

Miniaturised sensor development programme for studies of the space plasma and radiation environment

D. Kataria, A. Smith, R. Bedington

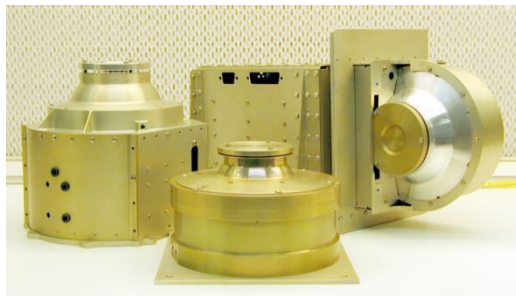
Mullard Space Science Laboratory
Dept. of Space and Climate Physics
University College London

Plan

- Introduction
- Miniaturisation programme
- Technology toolbox
- Ongoing Activities and Timeline
- Applications and Summary

Research at MSSL

- Department of Space and Climate Physics, University College London
- More than 40 years in space research and instrumentation
- Eight research groups supported by specialist engineers conduct our scientific research
- Currently 16 instruments in space
 - Earth, Mars, Venus, Saturn

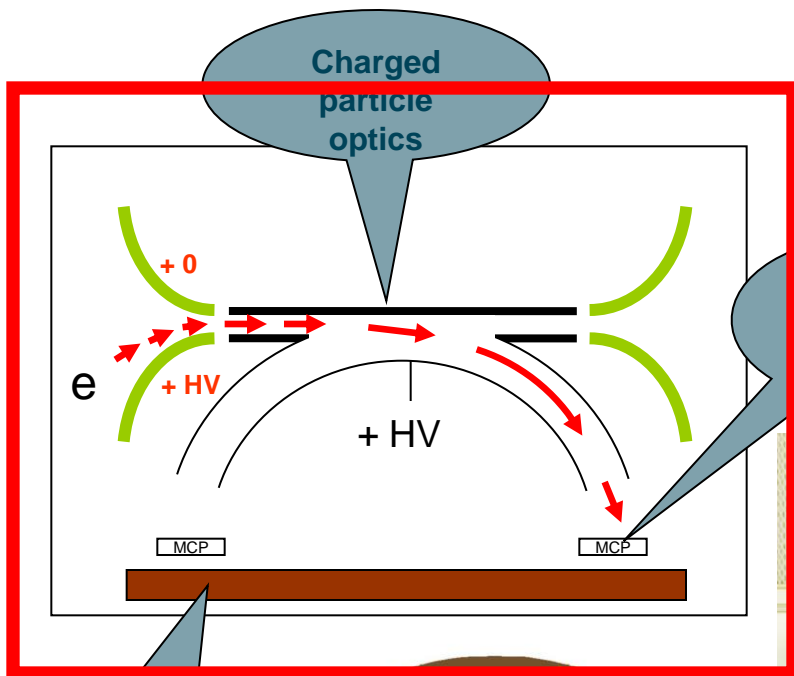


Missions – Past, Present and Future

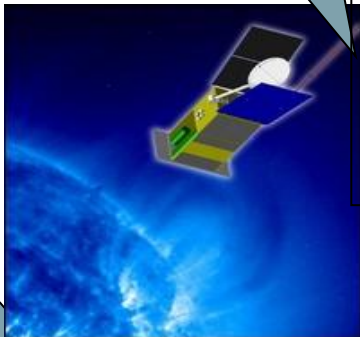
- Giotto
- Cluster II
- Cassini
- Mars Express
- Venus Express
- Solar B
- Solar Orbiter
- Cosmic Visions
- Cubesats/NanoSats
- Herschel
- STEREO
- SOHO
- XMM-Newton
- Swift
- Integral
- GAIA
- JWST - Nirspec
- Astrosat

Plasma Instrumentation

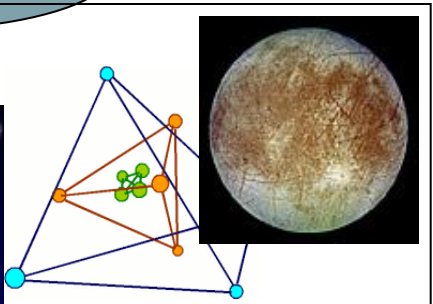
R&D Activities and Capabilities



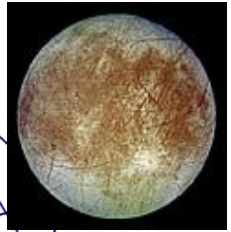
Mission Studies



Solar Orbiter

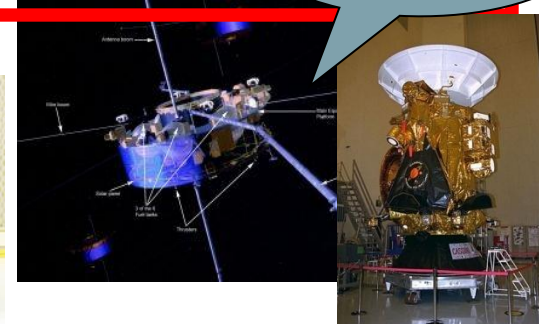
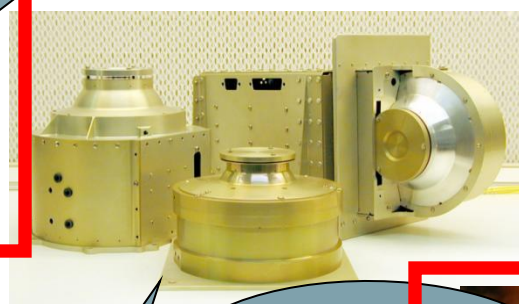


X-scale, EJSM

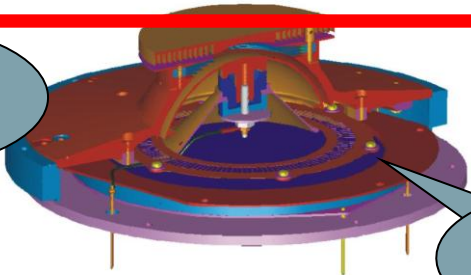


Detector research

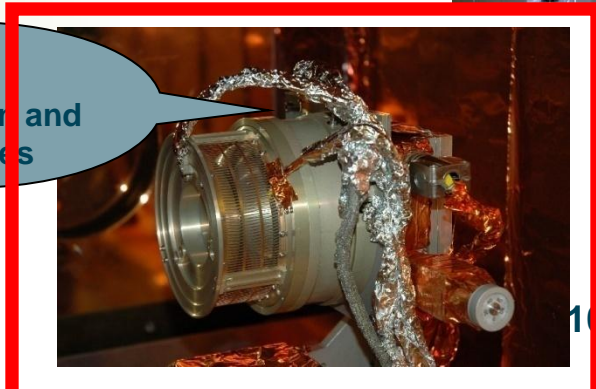
Post Launch support



Electronic subsystems



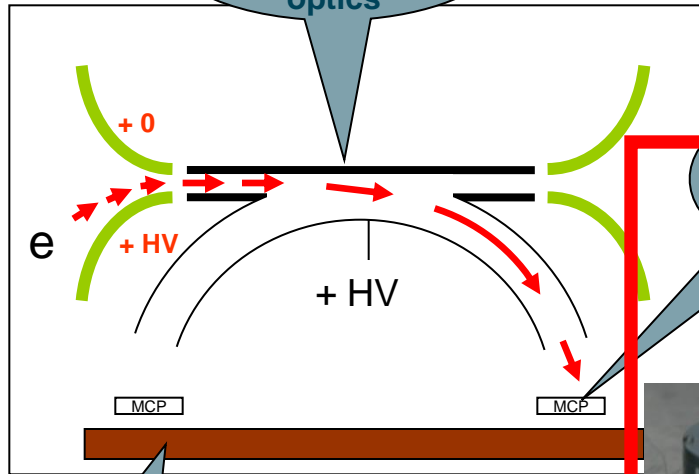
Test, Calibration and Facilities



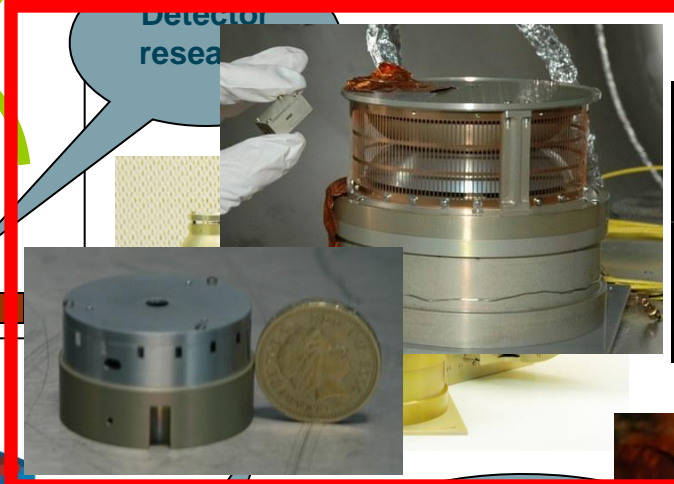
Design and build

Plasma Instrumentation R&D Activities and Capabilities

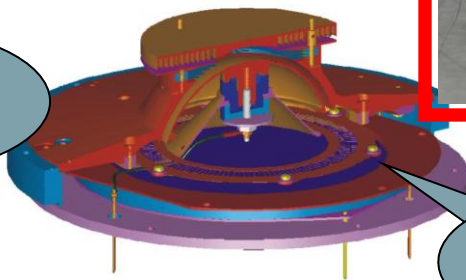
Charged particle optics



Detector research

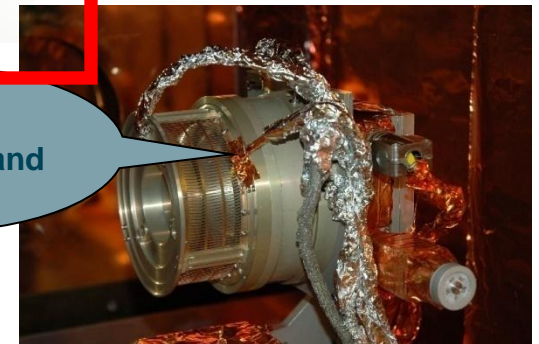


Electronic subsystems

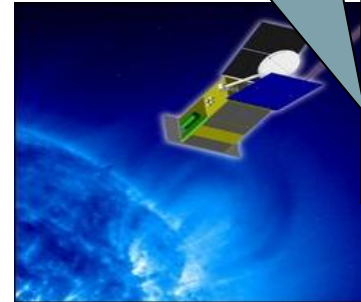


Design and build

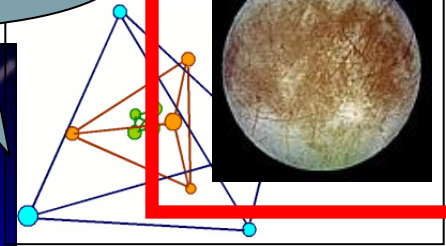
Test, Calibration and Facilities



Mission Studies

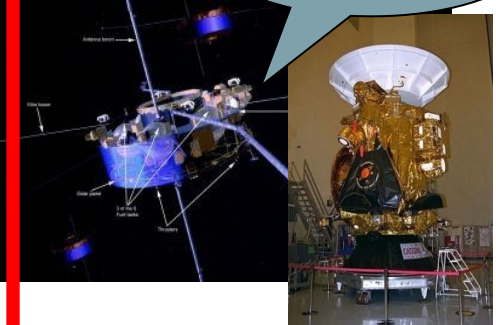


Solar Orbiter



X-scale, EJSM

Post Launch support



The Future is Small

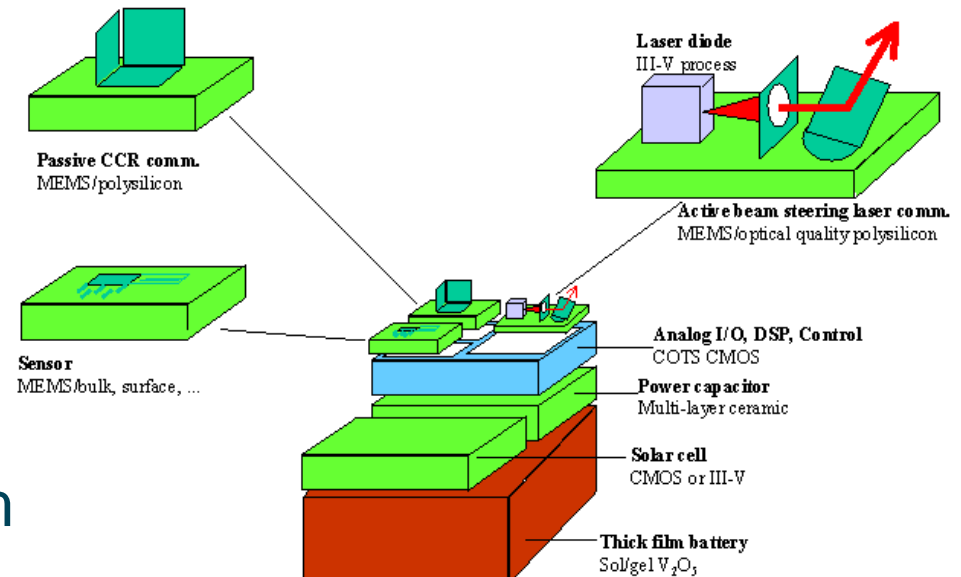
MEMS-based Analyser

- Aims
 - Low resource analyser development using MEMS-based (Micro-Electro-Mechanical Systems) fabrication techniques
 - Provide generic technologies suitable for creating highly integrated “matchbox” sized analyser systems: small, low resource, more capable
- Ideal for Space Applications

Future is small

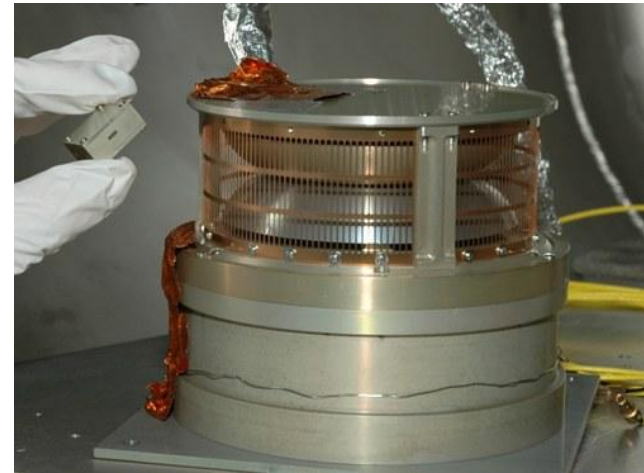
Generic Technology “Toolbox”

- Smart Dust
- Sensor system
- Processing Electronics
- Active/Conditioning system
- Wireless system
- Power system
- Integration and Packaging

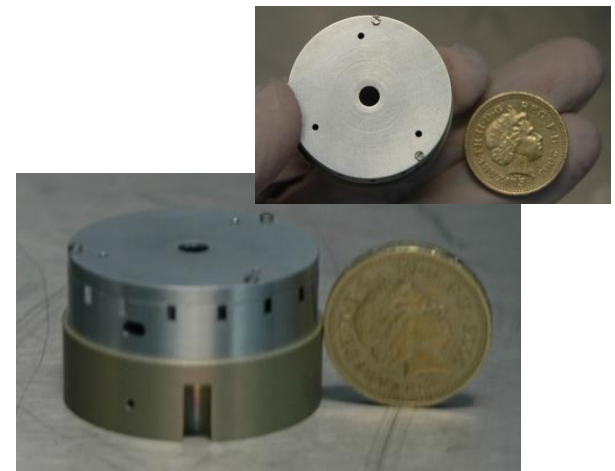


Sensor System Techniques

- Fabrication
 - Conventional, silicon, laser
 - Liga
- Analysis
 - Electrostatic, energy only
- Detection
 - MCPs, silicon
- Readout
 - Pixellated, position sensitive



Phase 1: Proof-of-concept Analyser
along with the IPA analyser



Phase 2: ChaPS: Technology
Testbed Prototype

Performance and Resources

Estimated

	Cassini ELS	Cluster LEEA	IPA	ChaPS
Particles detected	Electrons	Electrons	Electrons	Electrons and Ions
K Factor	6.14	6.14	6.14	~ 8
Geometric Factor	$8 \times 10^{-4} \text{ cm}^2 \text{ sr}$	$\sim 1.6 \times 10^{-4} \text{ cm}^2 \text{ sr}$	$\sim 1.2 \times 10^{-4} \text{ cm}^2 \text{ sr}$	$\sim 1 \times 10^{-4} \text{ cm}^2 \text{ sr}$
Energy Resolution	0.167	0.127	0.127	~0.22
Energy Acceptance	0.59eV – 26.4KeV	0.59eV – 26.4KeV	0.59eV – 26.4KeV	Few eV to 20KeV
Angular Resolution	$5.24^\circ \times 20^\circ$	$2.79^\circ \times 15^\circ$	$2.79^\circ \times 22.5^\circ$	$\sim 17^\circ \times \sim 21^\circ$
Angular Acceptance	$5.24^\circ \times 160^\circ$	$2.79^\circ \times 180^\circ$	$2.79^\circ \times 360^\circ$	$\sim 17^\circ \times 360^\circ$
Mass	Analyser head ~0.2 kg Total 1.4 kg	Analyser head ~0.2 kg Total 1.9 kg	Analyser head ~0.2 kg Total 0.9 kg	Analyser head ~ 0.01 kg Estimated Total < 0.3 kg
Power	2 W	2 W	1.5 W	0.4-0.5 W

Electronics and Conditioning Element

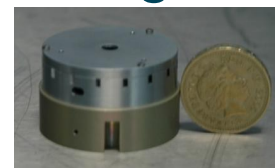
- **Electronics:**
 - Power supplies: HV and LV key focus
 - Miniaturised HVs under development for Solar Orbiter/Future missions
 - Choice of analyser geometry is key
 - FPGA: Strong heritage, rapidly evolving technology
 - Readout ASIC: Number of solutions currently available or under development within Europe
 - Integrated ASIC essential for low resource final solution
- **Conditioning: Ionizer for neutral particle studies**
 - Key requirement for number of MSSL/UCL interests
 - Low resource technique under development. Proof-of-concept testing in progress.

Current Timeline

- **Analyser Prototype**

- Continuing tests with proof-of-concept analyser
- Technology testbed prototype assembly and testing

- Full instrument – analyser, detector, electronics
- TRL 5/6 by early 2011

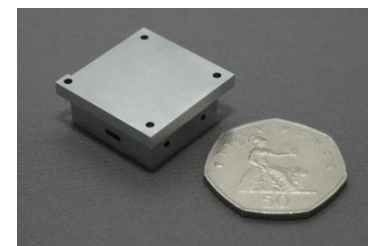
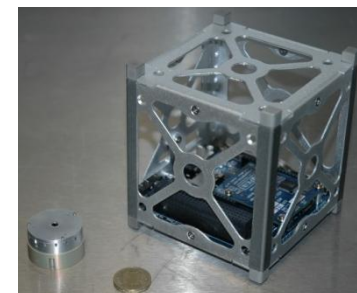


- Potential launch opportunities - 2011/2012 timeframe:

- UKube – UK CubeSat programme. Details being finalised. Competitive AO. Launch late 2011/early 2012
- Other potential opportunities

- Design modifications

- Performance “tailoring” and flight compatibility



Ongoing Developments

Activities and Timeline

- Ionizer development and testing
 - TRL 4 by Dec 2010. Integration and testing 2011
- Integrated MEMS Electrostatic Analyser and Silicon detector study
 - Large dynamic range, few eV to several MeV
 - TRL 5/6 in the 2011/2012 time frame
- Next Phase
 - Wireless functionality – Post doc joining in Sept/Oct.
 - Integration and Packaging - “Instrument-on-chip” solution
 - Feasibility study for self powered solution

Potential Missions and Applications

- Space Science
 - Cosmic Vision candidates – EJSM, M3
- Space Weather applications
- Upper Atmosphere Research
 - CubeSat constellation under study
- Spacecraft Subsystems
- Education and Outreach

Summary

- Strong heritage, aggressive miniaturisation
- “Toolbox” of generic technologies being developed
- Miniaturised proof-of-concept analyser undergoing testing and prototype testbed under development
- Continuing R&D Programme
 - Neutral particles - TRL 4 by end of 2010
 - Highly Integrated Sensor Systems – 2012 time frame
- Enabling technology, range of applications
 - Space Weather Constellations, CubeSat/NanoSat, Planetary

Acknowledgements

- UK STFC Funded programme
- STFC Case funded studentship
 - Industrial partner: Astrium UK, industrial supervisor Steven Eckersley
- ESA Innovation Triangle Initiative study

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