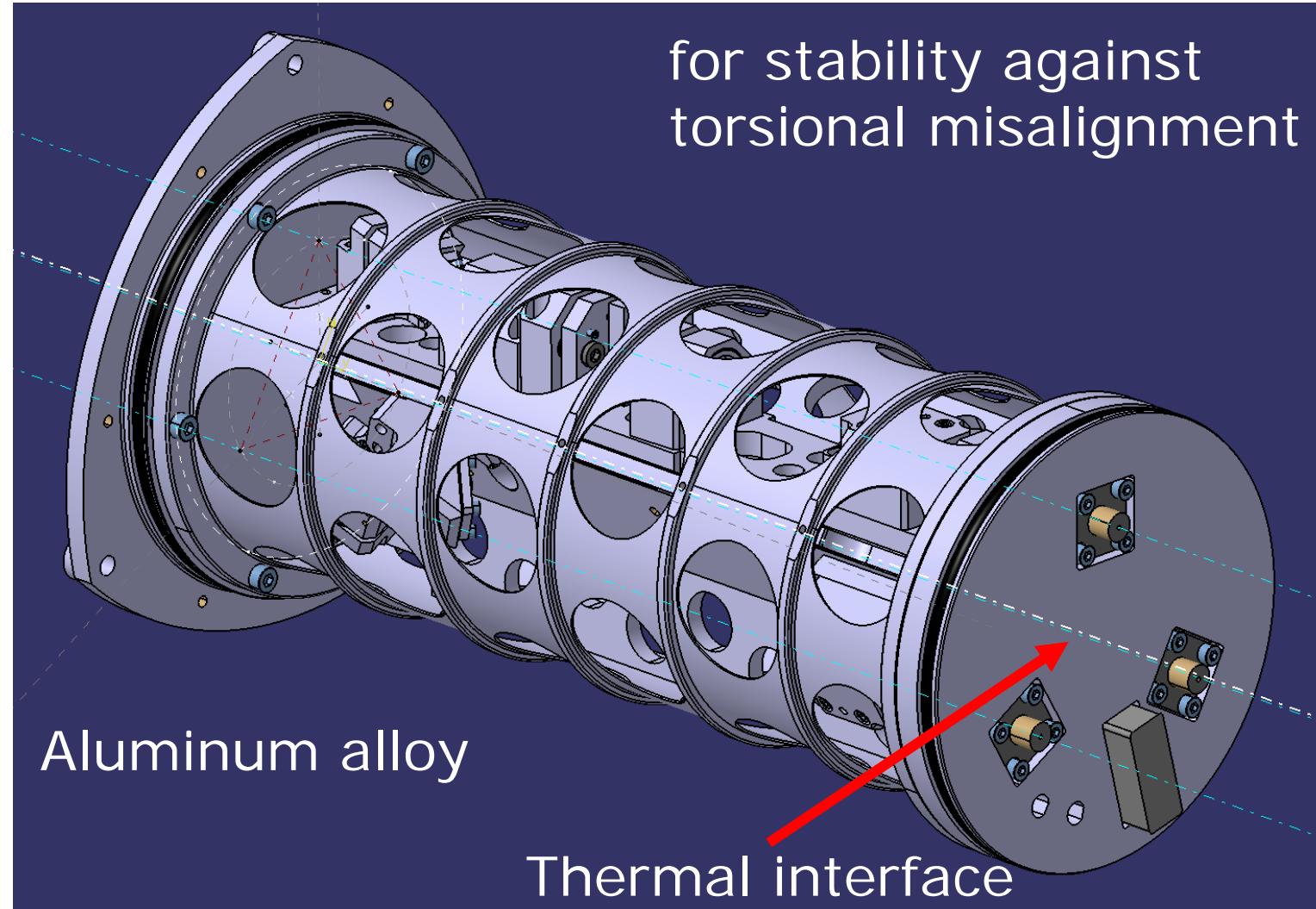
# *Mounting*



Laser output window

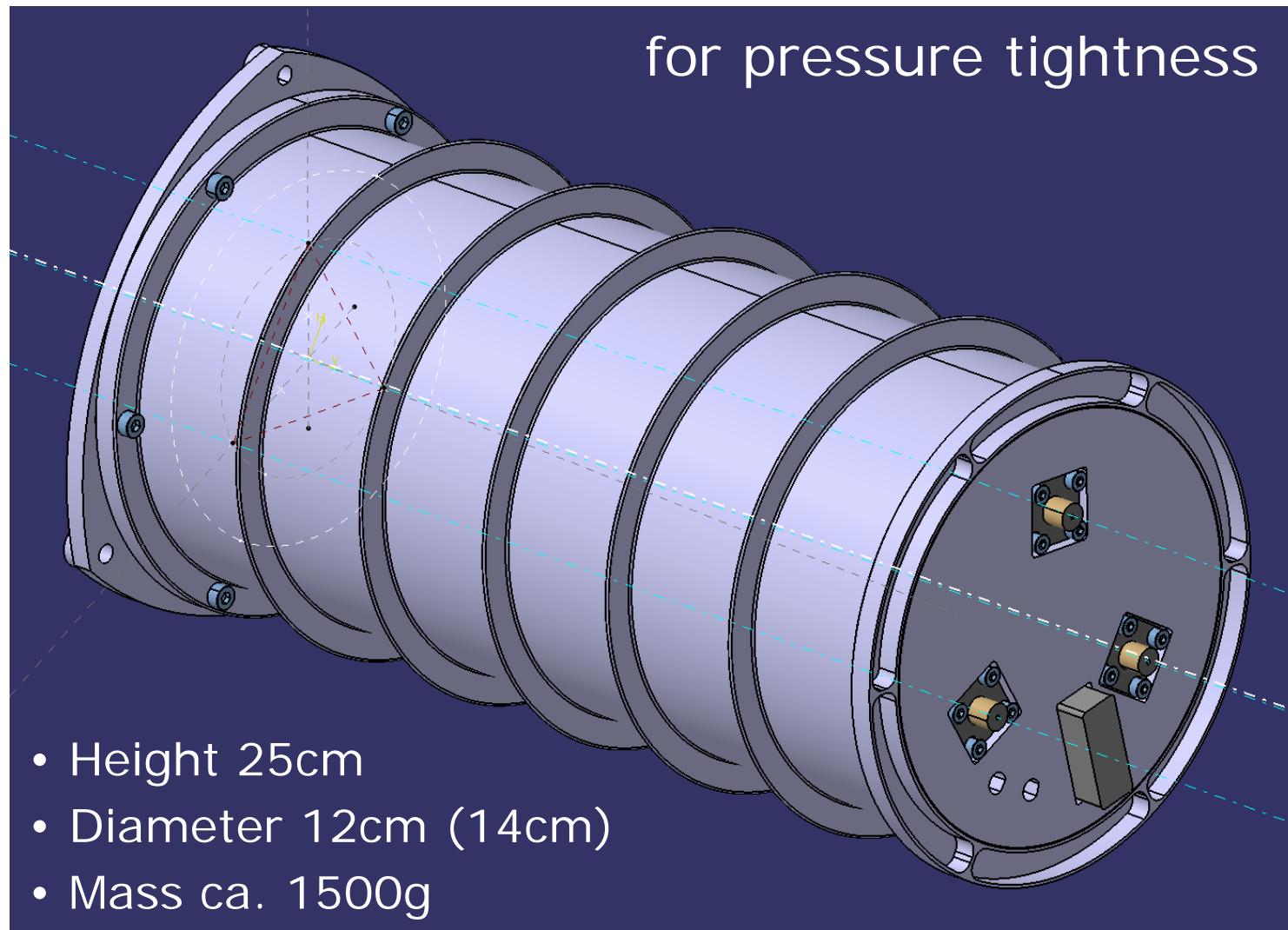
Mounting by  
M4 screws

# ***Stabilizing Jacket***



# *Sealing Jacket*

for pressure tightness



- Height 25cm
- Diameter 12cm (14cm)
- Mass ca. 1500g

## *Mechanical Design*

- Good volume to surface ratio -> low mass for pressurized box  
(Prototype Model ~1.5kg, Flight Model <1.0kg)
- Rugged optical bench due to highly symmetrical design (mainly symmetrical radial forces induced by pressure difference)
- Low thermally induced optical misalignment due to almost symmetrical thermal load
- Transport of dissipated heat via massive optical bench



## *Future Work*

- Thermal cycling test end of 2006
- Vibration tests of critical subcomponents
- Radiation hard optical components
- Mechanical redesign after thermal cycling / vibration tests
- Replacement of screw joints by irreversible joining techniques (welding, soldering, etc.)



Qualification procedure

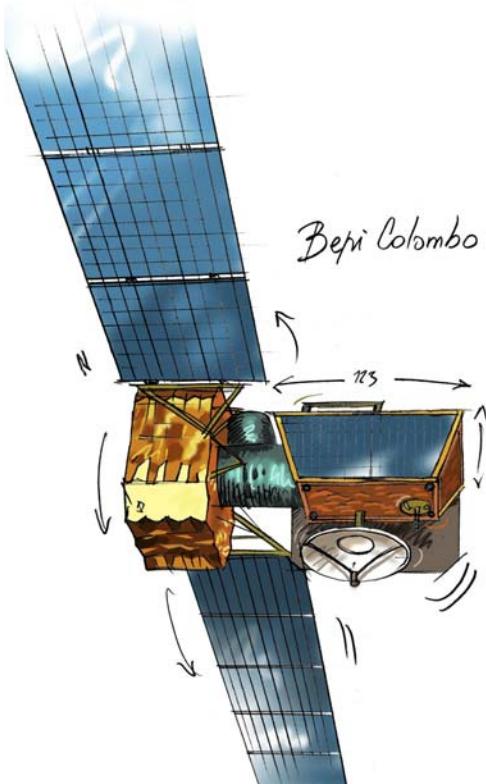
This work was funded by

- German Aerospace Center (DLR)
- Max Planck Institute for Solar System Research (MPS)

and performed within the framework of the BELA Laser Industrial Team



for



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