

2nd ESA/NASA Working Meeting on Optoelectronics



Qualification, Performance Testing and Characterization of Quasi-CW Laser Diode Arrays

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OUTLINE

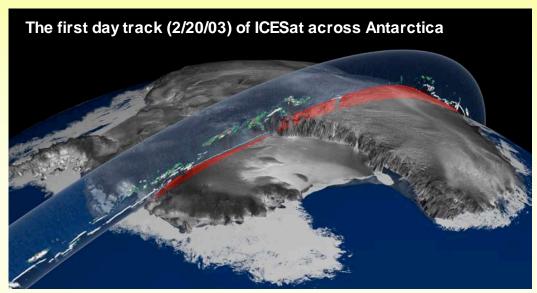


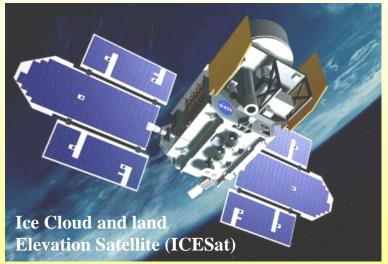
- Introduction and research objectives
- General characterization measurements
- Performance tests
- Future work



INTRODUCTION











Challenges for QCW LDAs for Space flight



- The arrays are the power source for laser and potentially a single point failure for the instrument
- LDAs are complicated devices with multiple failure mechanisms so predicting reliability is difficult
- QCW operation causes heating with every current pulse which puts repeated thermo-mechanical strain on device
- QCW market does not support the statistically verified reliability testing found in the telecom market.
- QCW LDAs are used in a many applications with different operational parameters which further fractures the QCW market
- Statistics are expensive because of the cost of the arrays
- Vendor designs, procedures, and tests change often in an effort to improve package design which can degrade (or negate) the statistics you gather on previous devices
- It is a competitive business so vendors can come and go



Research Objectives

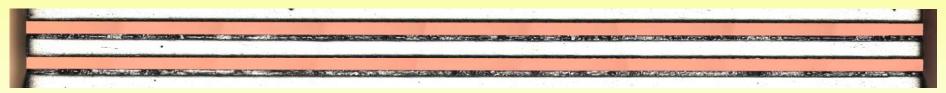


- → Develop procedures for purchasing, handling, storage and operation
- → Improve understanding of device operation and failure modes
- → Quantify affect of operational and environmental parameters on LDA performance
- → Develop prediction/screening capability
- → Enable improved reliability and performance of future missions



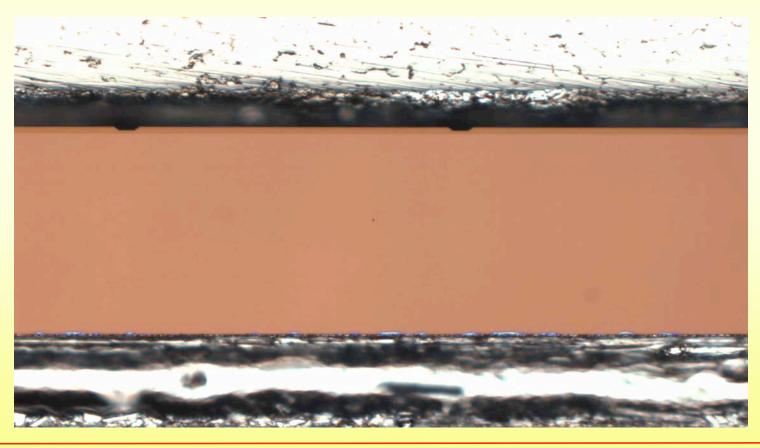
Microscopic Facet Inspection: Bright Field 50x 200x magnification





Stitched image of 2-bar array [Bar length is 1 cm]

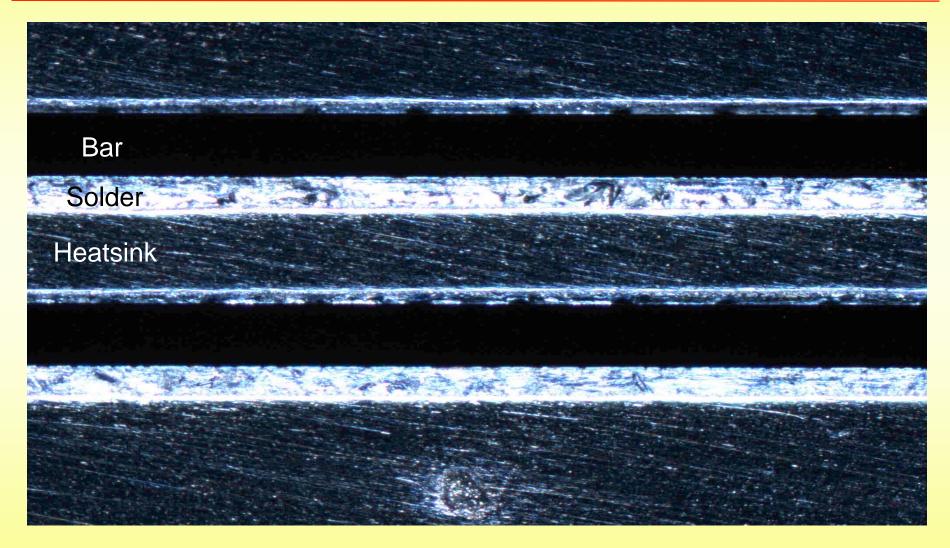
Single image showing resolution of 200x micrsocopy [Bar height is ~120 µm]





Microscopic Facet Inspection: Dark Field 50x Magnification



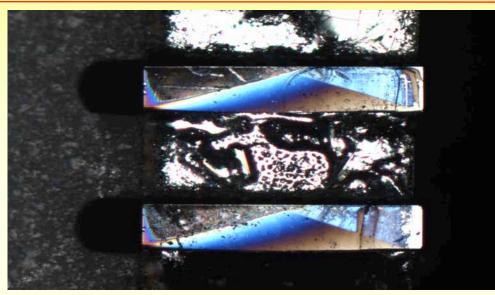




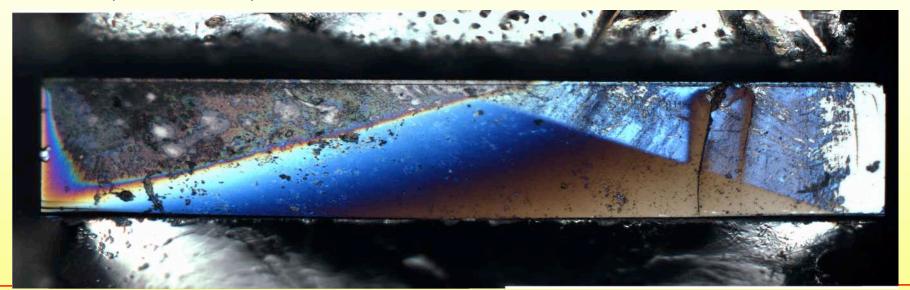
Microscopic Side View Inspection: Bright field 50x, 100x magnification



Bright field 50x



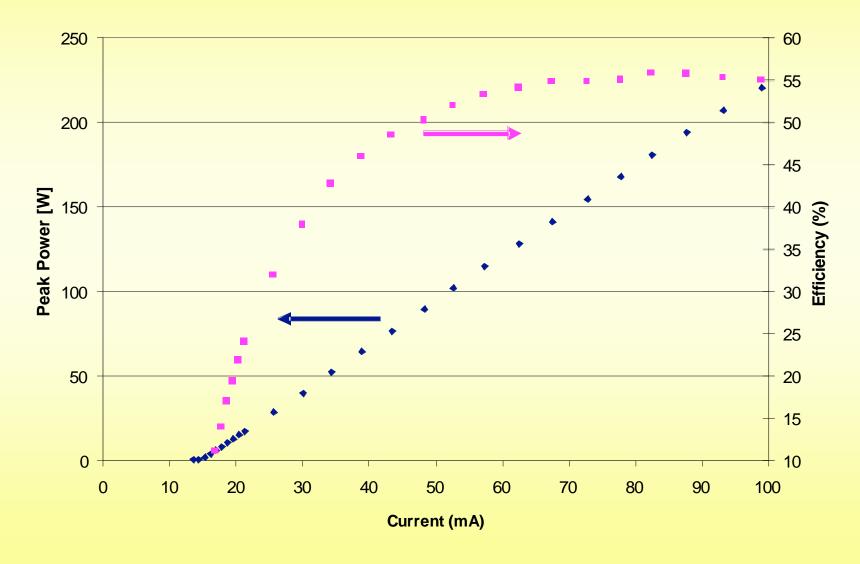
Bright Field 100x (stitched)





Power and Efficiency vs. Current

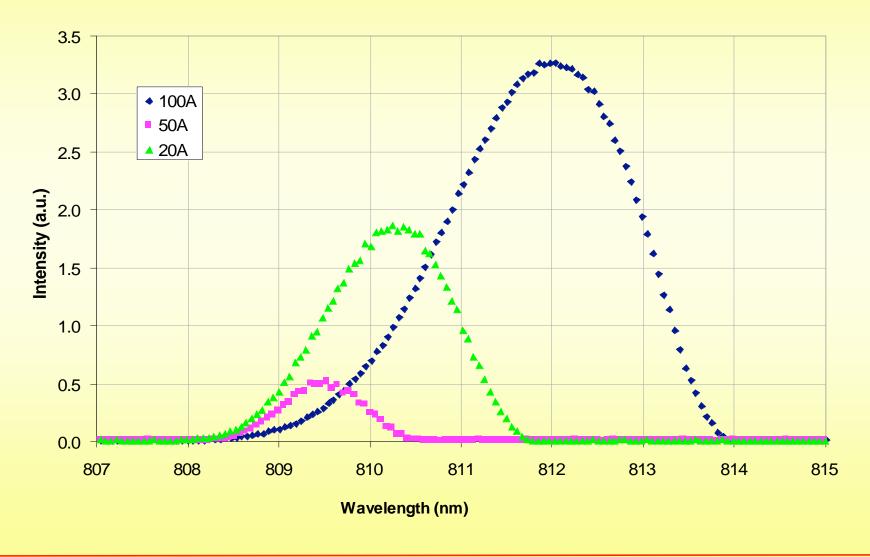






Averaged Optical Spectrum

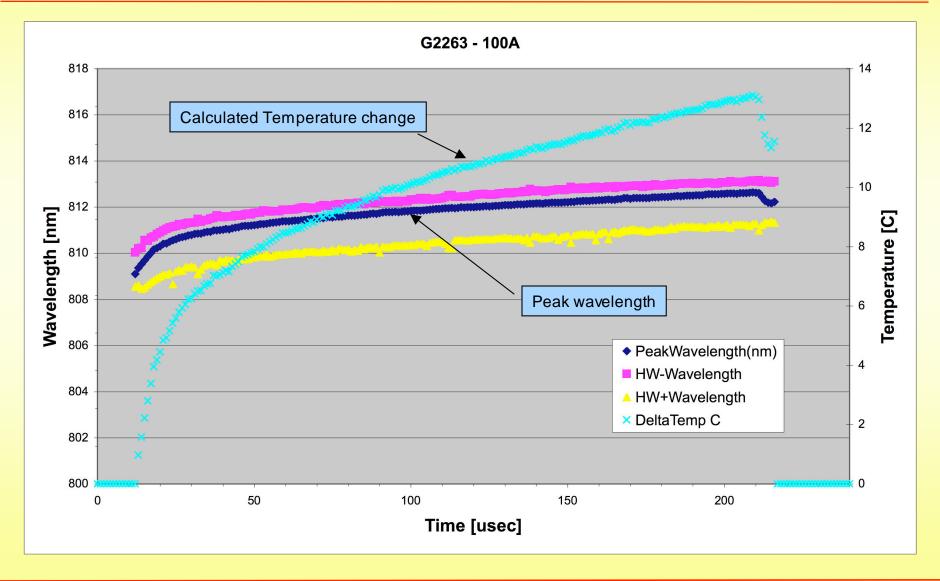






Time-resolved Spectrum



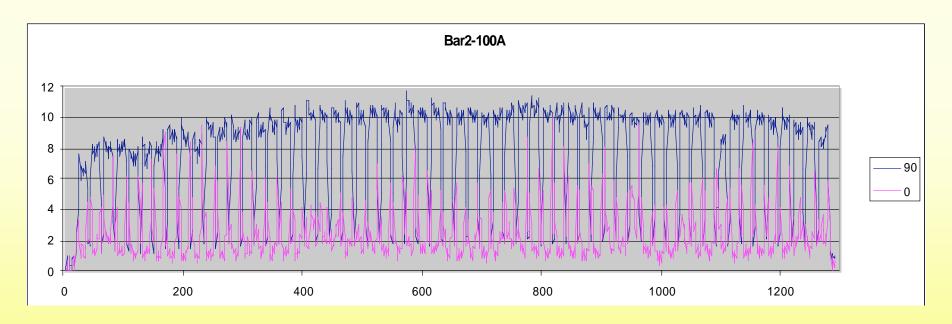




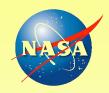
Near Field Inspection – 100A



Image of 2-bar array during operation showing optical power distribution

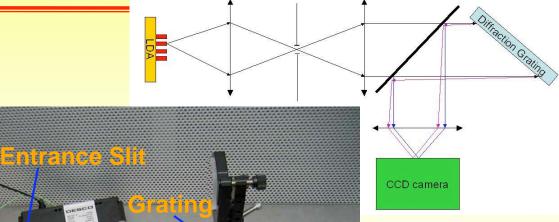


Polarization resolved near field measurement



Imaging Spectrometer





Entrance slit

ogibni

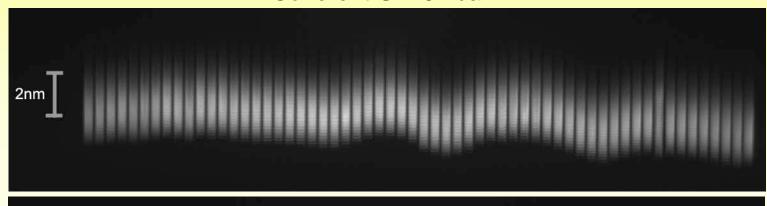


Wavelength

Imaging Spectrometer Measurement



Coherent G2261 bar#2



100A



20A

Emitter Position

NOTE: Bar tilting shown here is due to alignment and not wavelength shift

Zoomed image illustrating spectral resolution and cavity modes

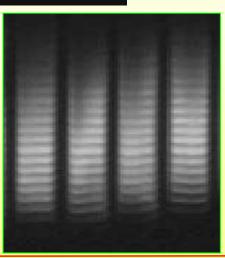
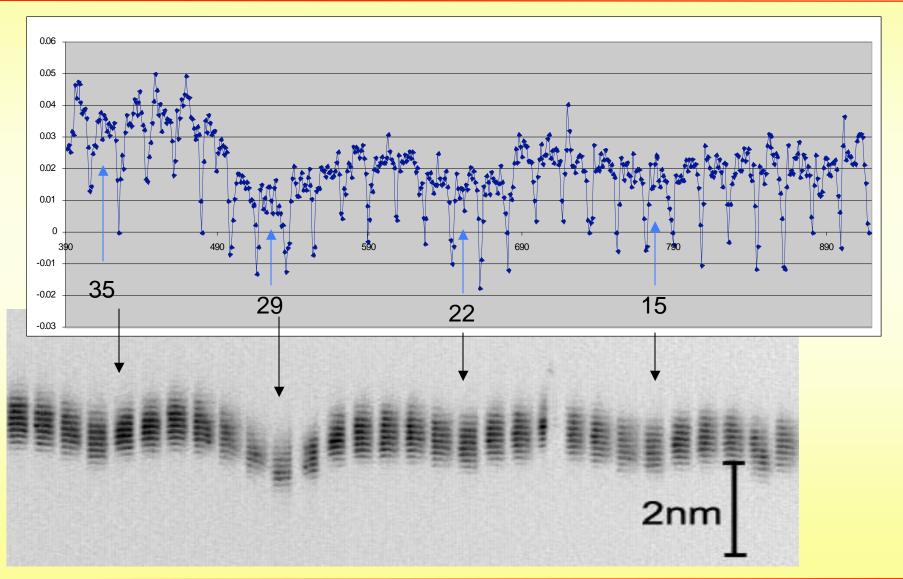




Photo Emission and Imaging Spectrum

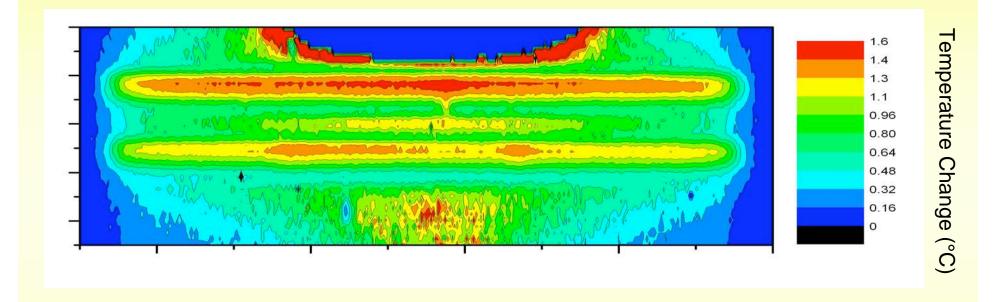






Thermal Image using Infra Red (IR) Camera





Bar Width (1 cm)



Characterization Procedure

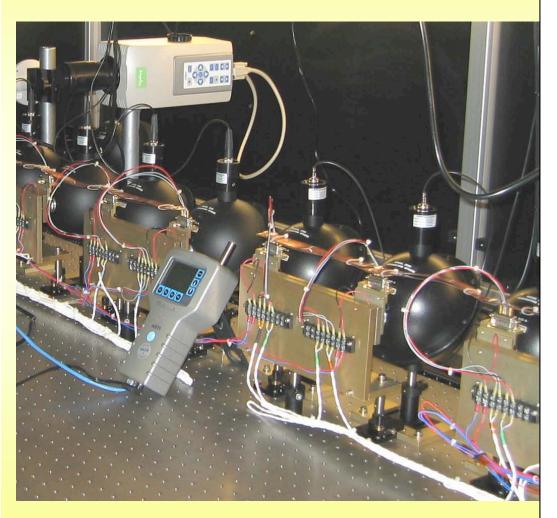


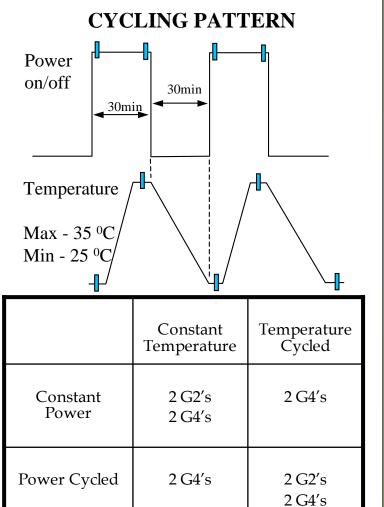
- Microscopic inspection
 - Facet inspections BF 50x, 200x, DF 50x
 - Side views BF 50x, 200x
- IR imagery
 - Averaged
- Spectral measurement
 - Spatially and temporally averaged
 - Spatially resolved
 - Temporally resolved
- Optical Power
 - Average, Peak, power /emitter, time-resolved,
- Near Field
 - Normal, polarization resolved
- Electronic Parameters
 - Current Pulse width, Amplitude, Efficiency, voltage, threshold current



Power / Temperature Cycling Test for MLA



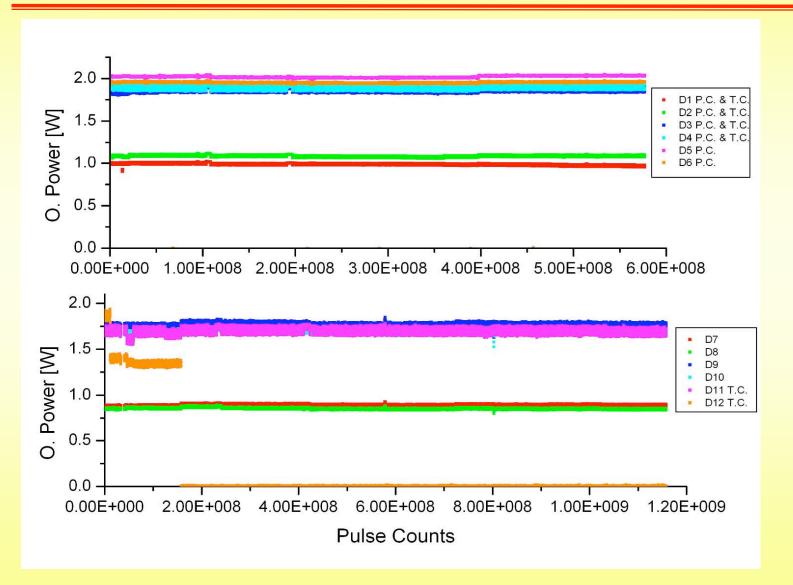






Optical Power vs. Pulse Count



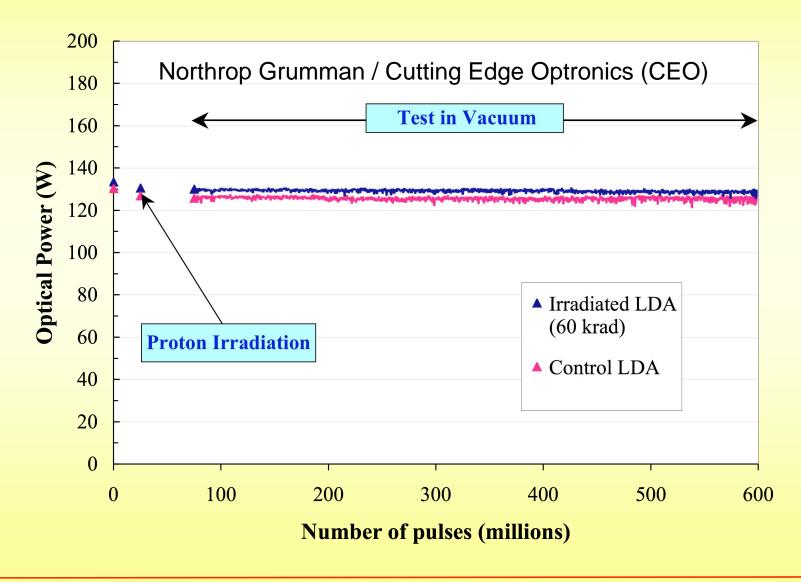


Mercury Laser
Altimeter
(MLA) era
arrays under
power and
temperature
cycling



Vacuum Results







Custom Vacuum Chamber



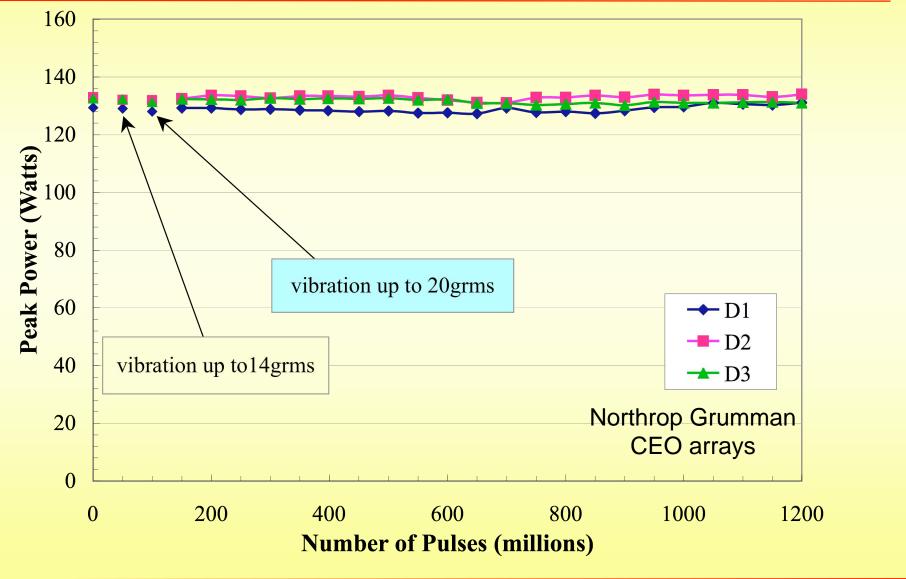
Custom vacuum chamber with 12 LDA test positions with windows for continuous inspection





Extended Operation of Vibrated Arrays



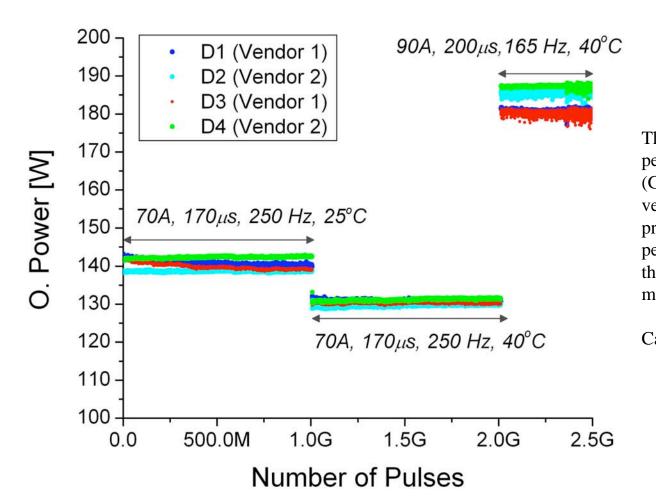




LOLA EM Test







This is an ongoing, accelerated performance test of 4 LDAs (G2 packages) to qualify two vendors, observe potential problems and compare performance to assist choosing the flight vendor for LOLA mission.

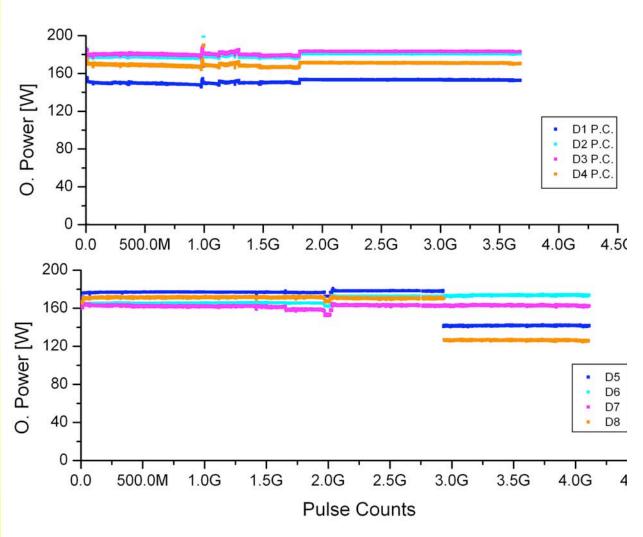
Calculated MTTF - >4 Billion



BioMM Test:

(CEO Arrays)





- Operating Conditions: I = 50 A, PW = 80 μs, f = 242 Hz, T= 25 °C.
- All LDAs have accumulated more than 3.6 billion pulses.
- 4 G-4 LDAs (top) are power cycled: ON cycle is 18 min.; OFF cycle is 2 min. [>14,000 cycles].
- 4 G-4 LDAs (bottom) are at constant power.
- Fluctuations in curves are due to test electronics and not indicative of changes in LDAs.
- Power drops in CW graph near 2.9 Billion pulses indicate bar failure.



Characterized Arrays



VENDOR	TYPE	NUMBER OF LDA	NUMBER OF BARS	
SDL	G 11	6	66	
JDL	G 16	5	80	
	total	11	146	
	G 2	14	28	
	G 4	37	148	
CEO	G 6	9	54	
	G 11	5	55	
	G 18	10	180	
	total	75	465	
	G 2	20	40	
Coherent Inc.	G 4	8	32	
Conerent inc.	G 6	8 2 2	12	
	G 16		32	
	total	32	116	
	G 2	5	10	
Nuvonyx	G 4	5	20	
	G 11	5	55	
	total	15	85	
TOTAL		133	812	
Uncharacterized				
Decade	G 2	5 2 7	10	
Decade	G 4	2	8	
	total	7	18	
Lasertel	G 2	5	10	
	G 4	2	8	
	total	7	18	

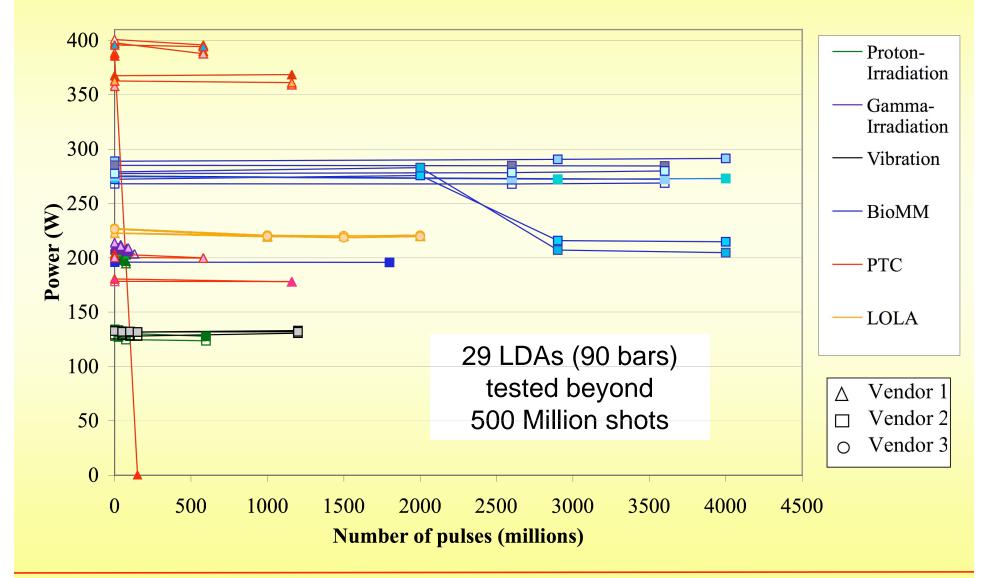
Total number of LDA
Total Hulliber of LDA
39
50
11
16
7
10

TOTALS PER VENDOR				
SDL	11			
CEO	75			
Coherent	32			
Nuvonyx	15			



Summary of Extended Testing During Program

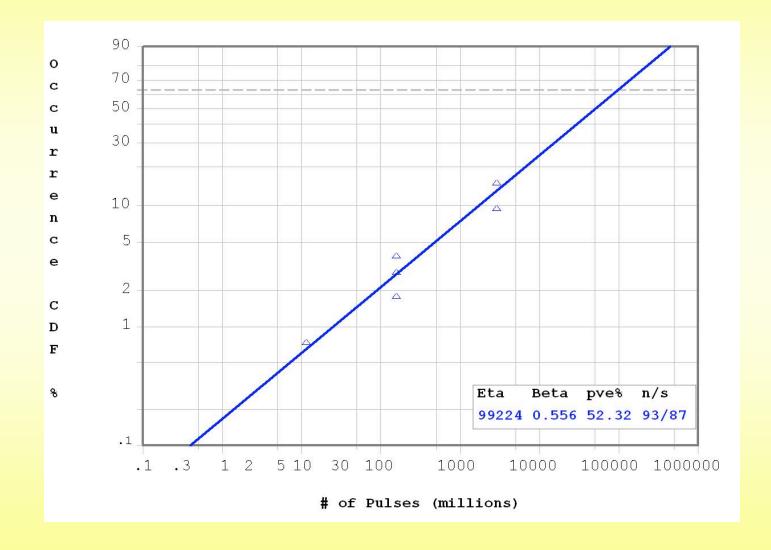






Weibull Analysis







Summary and Future



- Characterization measurements
- Power/Temperature cycling (MLA) test 1.15 billion pulses
- Finished vacuum/radiation test 600 million pulses
- Finished vibration test 1.2 billion pulses
- LOLA engineering model LDA test 2.5 billion pulses
- BioMM test continues ~4 Billion pulses, >14,000 power cycles
- Compiled results from all extended testing
- Statistical analysis of completed tests is being conducted
- Testing in air and vacuum for LOLA flight arrays is planned for this summer
- Correlate failures to initial measurements