

“MEMS Capacitive Accelerometers for Future Launchers”

SEPTEMBER 2010

Outline

THEON SENSORS

Time line - Status

Activity Description

Work Logic/Approach

Module Architecture

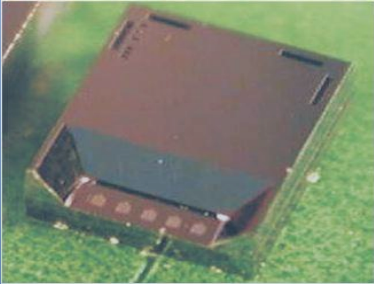
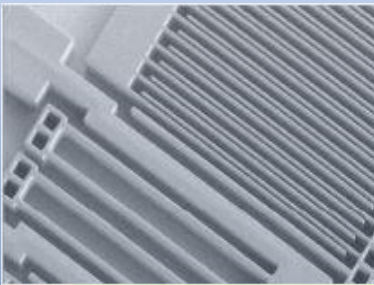
Accelerometer Component

MEMS

CMOS

Next Steps

- **Company**
- **Brief History**
- **Activity Description**
- **Development Approach**
- **Accelerometer Module Architecture**
- **Core Technology Presentation – MEMS + CMOS**
- **Next Steps**



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Process technology

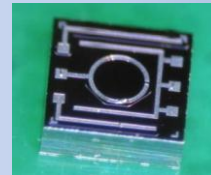
MEMS Surface Micromachining for Capacitive Inertial Sensors (XFAB)

MEMS Fusion Bonding for Capacitive Pressure Sensors (THEON-XFAB)

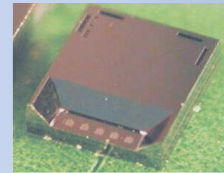
MEMS Bulk Micromachining for Resistive Flow Sensors (ISIT Fraunhofer)

0.18um Mixed-Signal CMOS (XFAB)

MEMS



Capacitive Pressure Sensor

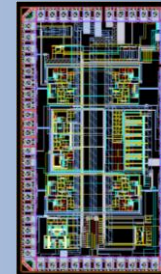


Capacitive Accelerometer

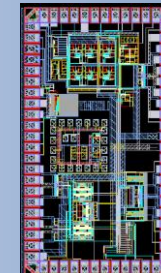


Flow

ASIC



Capacitive Interface



Resistive Interface

Peripherals

Digital Logic
(Processor for Calibration, Compensation, etc)

Power Supply
(DC-DC converters, Voltage regulators, etc)

COMMS
(SPI, I2C, custom)

Other
(Instrumentation Amplifiers, Balancers, Bandgaps, etc.)

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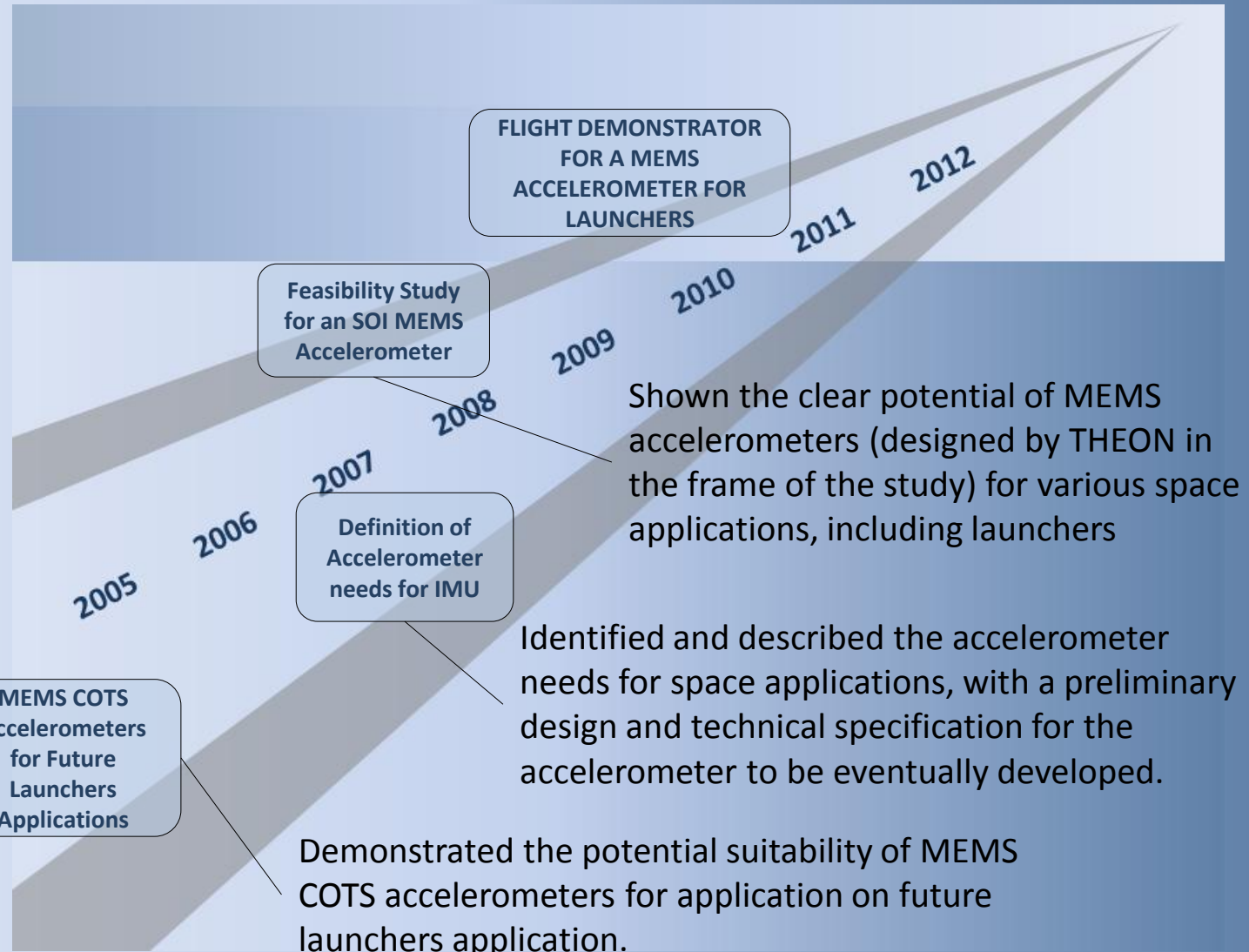
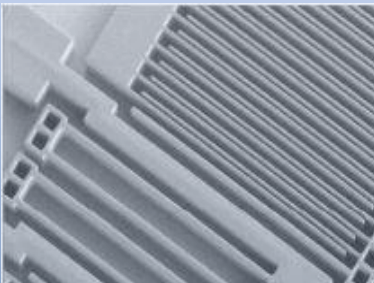
Module Architecture

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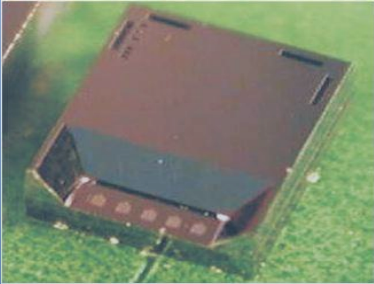
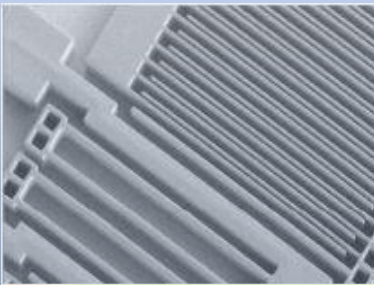
Next Steps

FLIGHT DEMONSTRATOR FOR A MEMS ACCELEROMETER FOR LAUNCHERS

- Started in NOV 2009, 38 months, 2 phases
- Industrial scheme: ASTRIUM ST Prime, THEON sub-contractor, ESA end-user
- Development of flight model for future ARIANE

MAIN ASSUMPTIONS

- Flight model, TRL= 8
- Be usable in any part of the launcher and at any phase during the flight mode including “upper stages” (low weight and small size objectives)
- Compliant with A5 launcher environments
- Compliant with all the mission requirements applied to the TM system
- Friendly adaptable to the current TM subsystem

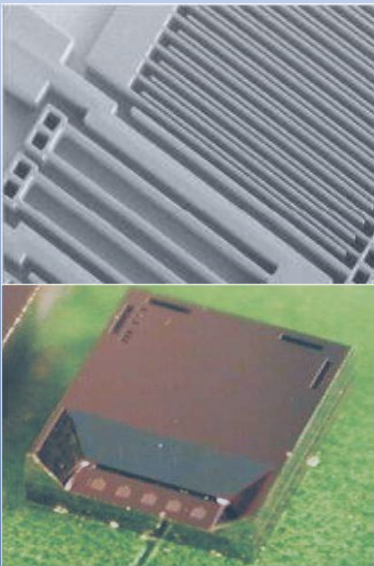
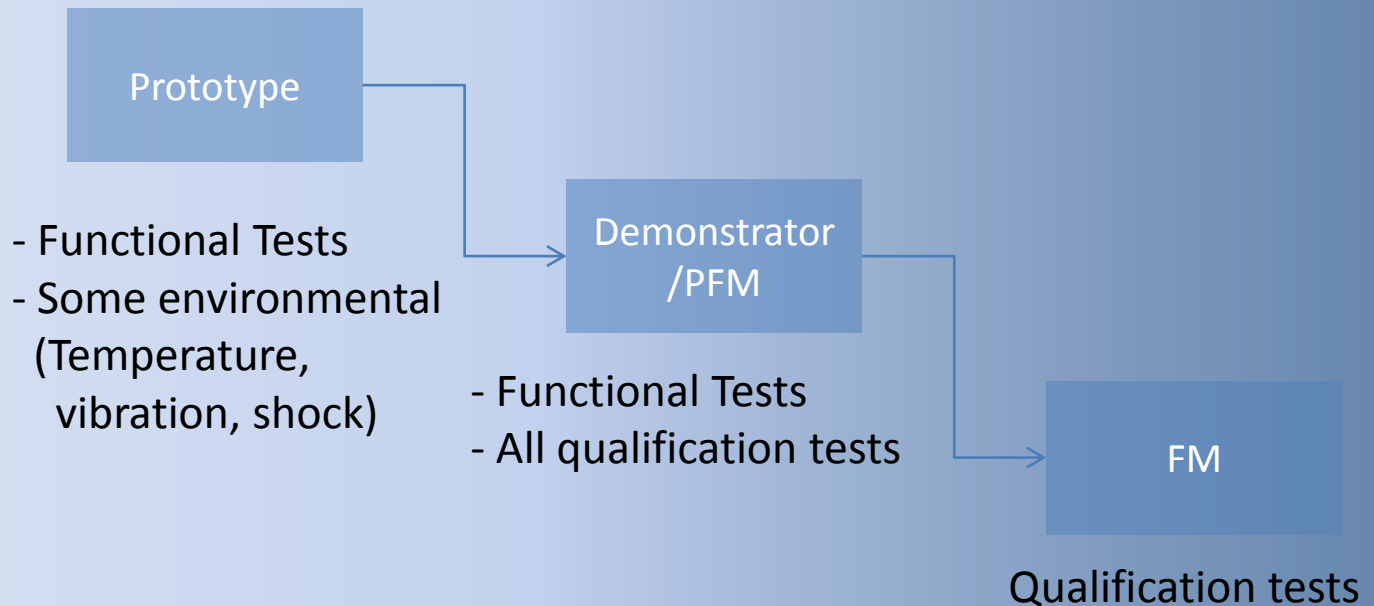


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Introduction of two novel features, for the Ariane 5 sensors:

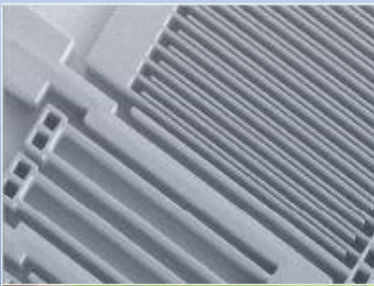
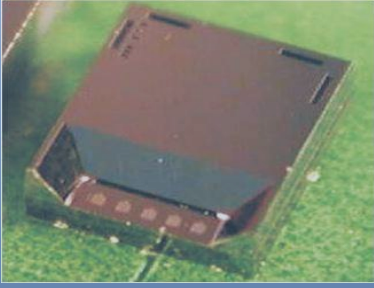
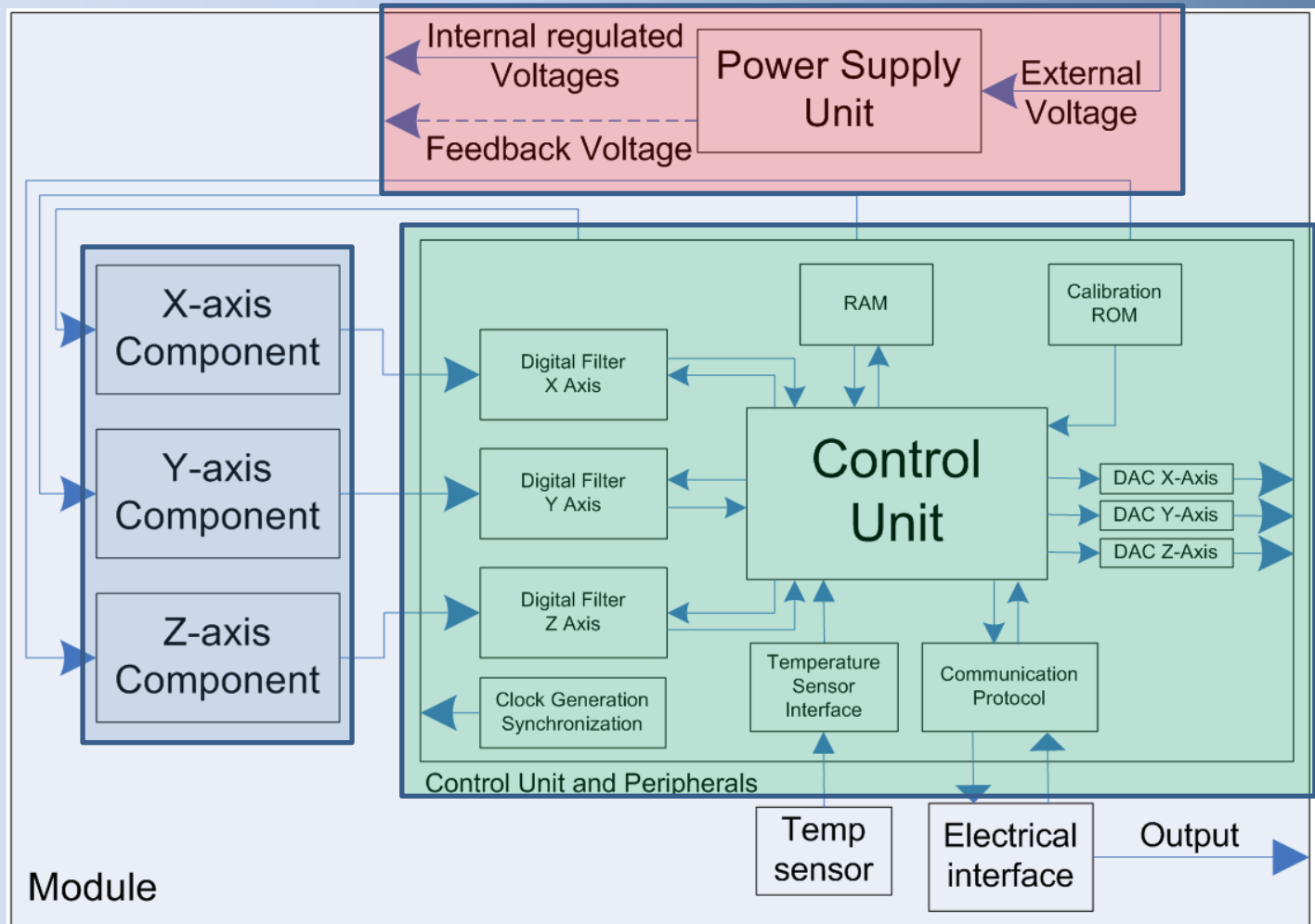
1. the MEMS technology
2. the digital interface

PROTO FLIGHT MODEL APPROACH

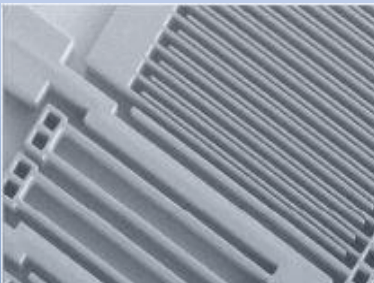
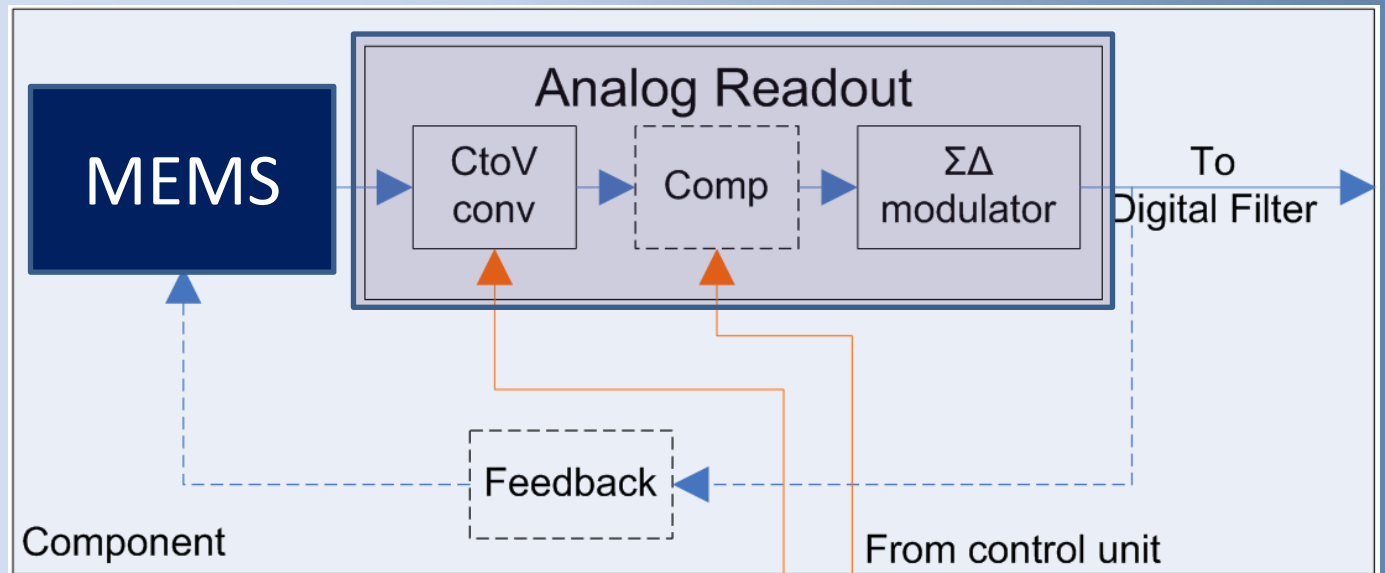


Module Architecture

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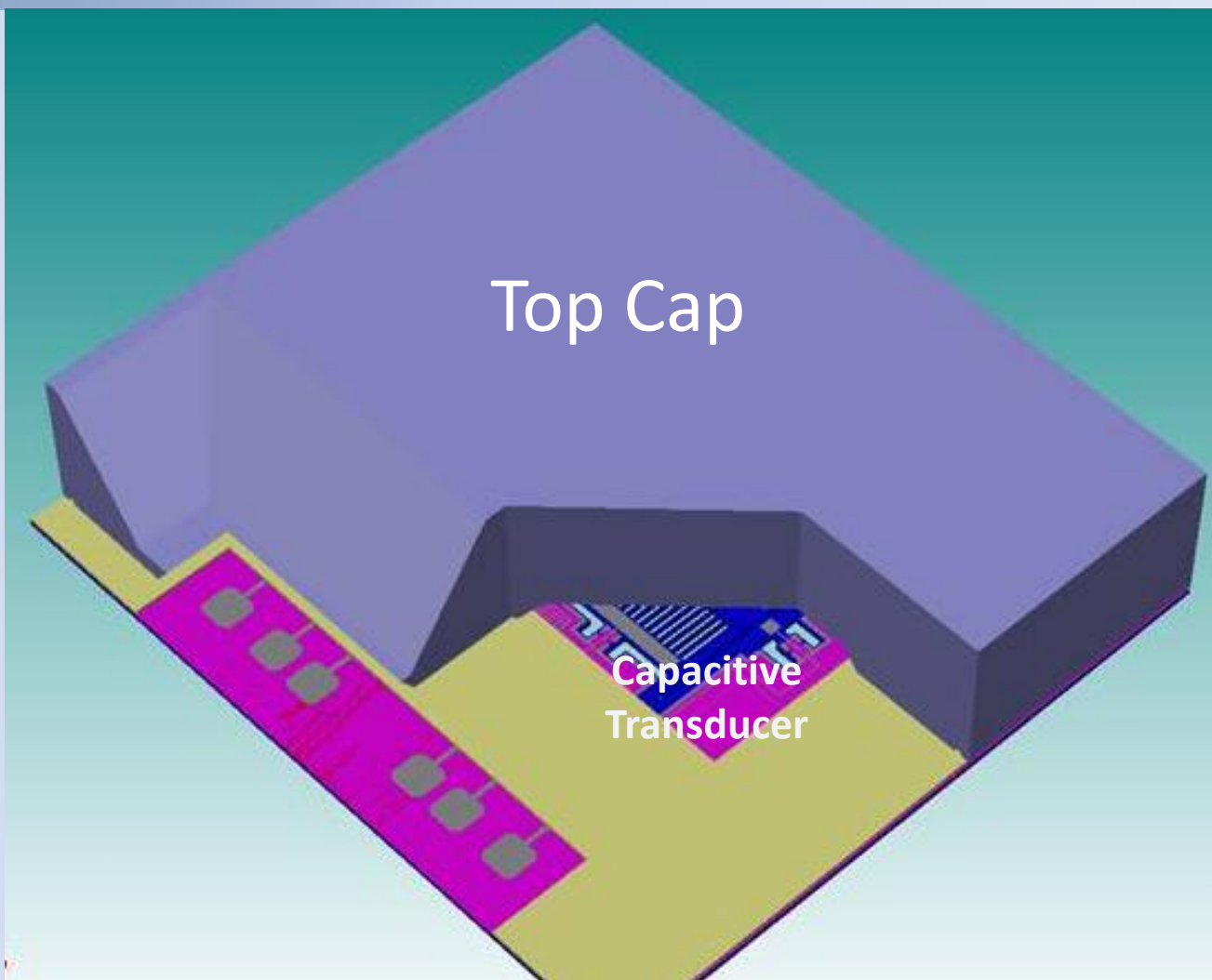
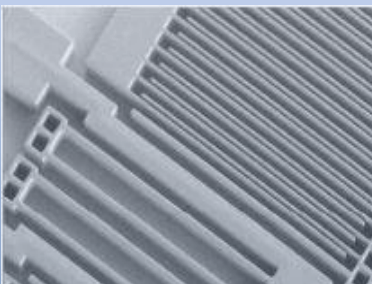
Module Architecture

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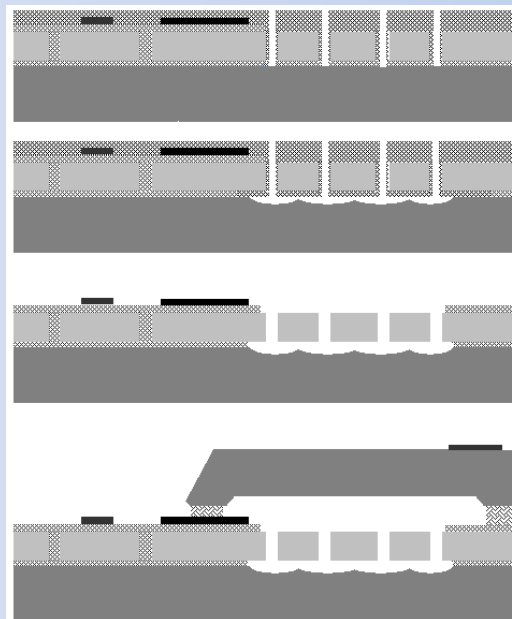
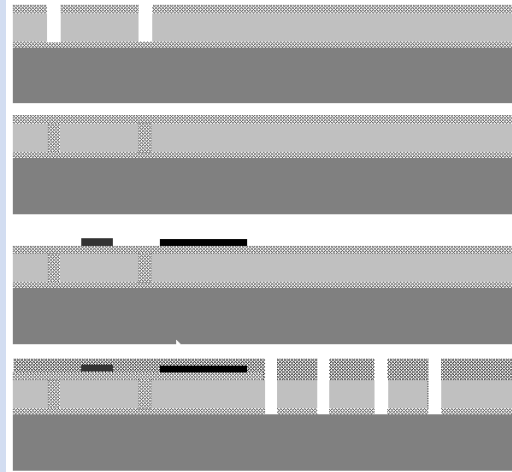
Module Architecture

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Next Steps



1. Etch isolation trenches

2. Fill isolation trenches, deposit intermediate isolator

3. Deposit and pattern metal layer

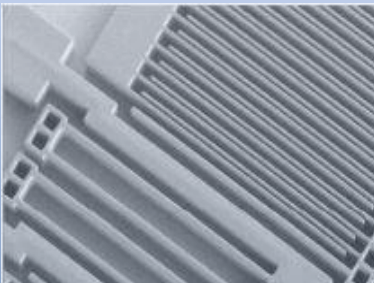
4. DRIE mechanical structure

5. Passivate sidewalls,
then open bottom of trenches

6. Isotropic etch to release mechanical structure
(structure width determines which parts are freed)

7. Strip etch mask, sidewall passivation and buried oxide
leaving single-crystal mechanical structure

8. Cap sensor wafer by glass frit wafer bonding



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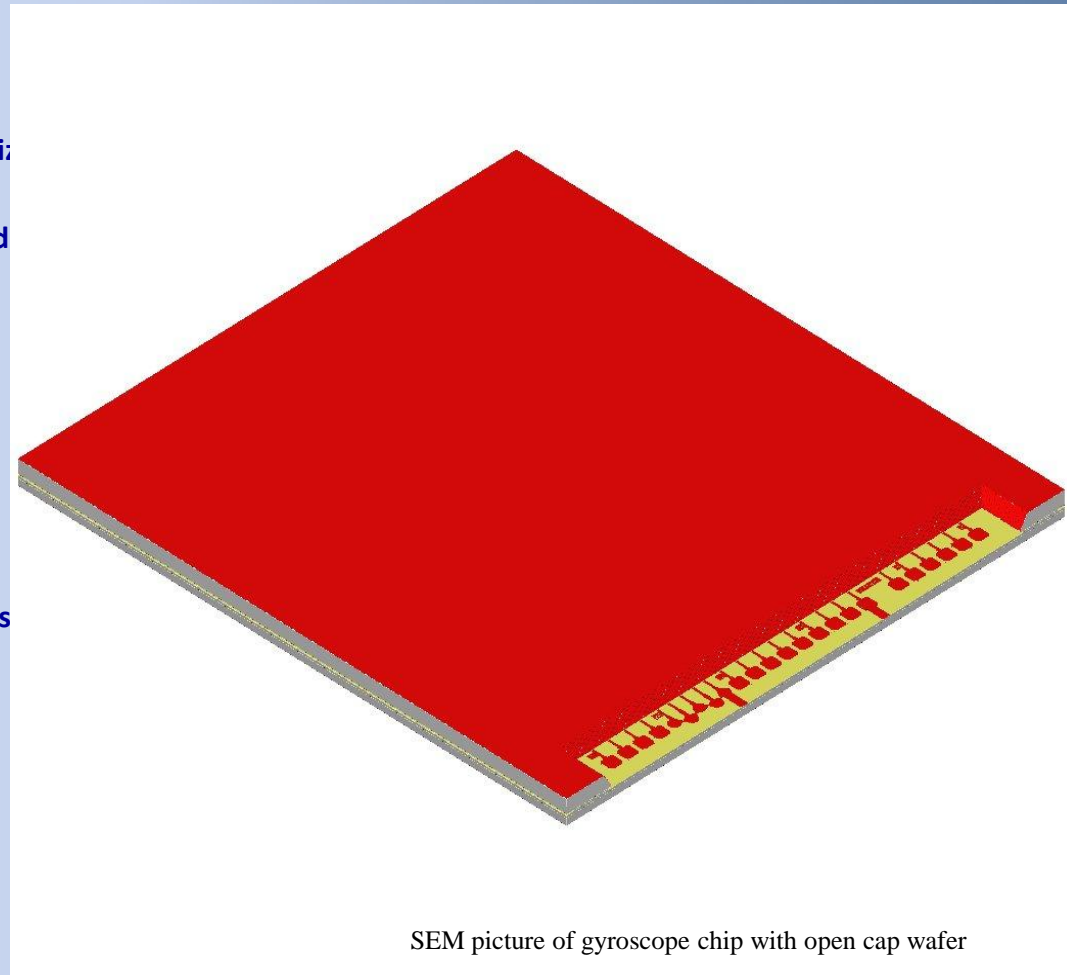
Accelerometer Component

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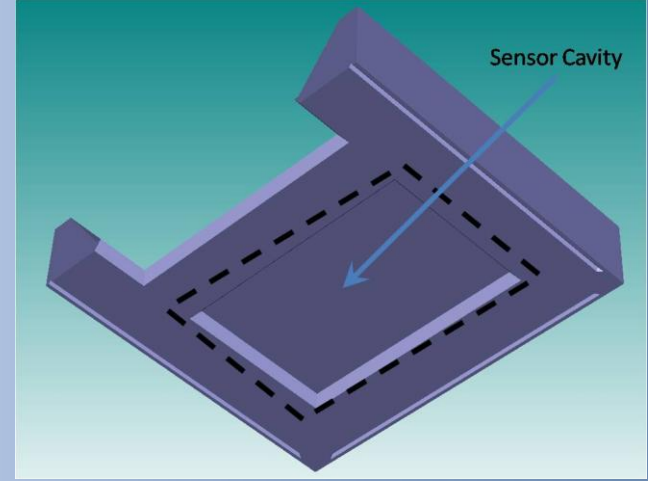
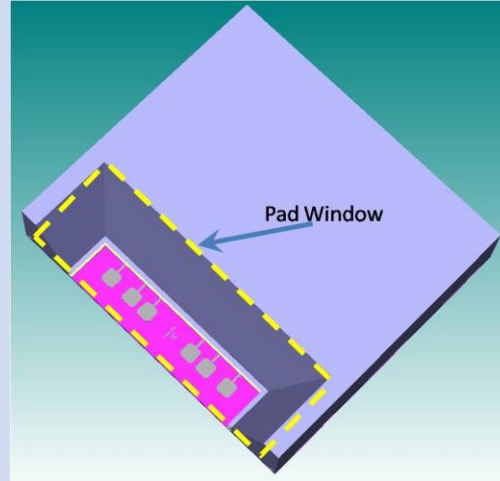
Next Steps

1. Etch isolation trenches
2. Fill isolation trenches, planarize
3. Deposit and pattern intermediate layer, passivate sidewalls, then open bottom of trenches
4. Deposit and pattern metal
5. DRIE mechanical structure
6. Passivate sidewalls, then open bottom of trenches
7. Isotropic release etch
8. Strip etch mask, sidewall passivation + buried oxide
9. Cap wafer by glass frit wafer bonding



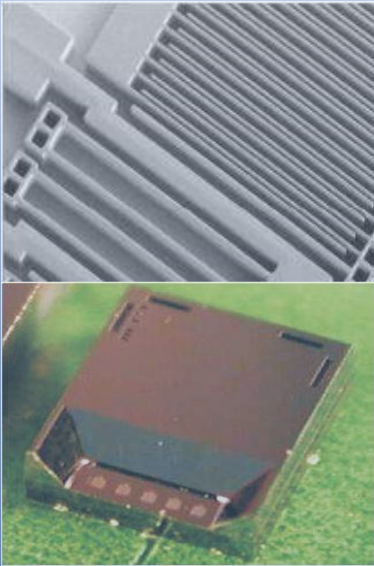
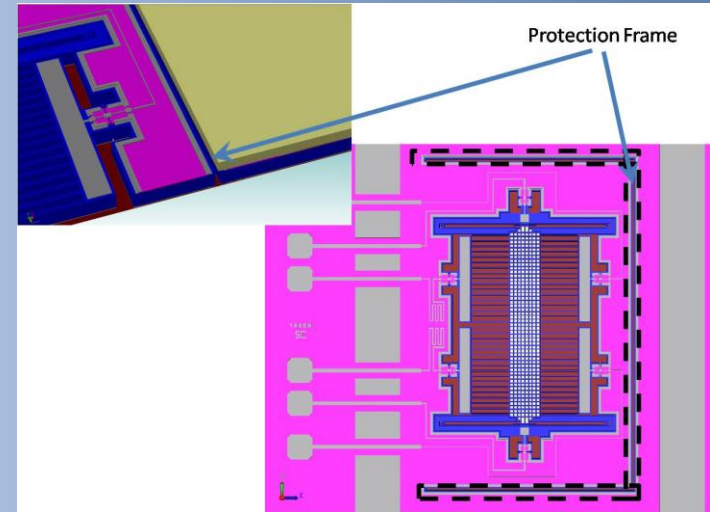
SEM picture of gyroscope chip with open cap wafer

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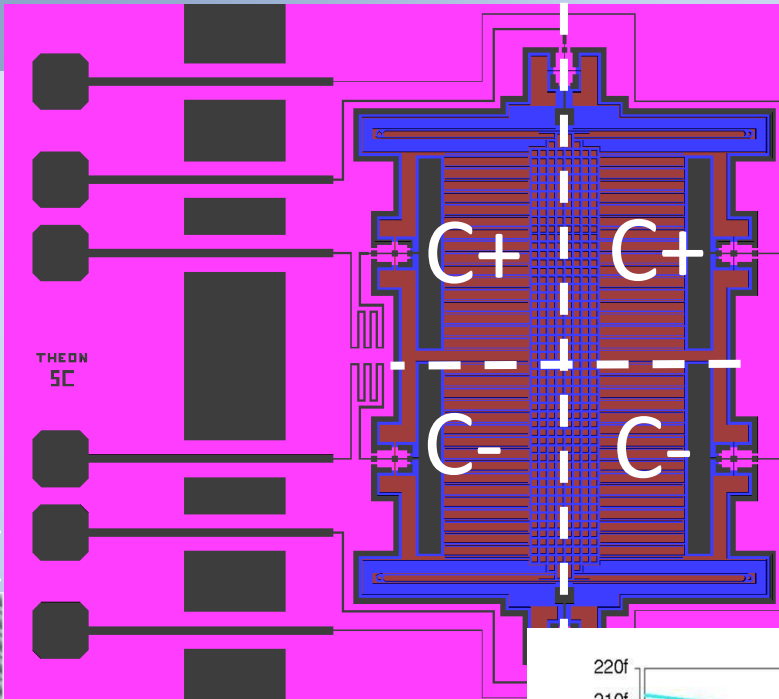
Wafer level encapsulation major steps

- The wafer cap preparation
- The frit glass bonding
 1. Deposition of the glass paste by screen printing
 2. Conditioning of the paste
 3. Actual bonding



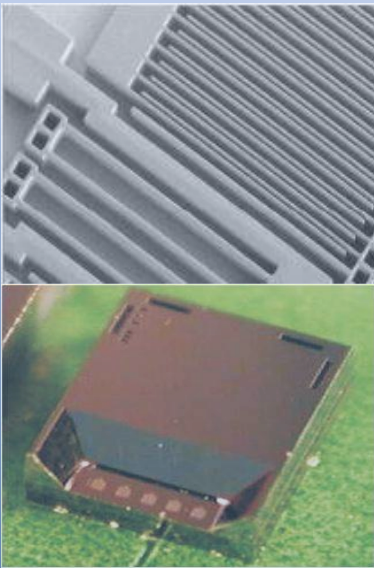
Capacitive MEMS Accelerometer

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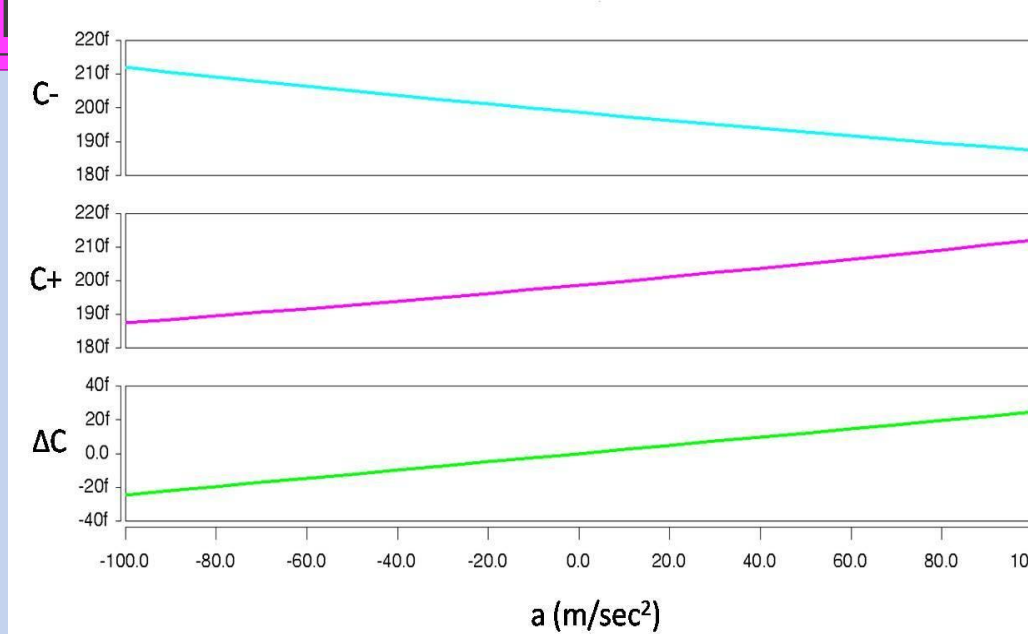


The structure has both a horizontal and vertical symmetry

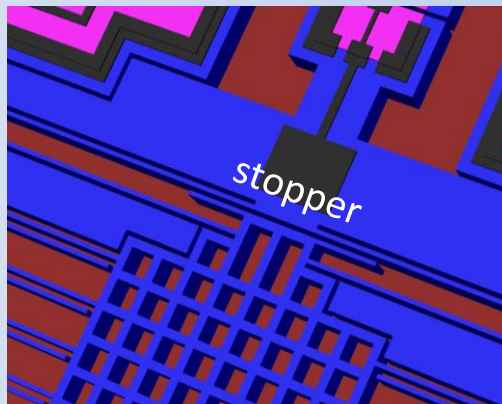
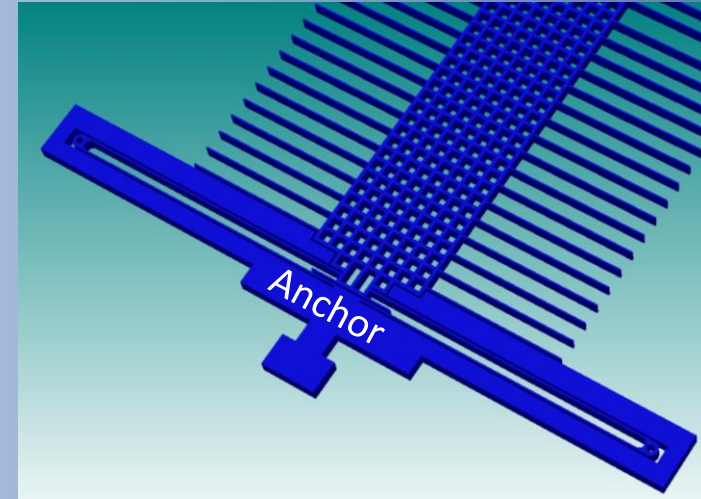
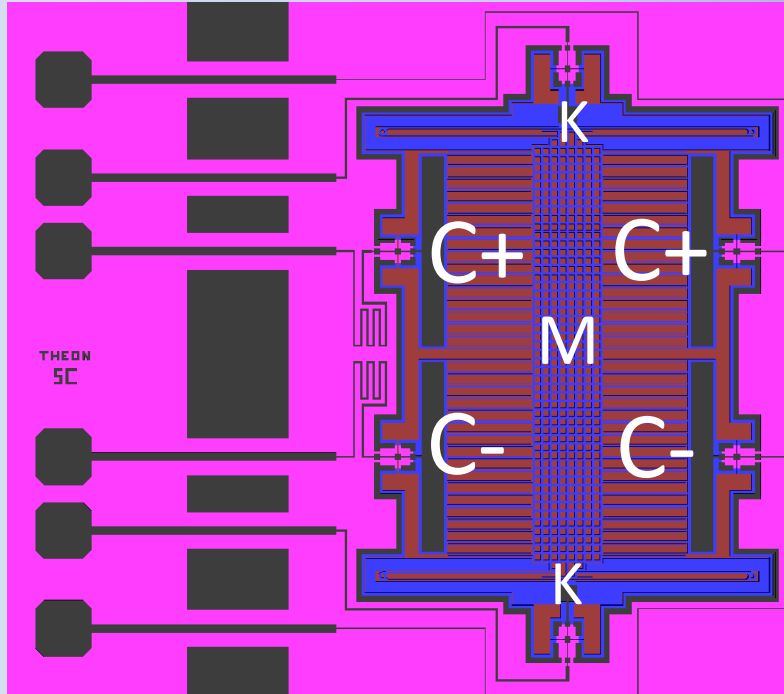
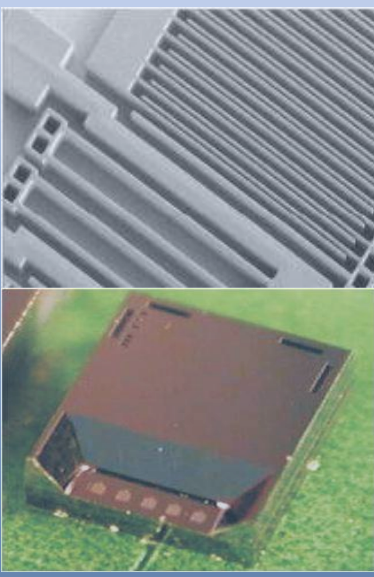
- The symmetry guarantees that:
- The four sensing capacitors are perfectly matched and therefore the common mode effects will be suppressed
 - The 2nd order non-linearity of the parallel plate capacitors will be eliminated
 - The requirement for a fully differential design for the readout electronics is met



Four parallel plate capacitors are used for the electromechanical energy transduction. The input acceleration displaces the proof mass and the acquired kinetic energy is converted to electrical energy which is stored in the four sensing capacitors



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The motion of the proof mass is limited by mechanical stoppers

The stoppers guarantee that:

- The maximum in-plane displacement of the proof mass is always lower than the gap between the fingers of the parallel plate capacitors so as to avoid a catastrophic short-circuit (i.e in case pull-in instability occurs)
- The maximum in-plane motion is kept within specific limits so as to avoid fracture of the mechanical springs



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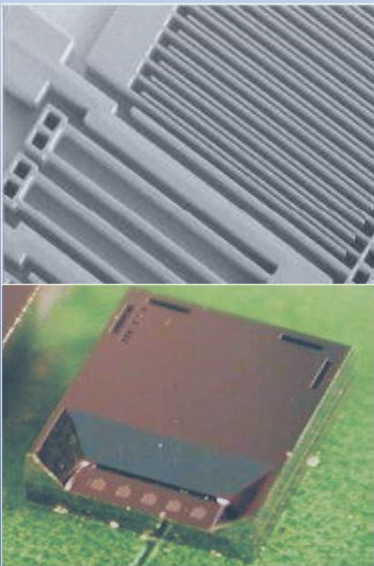
Next Steps

Technology (X-FAB 0.18 um mixed signal CMOS, XC018)

- Low cost CMOS technology
- Standard European CMOS technology (No subject to ITAR)
- Excelent “Analog” performance
- Compatibility/Future co-integration with MEMS

Analog Design Blocks

- Discrete-time (Switched Capacitor) configuration are used appropriate for applying noise reduction techniques
- Capacitance to Voltage Converter (sense transducer capacitance)
- $\Sigma\Delta$ modulator (Quantization of analog response)
- Compensator (to ensure stability in close loop designs)



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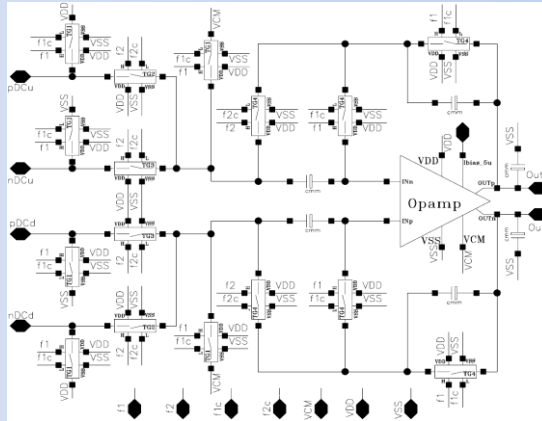
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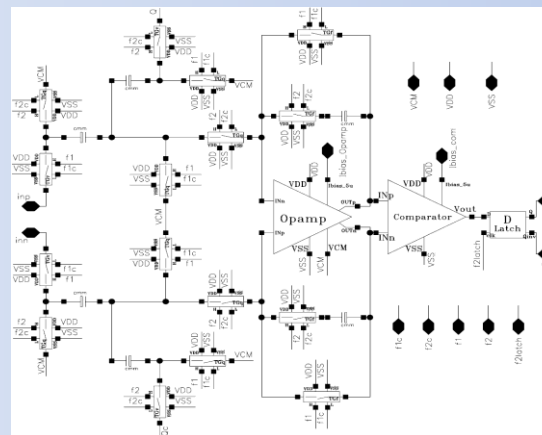
Next Steps

Capacitance to Voltage converter

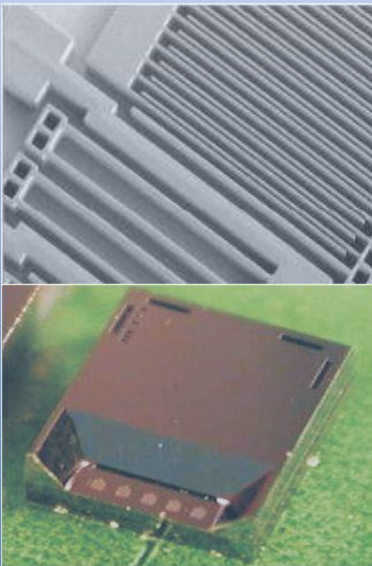


- The Capacitance of the Transducer is sensed
- Input Capacitance is translated into differential voltage
- Noise reduction techniques are applied

Sigma Delta ($\Sigma\Delta$) modulator



- Input Voltage is quantized (1Bit $\Sigma\Delta$ are appropriate for low bandwidth application)
- 1st of 2nd order $\Sigma\Delta$ modulators could be used regarding the desired quantization noise level
- Noise reduction techniques are applied



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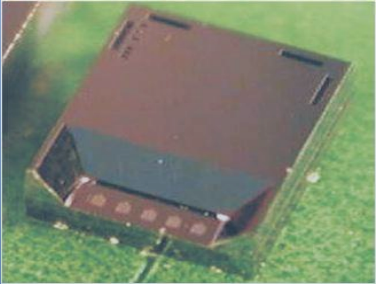
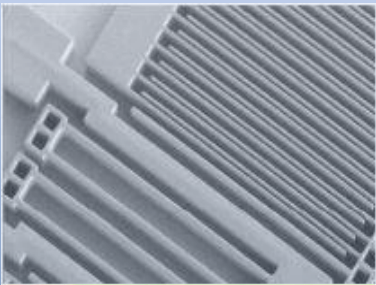
Next Steps

Short term

- Fabrication and functional verification of MEMS (Q1 2011)
- Fabrication and functional verification of ASIC (Q1 2011)

Mid term

- Fabrication and functional verification of the component (Q3 2011)



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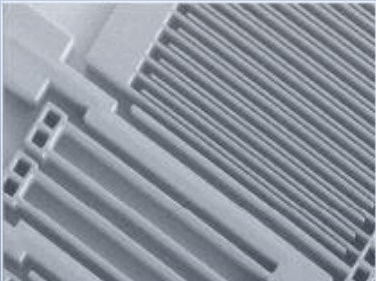
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Thank you for your attention

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