



**ESA-CNES Final Presentation Days 2013** 

# Space Radiation-Induced Permanent and Single Event Effects in 4Mpixel Global Pipelined Shutter CMOS Image Sensors

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# CONTEXT OF THE STUDY

- CMOS IMAGER ARCHITECTURE
- PROTON-INDUCED PERMANENT EFFECTS
- HEAVY ION-INDUCED SINGLE EVENT EFFECTS
- CMOS IMAGER BEHAVIOR IN LOW EARTH ORBITS
- CONCLUSIONS AND FUTURE WORKS



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#### **CONTEXT OF THE STUDY**

- CMOS Image sensors reach very good performances
  - Deep submicron technology
  - Process dedicated to imaging (→ pinned photodiode)
- CIS integrates CMOS electronic functions
  - Specific architectures, ADC, Register, state machine
- CIS could be used for scientific applications
  - Space missions, nuclear physic,...
  - Need low and stable noise



- CIS are sensitive to space radiation environment
  - Permanent effects due to TID and D<sub>d</sub> are on going work since the last decade
  - No dedicated study concerns Single event effects



### **DEVICE ARCHITECTURE**

# Global architecture

- 2048 x 2048 pixels (5.5µm pitch)
- Snapshot running mode
  Same integration time for all pixels
- 180 frames/sec
- Global pipeline ADC are integrated (10 bit or 12 bit)
- Registers, State machine and Microcontroller are integrated



COPS

#### **DEVICE ARCHITECTURE**

#### Pixel architecture

- 8T pixel with in-pixel correlated double sampling
  - Reset noise is removed
  - Full well capacity and Fill Factor are reduced



COPS

#### Pixel architecture

- 8T pixel with in-pixel correlated double sampling
- Pinned photodiode using additional transfer gate



#### **PROTON-INDUCED PERMANENT EFFECTS**

- Protons induce TID and D<sub>d</sub> in CIS leading to dark current increase
  - TID induces charge trapped and interface states
    - TID effects are reduced using PPD  $\rightarrow$  SCR are recessed from oxide
  - D<sub>d</sub> induces bulk defects in silicon
    - Main issue in CIS using PPD → Hot pixels



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#### **PROTON-INDUCED PERMANENT EFFECTS**



- Only the dark current increases with doses
- LEO mission → I<sub>dark</sub> < 9000 e<sup>-</sup>/s (4800 pA/cm<sup>2</sup>)

# → Dark current increase due to Proton do not limit the CIS used for High Resolution Earth Imaging in LEO

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#### PROTON-INDUCED PERMANENT EFFECTS

#### **Proportionality** with D<sub>d</sub>

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```
\Delta I_{dark} = K_{dark} \cdot V_{dep} \cdot D_d
```



#### Mean dark current increase is higher than UDF prediction

- Proton-induced dark current is mainly attributed to D<sub>d</sub>
- TID effects are not negligible  $\rightarrow$  Mainly due to transfer gate bias during integration (See Goiffon, RADECS 2011)

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Single event effects occur in several parts of CIS integrating CMOS electronic functions





Single event effects occur in several parts of CIS integrating CMOS electronic functions





Single event effects occur in several parts of CIS integrating CMOS electronic functions



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Single event effects occur in several parts of CIS integrating CMOS electronic functions



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# Test bench setup

#### UCL High LET cocktail

LET

[MeV/mg/cm<sup>2</sup>]

3.3

6.4

15.9

40.4

67.7

Range

[µm Si]

59

45

40

39

37

lon

15 N 3+

<sup>20</sup> Ne <sup>4+</sup>

<sup>40</sup> Ar <sup>8+</sup>

<sup>84</sup> Kr <sup>17+</sup>

<sup>124</sup> Xe <sup>25+</sup>



Cnes

- Software developed
  - Software 1: Frame per frame writing
    - Count SEE (SET, SEU and SEL)

Tint	read+ stor	Tcontrol + Twrite	<u> </u>
1.3ms	82ms	96ms	

Software 2: 100 frame per 100 frame writing

Temporal degradation study (SET and SEU)



#### • Software 3: Read and Write in Registers

Count SEU in registers (Sampling time = 200ms)



#### SET analysis

- Software 1: Frame per frame writing
  - Comparison of each frame to the reference image
  - SET detection threshold 3σ (reference image)



#### **SET** analysis

- Software 1: Frame per frame writing
  - Comparison of each frame to the reference image
  - SET detection threshold  $3\sigma$  (reference image)
  - Number of impacted pixels increase with LET
    - $EPI = 5\mu m$

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Azote

Shallow trench isolation is used

Néon



Argon (3,3 MeV/mg/cm<sup>2</sup>) (6,4 MeV/mg/cm<sup>2</sup>) (15,9 MeV/mg/cm<sup>2</sup>) ~75 pixels

- SET analysis
- Cross section do not change significantly with LET



### SEU analysis

#### SEU in column ADCs: software 1

■ Detection of isolated pixel → ADC SEU : Numerous events



# SEU analysis

- SEU in Gain register: software 1
  - Detection of full image change → Gain SEU : Rare event



### SEU analysis

- SEU in registers: software 3
  - Register state: 0 and Register state:  $1 \rightarrow Same behavior$





# SET results

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- Heliosynchrone orbit 700kms
- Full 2048 x 2048 pixel array Device 30% ON

# SET / day					
Shielding	Wo eruption	Worse 24h	Worse 5mn		
3,7mm d'alu	1,42E+00	3,46E+02	1,29E+03		
6,0mm d'alu	7,98E-01	1,41E+02	5,20E+02		
12mm d'alu	4,05E-01	3,49E+01	1,26E+02		
24mm d'alu	2,33E-01	8,14E+00	2,87E+01		

# → The occurrence do not significantly impact the mission → Possibility to reduce the number of impacted pixel using DT

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# SEU results

- Heliosynchrone orbit 700kms
- Full 2048 x 2048 pixel array Device 30% ON

	# SEU gain	/ day			# SEU ADC	C/day	
	Wo		Worse		Wo		Worse
Shielding	eruption	Worse 24h	5mn	Shielding	eruption	Worse 24h	5mn
3,7mm d'alu	1,06E-05	2,42E-03	8,88E-03	3,7mm d'alu	2,57E+02	5,59E+04	2,07E+05
6,0mm d'alu	8,36E-06	8,10E-04	2,94E-03	6,0mm d'alu	1,58E+02	2,44E+04	8,91E+04
12mm d'alu	6,27E-06	1,74E-04	6,18E-04	12mm d'alu	8,91E+01	6,77E+03	2,43E+04
24mm d'alu	4,51E-06	4,37E-05	1,48E-04	24mm d'alu	5,28E+01	1,79E+03	6,32E+03

#### → The occurrence do not significantly impact the mission

→ Important number but only isolated pixel error: could be remove with QI software

# SEU results

- Heliosynchrone orbit 700kms
- Full 2048 x 2048 pixel array Device 30% ON

# SEU registrers (State 1) / day				
Shielding	Wo eruption	Worse 24h	Worse 5mn	
3,7mm d'alu	6,37E-02	1,53E+01	5,67E+01	
6,0mm d'alu	3,67E-02	6,39E+00	2,34E+01	
12mm d'alu	1,89E-02	1,62E+00	5,86E+00	
24mm d'alu	1,07E-02	3,83E-01	1,35E+00	

#### $\rightarrow$ All register do not impact the image

→The occurrence do not significantly impact the mission



# SEL results

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- Heliosynchrone orbit 700kms
- Full 2048 x 2048 pixel array Device 30% ON

# SEL / day				
Shielding	Wo eruption	Worse 24h	Worse 5mn	
3,7mm d'alu	4,23E-04	1,08E-01	4,02E-01	
6,0mm d'alu	2,57E-04	4,11E-02	1,51E-01	
12mm d'alu	1,47E-04	9,36E-03	3,36E-02	
24mm d'alu	9,18E-05	2,07E-03	7,23E-03	

#### $\rightarrow$ Anti-Latchup system is mandatory for the mission

 $\rightarrow$  Shut down the system is required during solar eruptions



### CONCLUSIONS

- Space radiations-induced permanent and single event effects are investigated in CIS fully integrated
- Permanent effects are mainly due to displacement damage
- Single event effects appear in CMOS electronic functions
  - SET in pixel array
  - SEU in column ADC and registers
  - SEL in numerical part
- CMOS Images Sensors fully integrated could be used for High Resolution Earth imaging in LEO
  - Permanent effects seem to do not limit the mission
  - Image correction could cancel isolated error pixel (due to ADC SEU)
  - SEU register are low and instrument availability stays in specification
  - Anti-Latchup system is required

#### **FUTURE WORKS**

- Guideline redaction could help the single event tests
  - Fixe the methodology to test the devices
- Complementary tests on this device
  - Single event study using proton irradiations
  - Test deeper EPI (12µm) → Observe the increase of SET impact size
  - Test other EPI resistivity → Observe SEL behavior
- Single event tests are on going work in other CIS device

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- Devices from other foundry
- Devices using RadHard library

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