



## Micro-sources of energy for wireless sensor applications

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The logo for Liten is the word 'liten' in a bold, blue, sans-serif font.

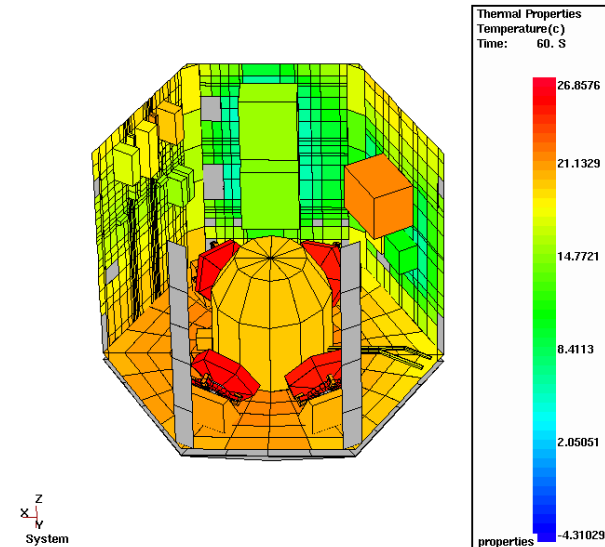


- ❖ Energy source available on-board a spacecraft
- ❖ Characterization of the commercial thermo-generator
- ❖ Characterization of the CEA batteries
- ❖ Conclusion and perspectives

**liten**

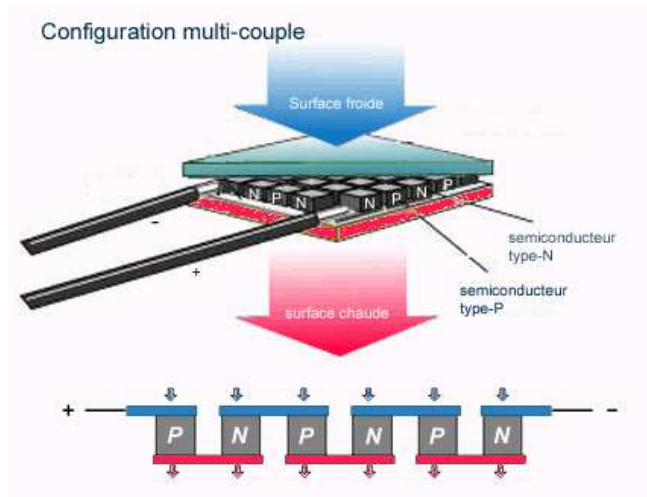
## Thermal mapping for a LEO spacecraft

	Average orbital T° (°C)	Radiative sink T° (°C)	Delta T° (°C)
lowest T° structure or equipment	-4	19	-23
highest T° structure or equipment	35	19	16
Power distribution unit	27	19	8
On board computer	25	19	6
DCU	24	19	5
Transponders	20	19	1
payload video unit	35	19	16
Batteries	10	19	-9
RIU	32	19	13
Wheels	26	19	7
TWTs	14 to 22	19	-5 to 3
External radiator -x,+Y T°	18	-69	<b>87</b>
External radiator +Y T°	18	-50	68
External radiator +x,+Y T°	9	-69	78
External radiator -x,-Y T°	19	-33	52
External radiator -Y T°	-5	-88	83
External radiator -x,-Y T°	15	-59	74

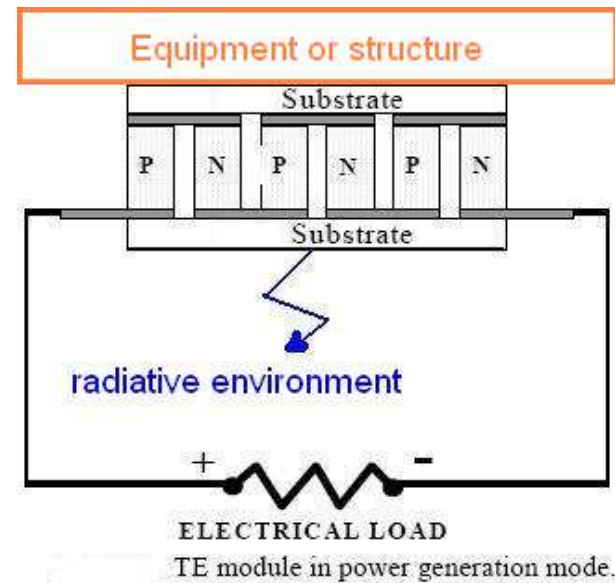


## Thermal gradient close to 23°C inside the spacecraft and 87°C outside the spacecraft

- Thermoelectric devices: direct energy conversion from thermal energy (T gradient) into electrical energy (Seebeck effect)

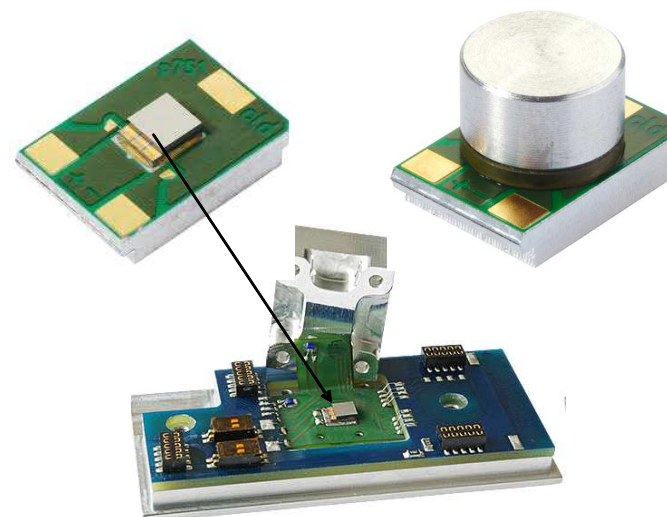
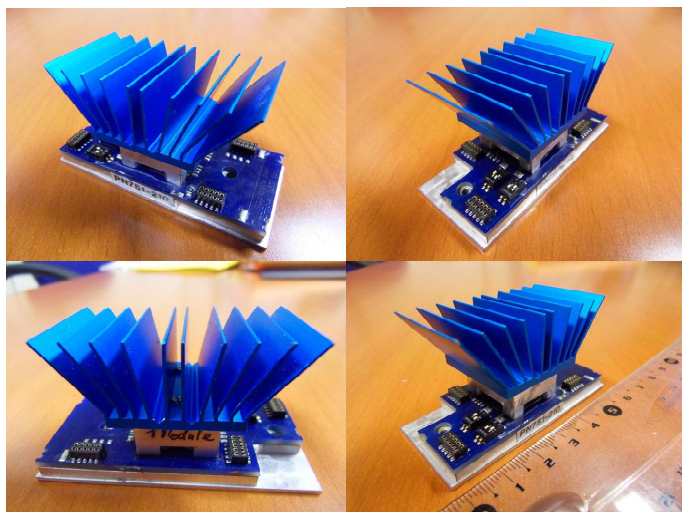


- The more realistic way to use the thermoelectric device



# Characterization of a commercial TEG

## □ The thermoelectric device : Micropelt

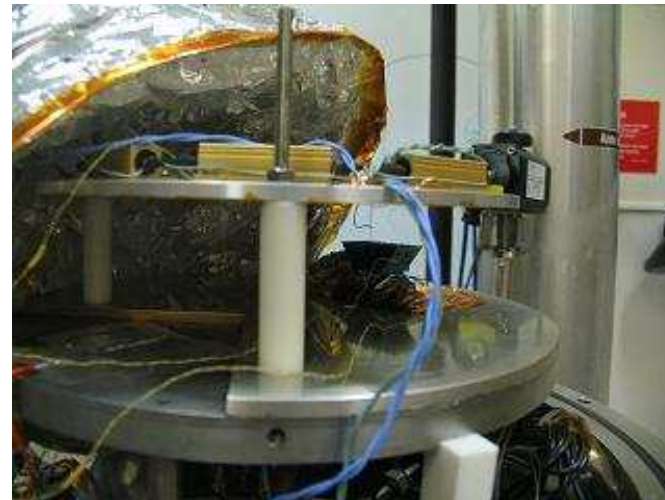


Dimensions	Base Total height incl. standard heat sink Weight	27 mm x 60 mm 33.5 mm 51g
Electrical characteristic of TGP 751 (neat)	Thermal resistance R Seebeck voltage Electrical resistance	18K / W 110mV / K ( $T_{amb} = 25^{\circ}\text{C}$ ) 240 to 350 $\Omega$ (typical 300 $\Omega$ )
Operating temperature	For TGP751 (neat) For the module - Substrate side - Ambient side	[-40 $^{\circ}\text{C}$ ; + 140 $^{\circ}\text{C}$ ]  Max. 105 $^{\circ}\text{C}$ Max. 85 $^{\circ}\text{C}$

$$\Delta T_{ap} = T_s(\text{ubstrate}) - T_a(\text{mbient})$$

## □ Evaluation of the electrical performances under vacuum ( $10^{-4}$ mBars)

- ✓ Conventional measurements at  $T_a = 20^\circ\text{C}$  and  $50^\circ\text{C}$ 
  - In open circuit
  - On calibrated resistances:  
⇒  $100\Omega$ ,  $250\Omega$ ,  $450\Omega$ ,  $750\Omega$  and  $900\Omega$
- ✓ In dynamic conditions
  - Controlled-computer power alimentation



$$\Delta T_{ap} = T_s(\text{ubstrate}) - T_a(\text{tmosphere})$$

## Experimental conditions and devices used in vacuum

- ✓  $T_s$  managed
  - ✓ By a thermo-regulated plate:  $-15^{\circ}\text{C} \rightarrow 30^{\circ}\text{C}$
  - ✓ Several thermocouples on the heat sink of the TEG, on the plate, ...
- ✓  $T_a$  fixed
  - ✓ By a plate equipped with heating resistances above the TEG
  - ✓ Regulation ensured by a "floating" thermocouple
- ✓ Thermal isolation by MLI coverage improved at each new set up
- ✓ For various  $\Delta T$ , measurements of :
  - ✓  $U$  = output voltage of the TEG
    - ✓ via **the plug-on module for the conventional measurements**
    - ✓ Via **the booster module for the test in "dynamic conditions"**
  - ✓  $R_b(\text{ottom})$  = resistance of the PT100 embedded on the basis of the TEG  $\rightarrow T_1$
  - ✓  $R_t(\text{op})$  = resistance of the PT100 embedded on the TEG top  $\rightarrow T_2$
  - ✓  $T_s$  = temperature of the heating or cooling plate by a thermocouple put above
  - ✓  $T_a$  = temperature of the atmosphere by a thermocouple
- ✓ Inversion of the heat flux is possible  $\rightarrow$  the polarity of the TEG reverses

Multimeter  
multichannels

$\rightarrow$  necessity to switch manually the TEG output polarity when the sign of  $\Delta T_{ap}$  changes  $\rightarrow$  not automatic !

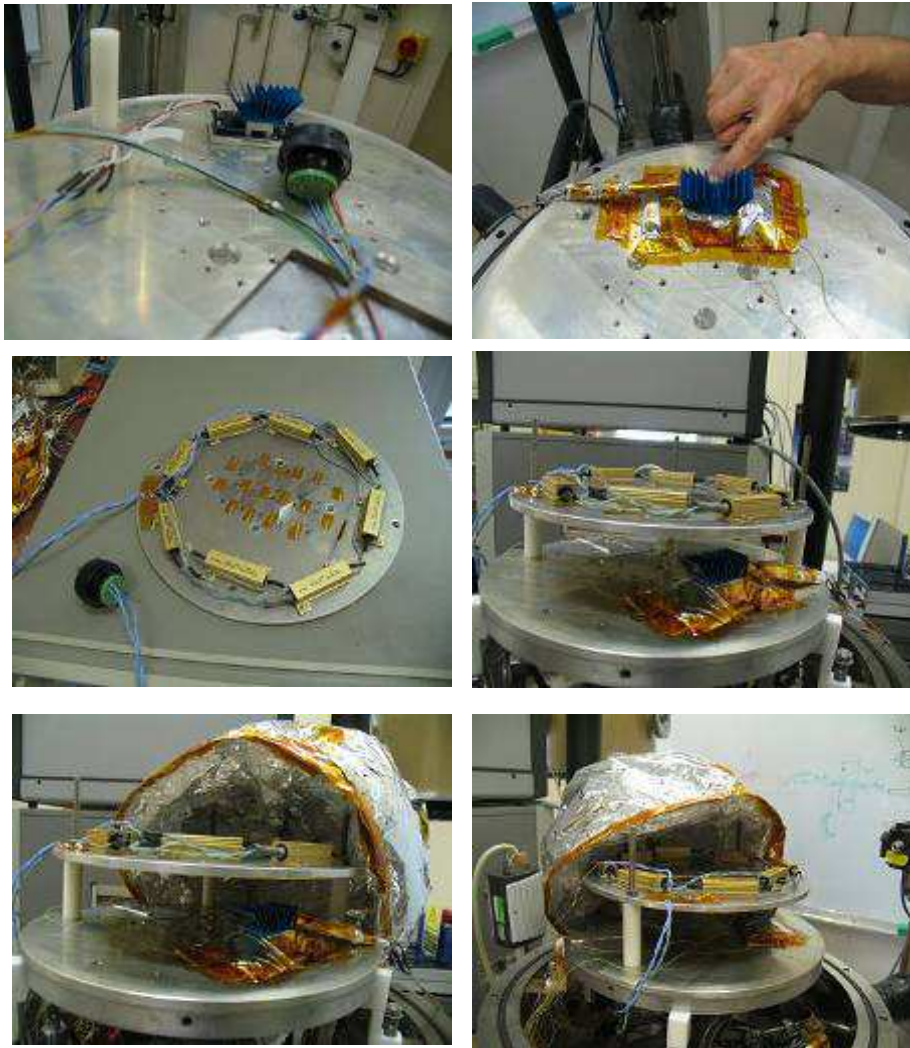


# Characterization of a commercial TEG

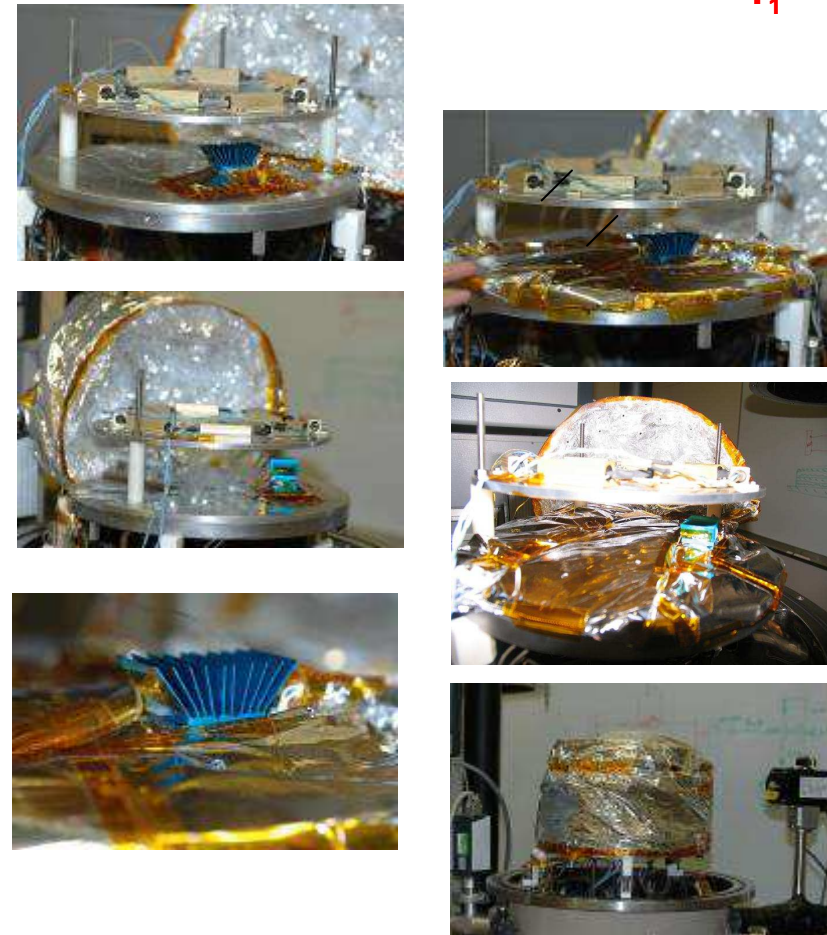
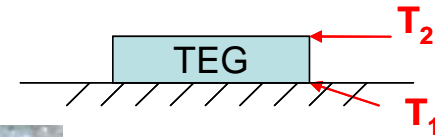
$$\Delta T_e = T_1 - T_2$$

Tests under vacuum (CNES),  $U = f(\Delta T_e)$

at 20°C



at 50°C





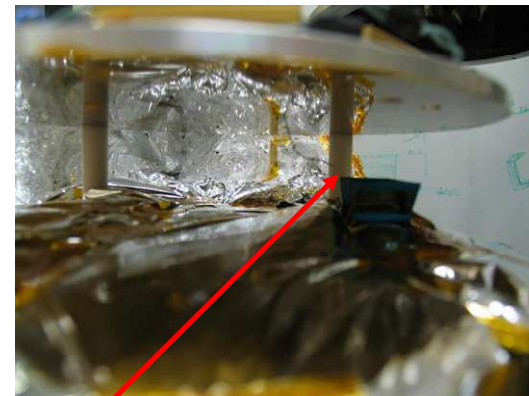
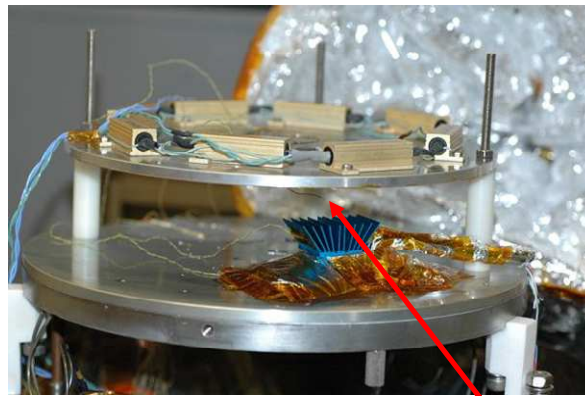
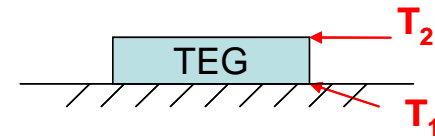
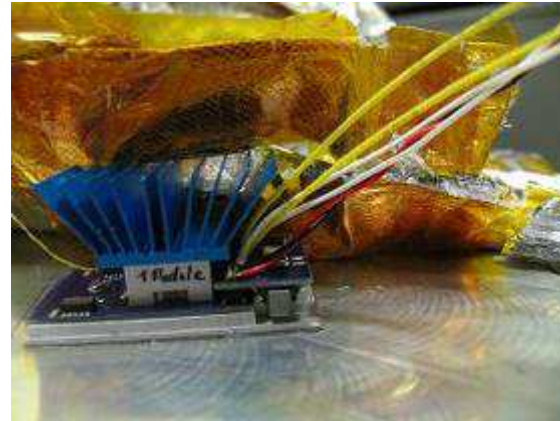
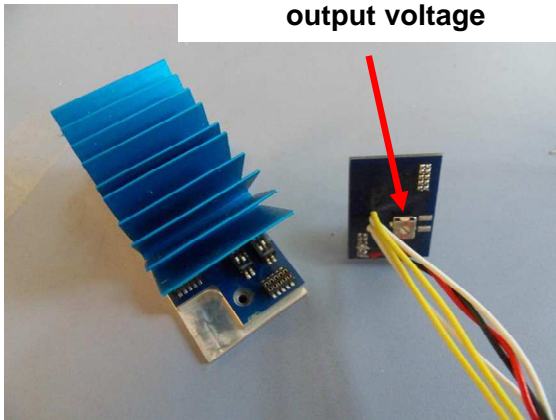
# Characterization of a commercial TEG

Tests under vacuum (CNES), dynamic conditions

$$\Delta T_e = T_1 - T_2$$

$T_a = 50^\circ\text{C}$ ,  $T_s = -10^\circ\text{C}$ ,  $P \sim 10^{-4}\text{mBars}$

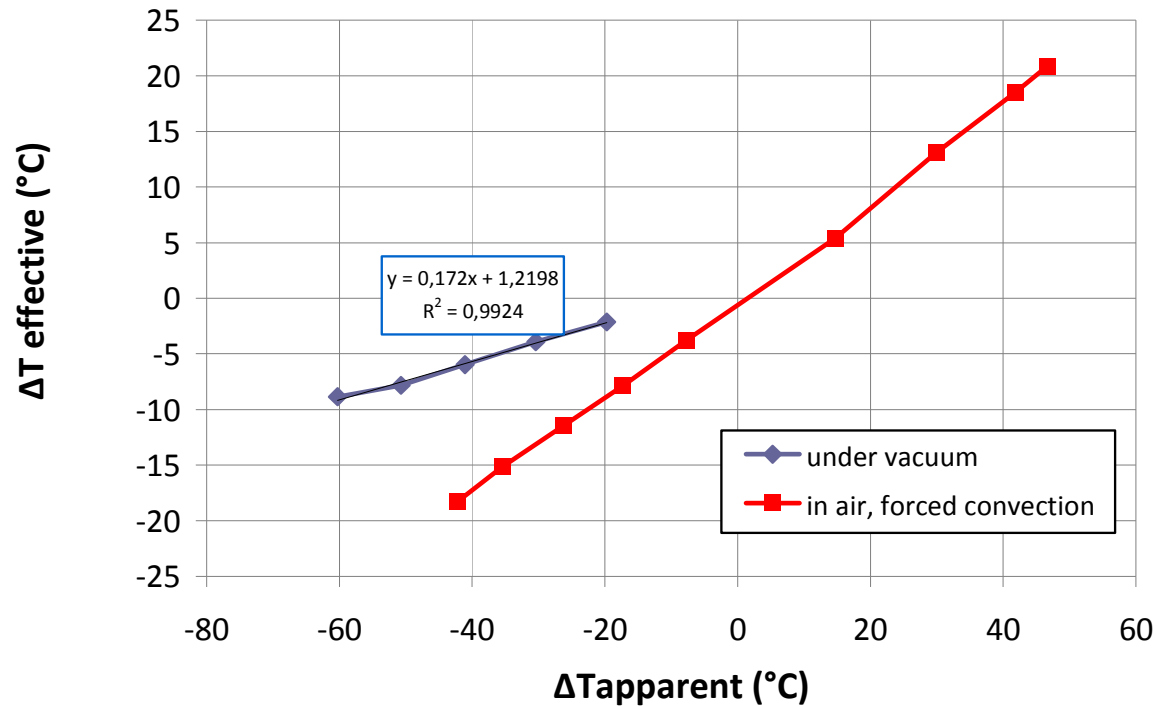
Potentiometer for adjusting the constant output voltage



Position of the floating thermocouple

Apparent and effective thermal gradient  $\Delta T_{app} = T_s - T_a$  and  $\Delta T_e = T_1 - T_2$

T = 50°C - Comparison under vacuum / in air with forced convection



➔ in vacuum, a very high apparent gradient is needed to reach equivalent  $\Delta T_e$ , compared to in air

For  $\Delta T_e = -8^\circ\text{C}$ ,  $T_s = 30^\circ\text{C}$  in air and  $-10^\circ\text{C}$  under vacuum at  $T_a = 50^\circ\text{C}$

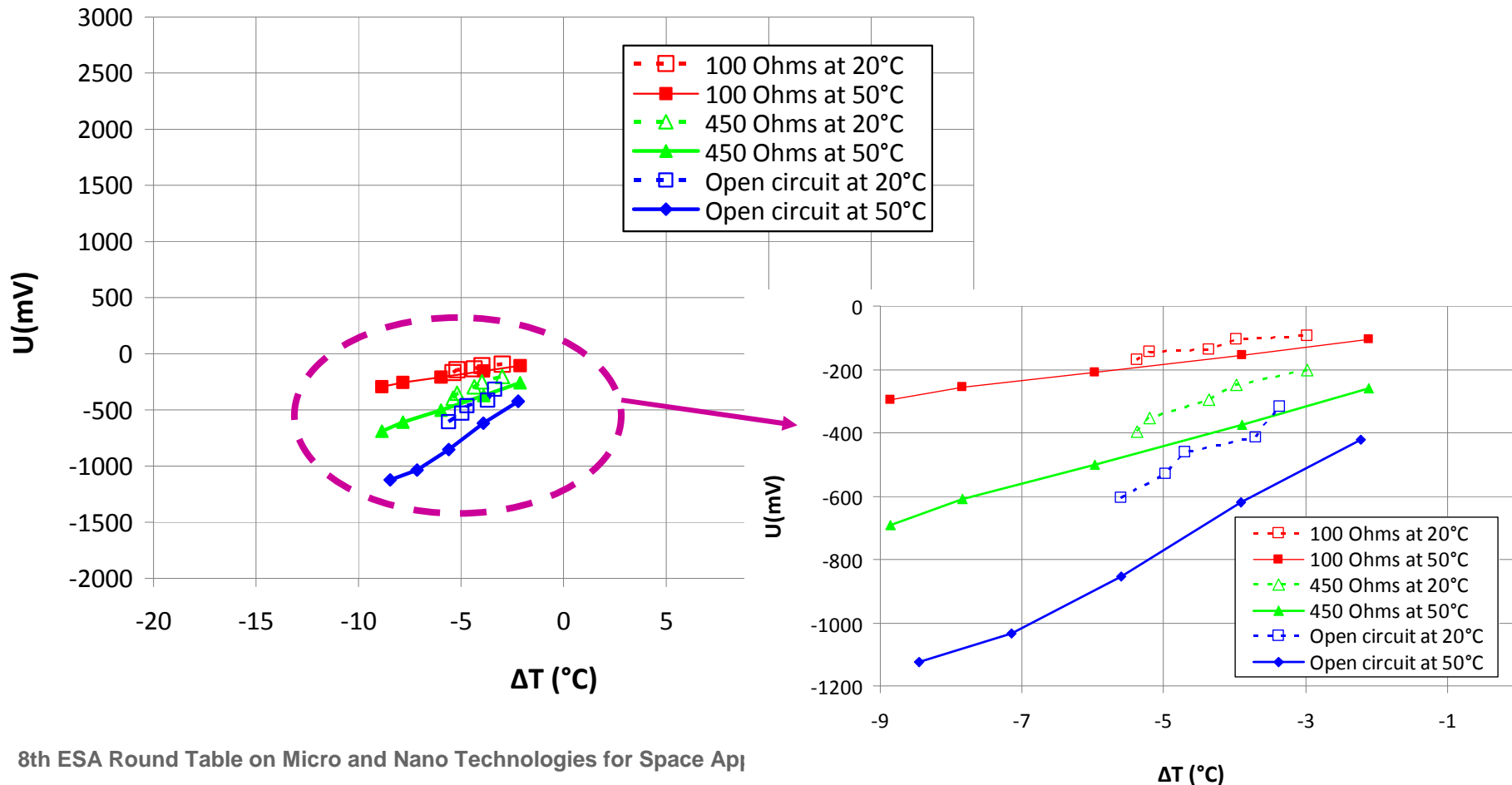
## Behaviour under vacuum ( $10^{-4}$ mBars) – static conditions

$$\Delta T = T1 - T2$$

- Restricted domain of  $\Delta T$ , so U compared to in the air
- similar performances at 20°C and 50°C

Under vacuum at 20°C and 50°C,  
In open circuit and on load,

Under vacuum ( $10^{-4}$  mBar) at 20°C and 50°C



## Behaviour under vacuum ( $10^{-4}$ mBars) – static conditions

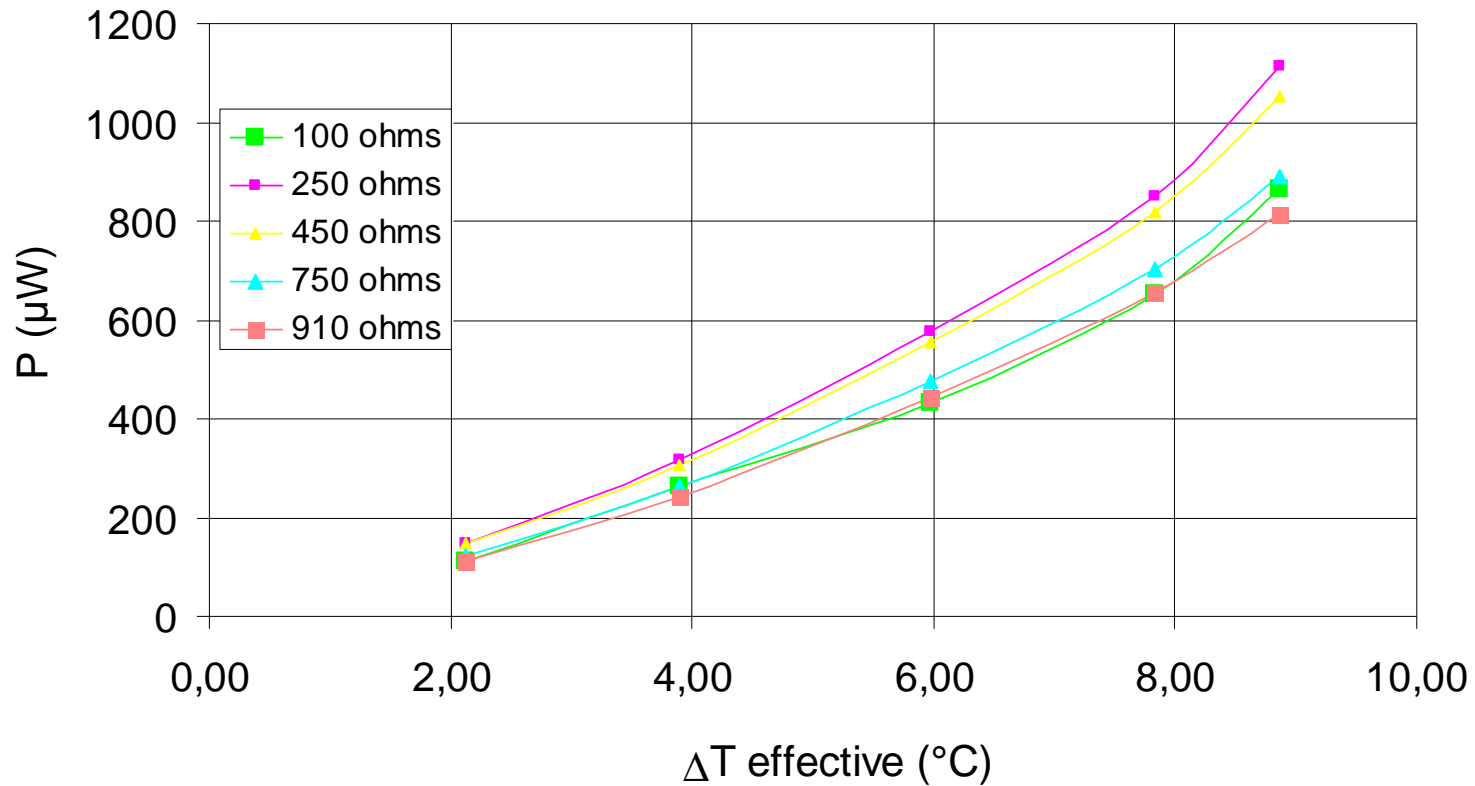
$$\Delta T = T1 - T2$$

→ thermoelectric power of 100mV/K (supplier data: 110mV/K)

→ Max output power developed at 50°C on  $R = 250 - 450\Omega$  with  $\Delta T = 9K \rightarrow 1.1mW$

Compatible with supplier data

TEG power at 50°C under vacuum



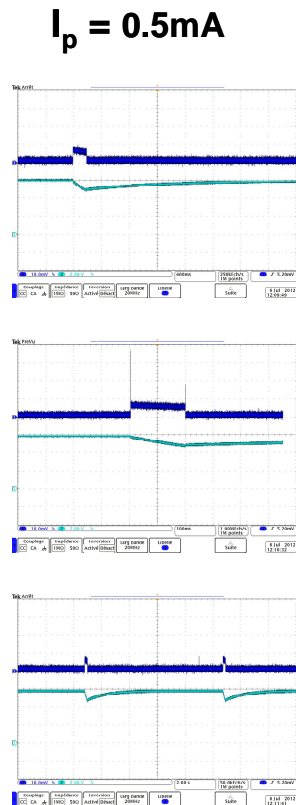
## Behaviour under vacuum ( $10^{-4}$ mBars) in "dynamic conditions"

$$\Delta T = T_1 - T_2$$

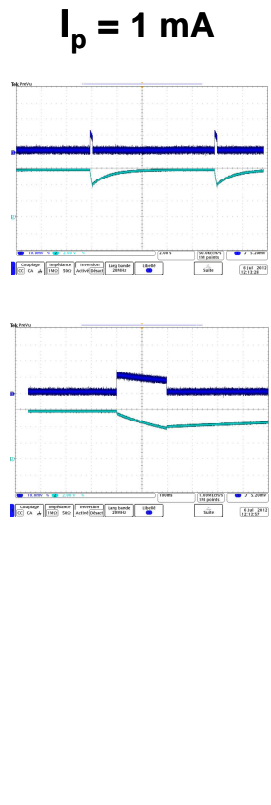
→ when  $I_p < 5\text{mA}$ , voltage drop but back to the base line of 6V

One current pulse per 10s

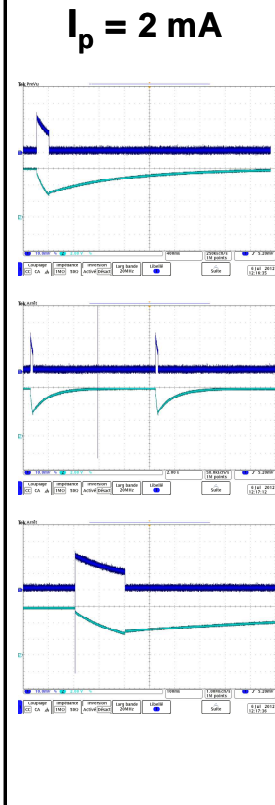
$\Delta t_{\text{pulse}} = 200\text{ms}$   
Base line: 6V



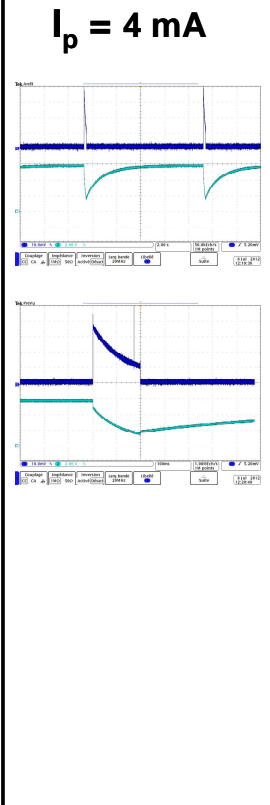
$\Delta U = -1\text{V}$   
 $\tau = 2\text{s}$



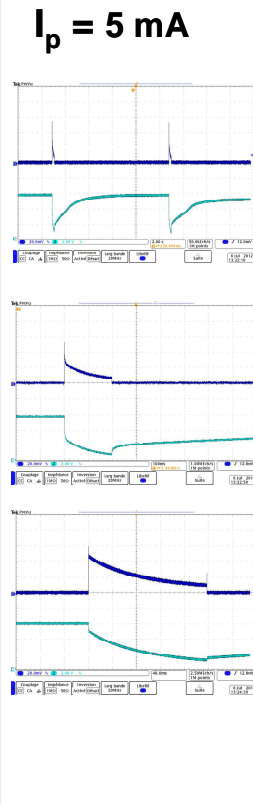
$\Delta U = -2\text{V}$   
 $\tau = 4\text{s}$



$\Delta U = -3\text{V}$   
 $\tau = 3.2\text{s}$



$\Delta U = -4\text{V}$   
 $\tau = 5\text{s}$



$\Delta U = -5\text{V}$   
 $\tau = 5\text{s}$

$T_a = 52.5^\circ\text{C}$   
 $T_s = -10.3^\circ\text{C}$   
 $\Delta T_{\text{ap}} = 62.8^\circ\text{C}$   
 $\Delta T = 9.8^\circ\text{C}$   
 $U_0 = 4.1\text{V}$



## Behaviour under vacuum (10<sup>-4</sup>mBars) in "dynamic conditions"

→ the TEG unit equipped with the booster module is capable to answer to pulses of ~ 10mW

$$\Delta T = T_1 - T_2$$

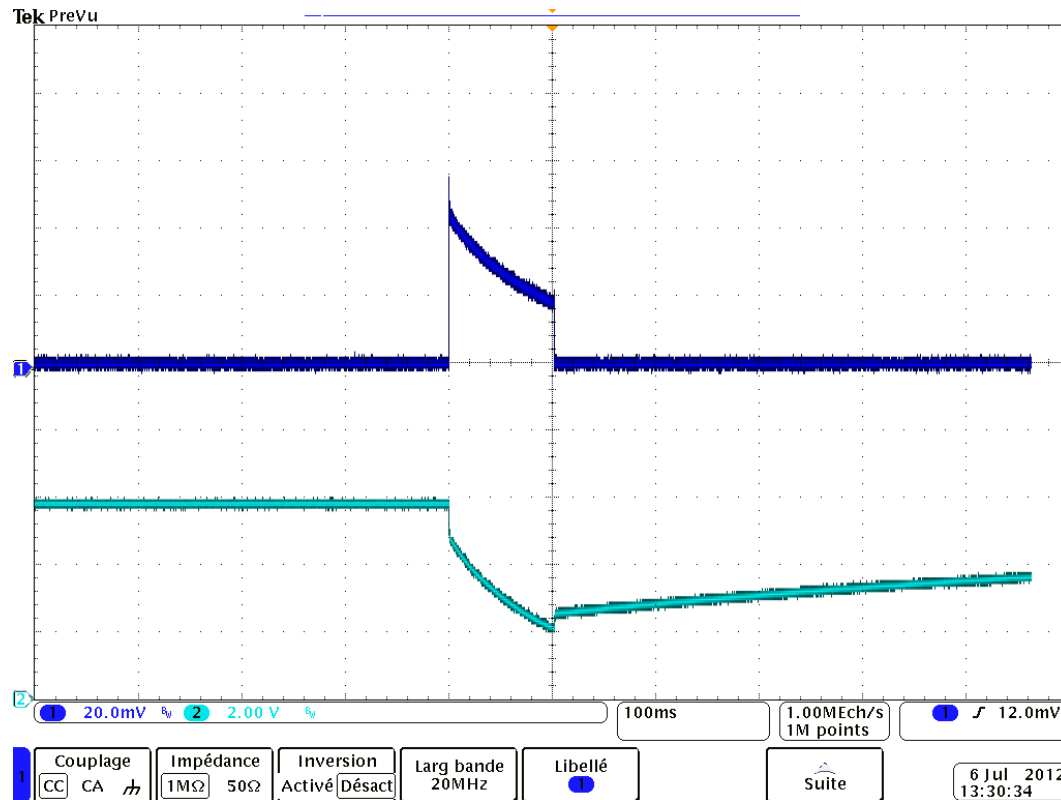
$$T_a = 52.5^\circ\text{C}$$

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$$\Delta T_{ap} = 62.8^\circ\text{C}$$

$$\Delta T = 9.8^\circ\text{C}$$

$$U_0 = 4.1\text{V}$$



One current pulse per 10s

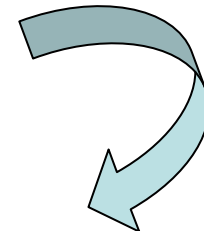
$$I_{\text{pulse}} = 5\text{mA}$$

$$\Delta t = 100\text{ms}$$

Base line: 6V

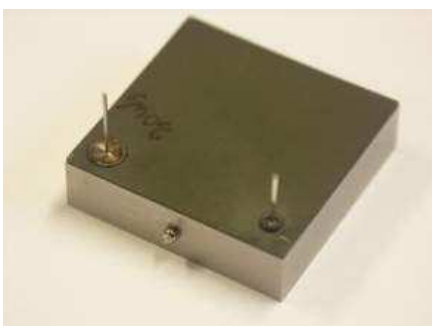
$$I_{\text{average}} \sim 3\text{mA}$$

$$U_{\text{average}} \sim 3.2\text{V}$$



**P ~10mW in pulse**

## □ Typical specifications of the CEA batteries



Technology	NCA / Graphite (CEA electrodes)
Nominal Capacity (mAh)	450
Nominal voltage (V)	3.6
Nominal Energy (Wh)	1.62
Cut-off Voltages (V)	3.0 - 4.0
Internal Resistance (mΩ)	175
Overall dimensions without pins (mm)	30 x 30 x 8.4
Volume (cm <sup>3</sup> )	7.6
Mass (g)	15.5
Volumetric Energy Density (Wh/L)	213
Gravimetric Energy Density (Wh/kg)	105

- ✓ Stored in cold chamber (-6°C) since September 2010
- ✓ 4 batteries selected from their nominal capacity and their level of hermeticity (for sustaining the test in vacuum)

↪ PR028, 036, 038 and 044

CEA battery	PR028	PR036	PR038	PR044
Nominal capacity (mAh)*	471	475	495	400

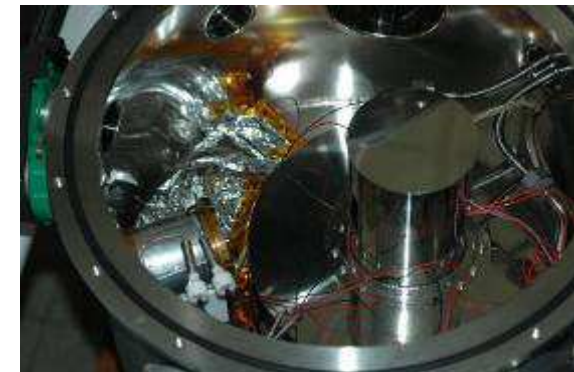
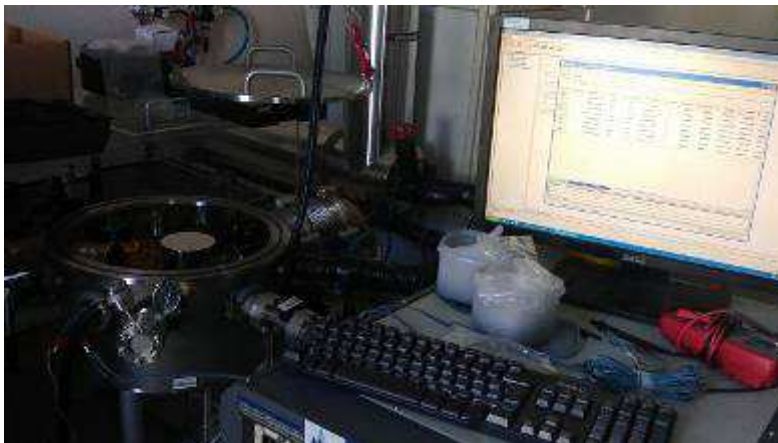
## Cycling tests under vacuum ( $2 \cdot 10^{-5}$ mBars) , Ta = 27°C (not regulated)

### □ The protocol of the cycling tests

- Cut off voltage : 3.0 and 4.0V
- Discharge and charge rates applied:
  - 3 cycles at C/10-D/10 ⇔ 60h
  - 4 cycles at C/5-D/5 ⇔ 40h
  - 4 cycles at C/2-D/2 ⇔ 16h
  - 5 cycles at C/2-D ⇔ 15h
  - 8 cycles at C/10-D/10 ⇔ 160h
- Measurement of the internal resistance during the cycling tests

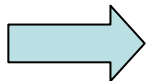
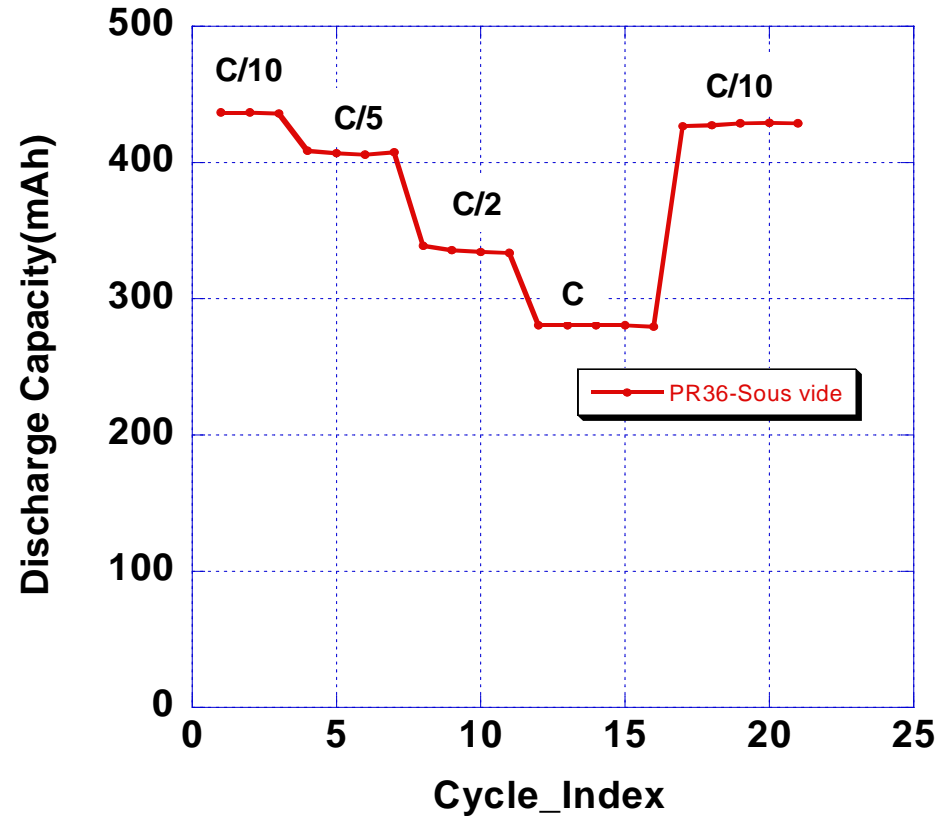
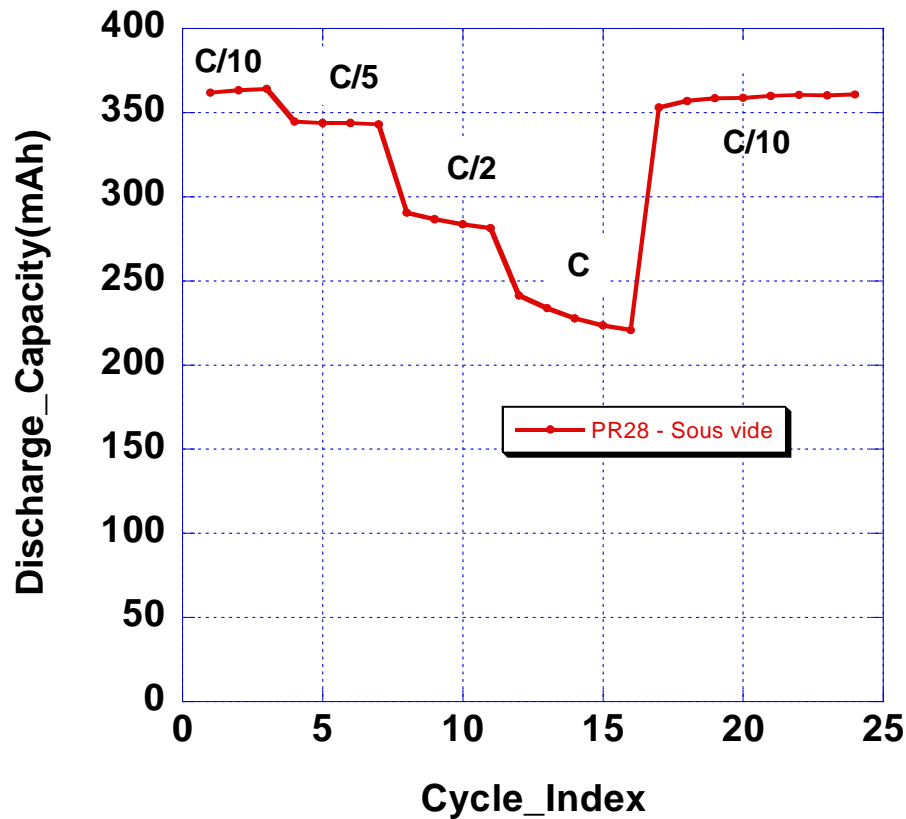
9 days

PR028 and 036



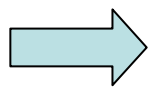
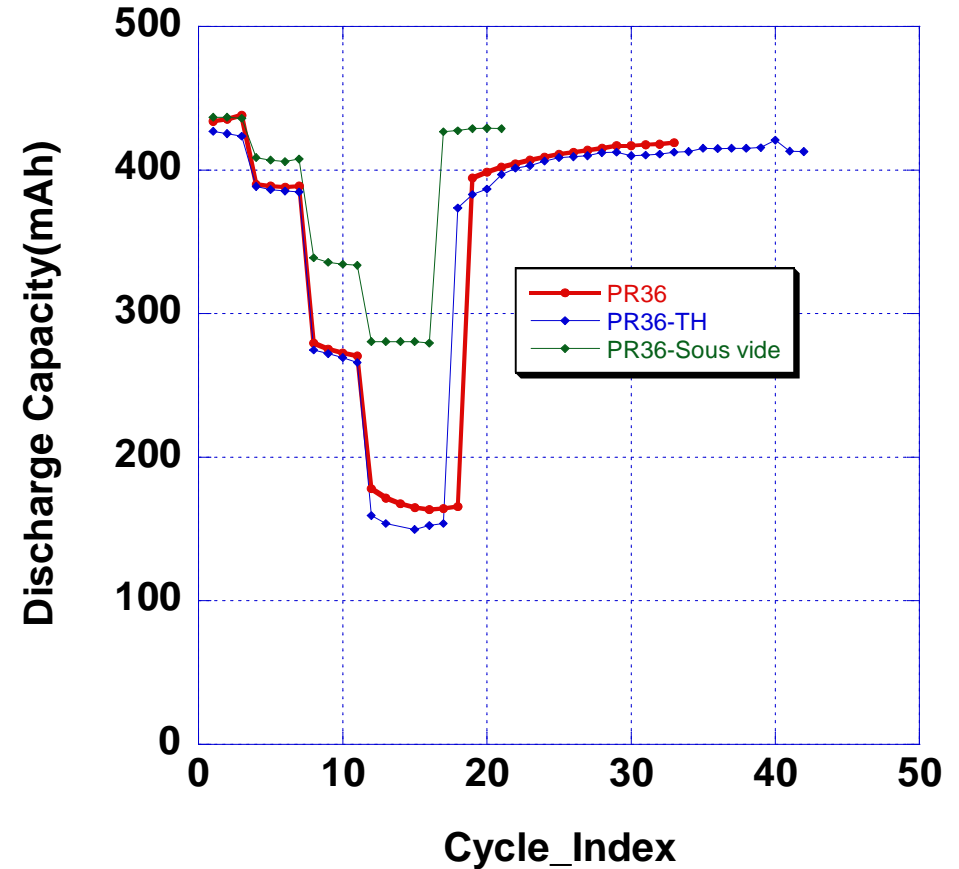
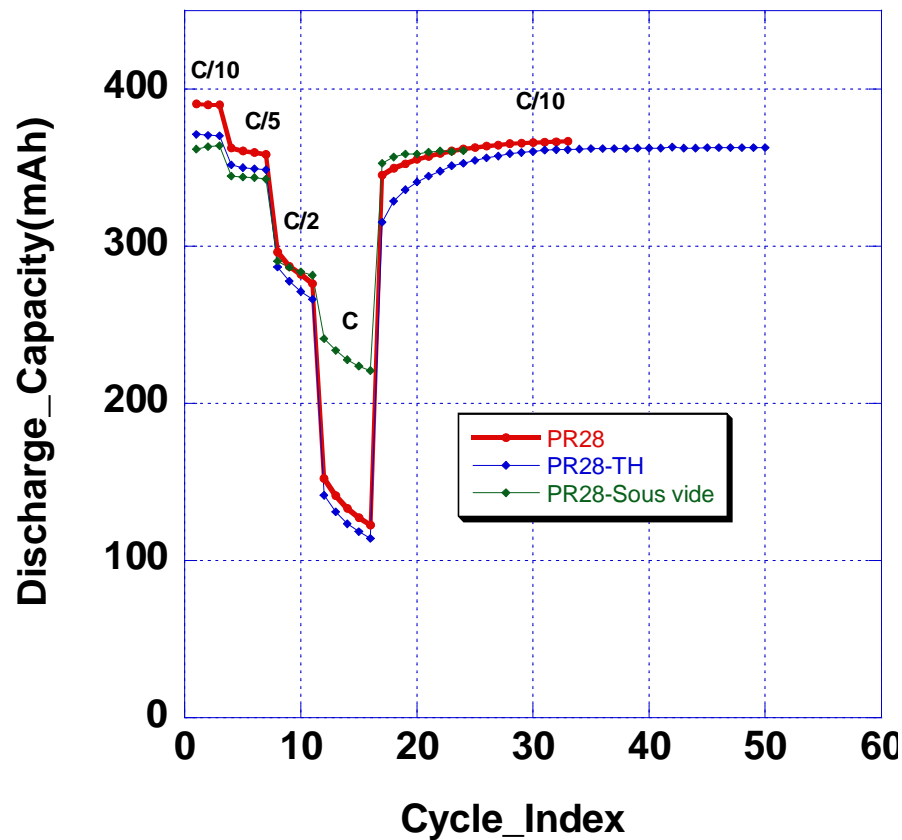
Protected in MLI coverage

## Cycling performances under vacuum



Relevant performances (in particular PR036)  
 → similar capacity at low rates up to C/2 rate  
 → higher capacity at higher rate

## Cycling performances in air before and after the thermal cycling and under vacuum



Promising performances under vacuum  
 → effect of internal pressure ?  
 → effect of  $T_{\text{ambient}} = 27^{\circ}\text{C}$  ?



**PR028 and 036 resistant towards vacuum and thermal conditions**



- Held of the TEG and the batteries to vacuum and thermal conditions **OK**
  - electrical performances similar, even better than in air for the batteries
  - TEG : 100mV/K, max 10mW in pulse (under  $\Delta T_{ap} = 60^{\circ}\text{C}$ )
  - Batteries : 370 to 420mAh at C/10 – Up to ~300mAh at C
  
- To improve the local gradient of the TEG → to improve the output power
  - To analyse a TEG unit without the heat sink → heat thermal model and association in series and/or in // of several TEG chips
  - To equip the thermal chamber (thermal screens) for a best management of the thermal environment

→ *test of evaluation in these new improved conditions*
  
- To propose another Li-ion technology with reduced working potential range
  - Based on another technology → 1.9V or 2.3V
  - Lower nominal capacity in rigid casing for vacuum (TVM, laser welded) → 20mAh? or less
  - Evaluation in cycling in space environment

→ *evaluation of the coupling small battery / TEG / Ultra Low Voltage Power Management*

**Special thanks to:**

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**Thank you for your attention**