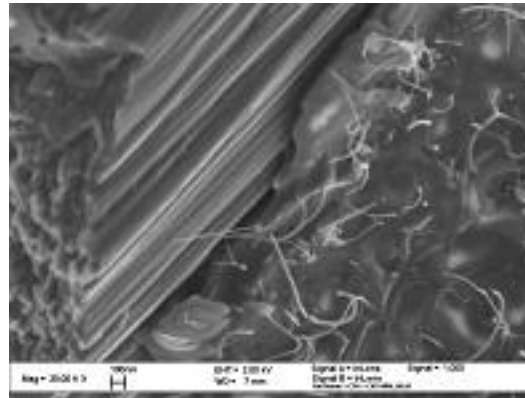


Carbon Nanotube Doped CFRP's - Challenge and Improvement for Space Structures?

7th ESA Round-Table on MNT for Space Applications



Maximilian Klebor
Felicitas Hepp, Ernst K. Pfeiffer
Pierre Lodereau, Laurent Pambaguian

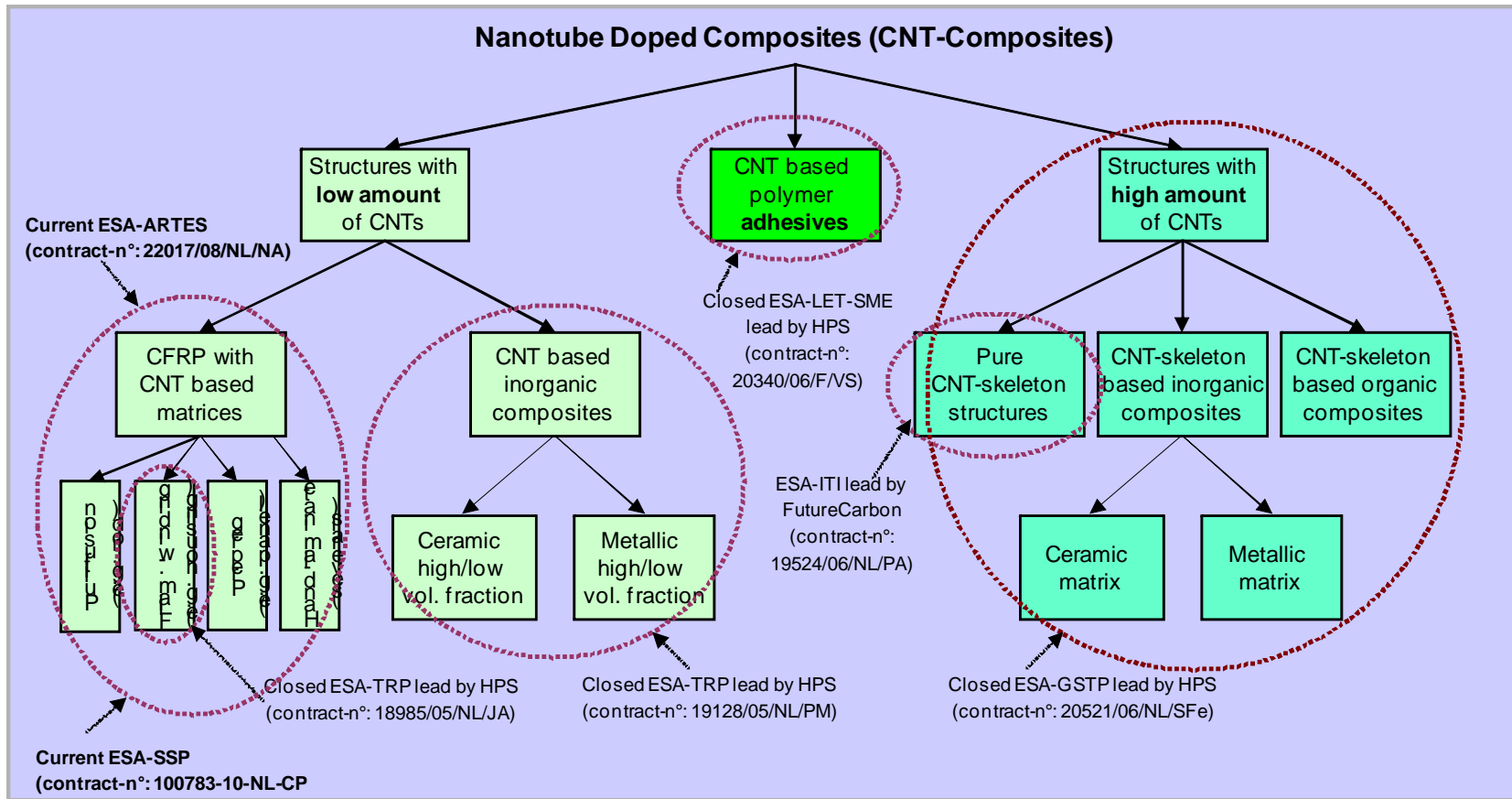
Noordwijk, September 15th, 2010

Motivation of CNT Doping

Why doping CFRP's with CNT's although cost increase is involved?

- Improvements in material behaviour can:
 - directly or indirectly reduce mass of current CFRP parts on spacecrafts
 - allow for the substitution of parts that are currently made from metals and other materials
- Reduction of mass allows e.g. for more payload (mass = money)

CNT Heritage Overview at HPS

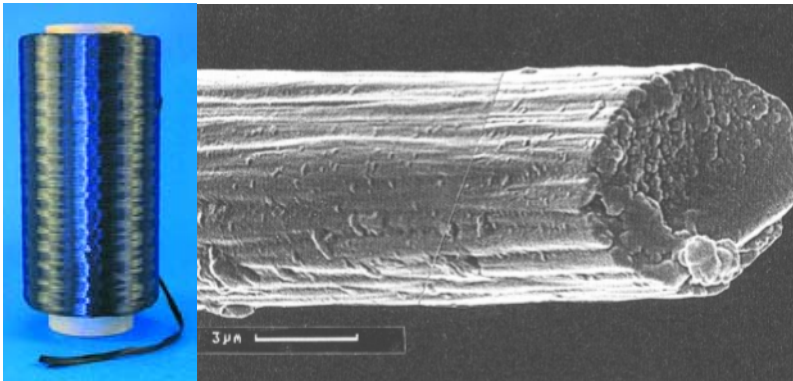


In this presentation only CNT-CFRP's are addressed.

CNT-CFRP

We call a CNT-CFRP when it consists of the following ingredients:

Carbon (reinforcing) fibres



Left: carbon fibre roving bobbin

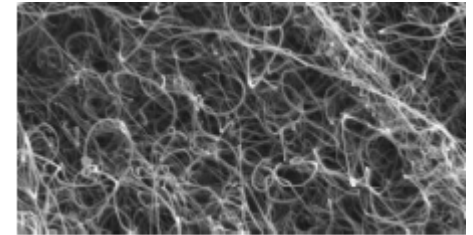
Right: single carbon fibre

Polymeric resin (system)



Epoxy resin

Nano dopant: CNT

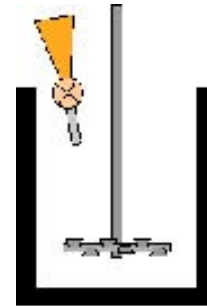


CNT network

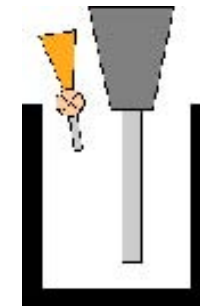
CNT-Resin Production

Depending on the ingredients and on the targeted CNT-resin viscosity different processes are suitable or a combination of them:

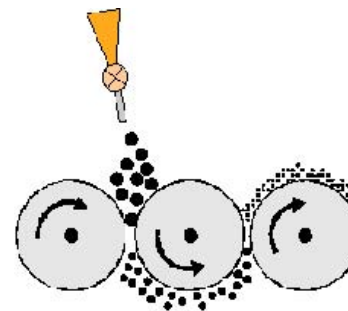
- Dissolver
- Ultra sonic
- Roll (single, multiple)
- Centrifuge (single, multiple)
- Etc.



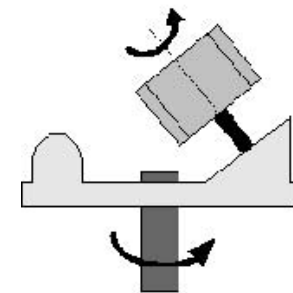
Dissolver



Ultra sonic



(Multiple) roll



(multiple) Centrifuge

Images by FutureCarbon

Specimen Production

Different CNT-CFRP manufacturing techniques were explored:

- Hand laminate, vacuum assisted
- RTM
- “Small scale” pre-pregging
- “Industrial scale” pre-pregging
- Winding
- Pultrusion

*CNT-CFRP winding
at Invent*

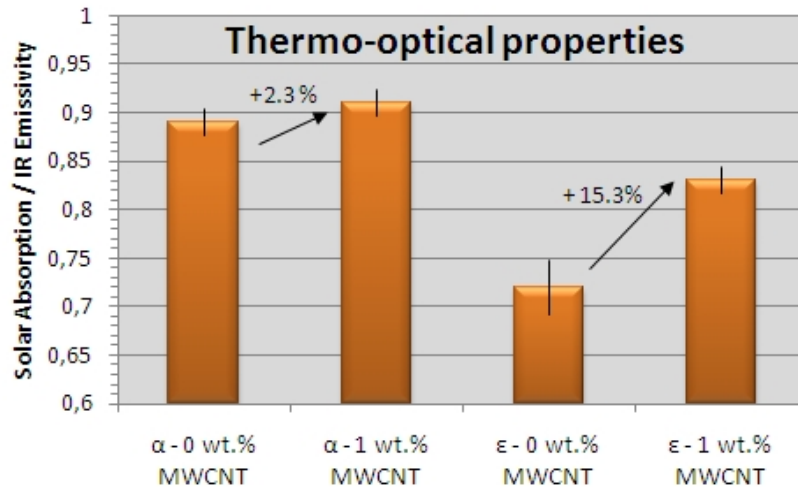


*CNT-CFRP pultrusion at Inegi
(lab scale)*

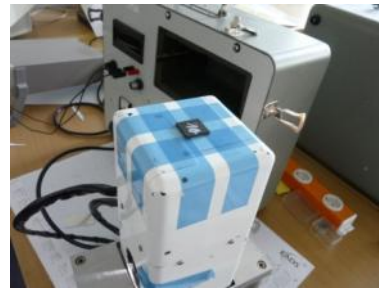


Prepreg production

Results –Thermo-optical Properties



Solar absorption and infrared emissivity measured acc. to ECSS-Q-ST-70-09C

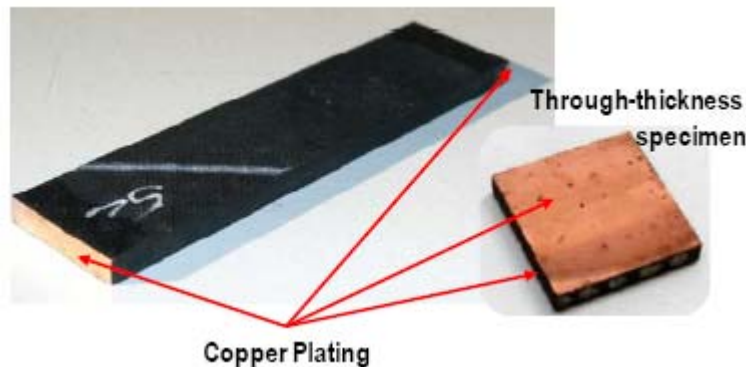


Test apparatus Gier Dunkle DB100 / Gier Dunkle MS251 of Astrium Ottobrunn

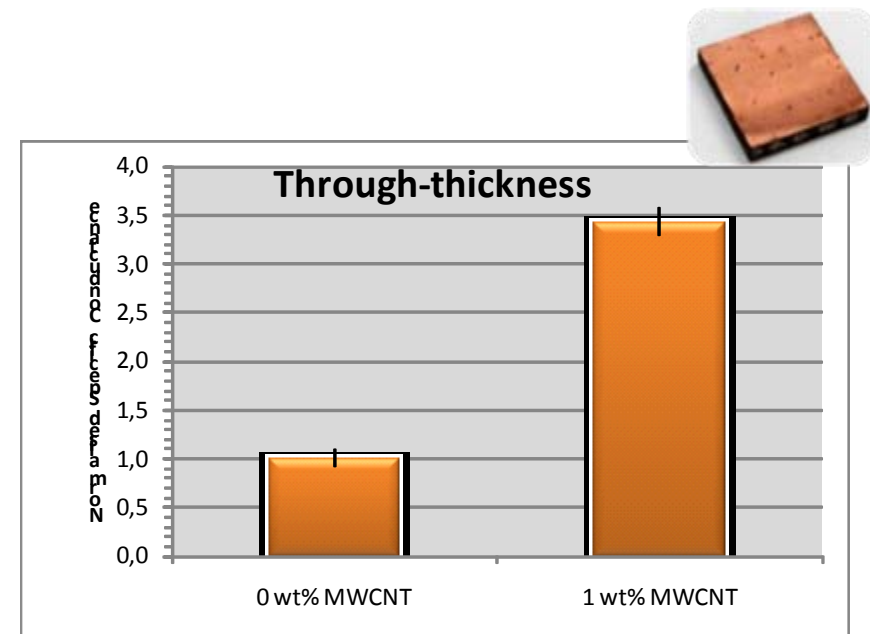
- All present tests with different ingredients show a high increase of infrared emissivity (ϵ), i.e. approx. 10% - 19%
- Only minor increase in solar absorption
→ Radiator application?
- 1wt.% as-produced MW-CNT but also other filler fractions were examined

Results – Electrical Conductivity

- Fibre in-plane electrical conductivity could not be improved significantly
- Transverse electrical conductivity could be improved by a factor of approx. 2 - 4



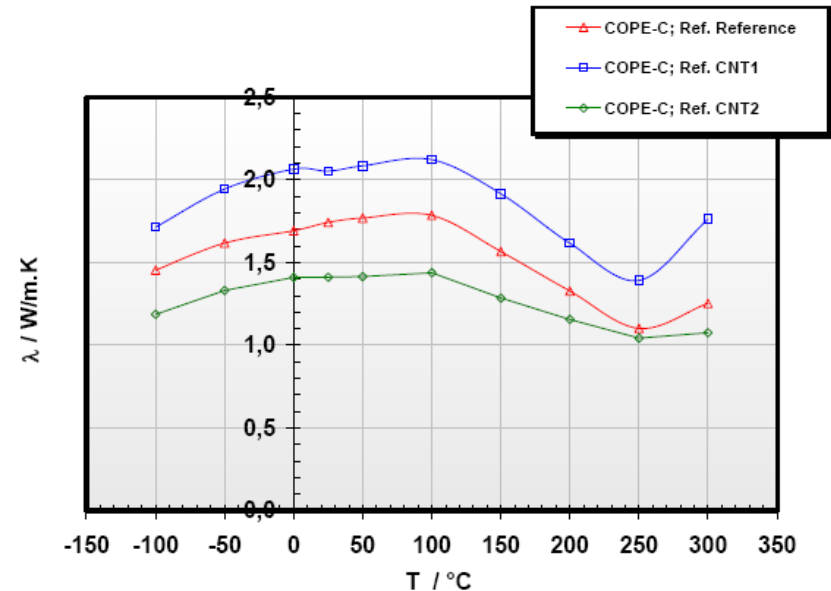
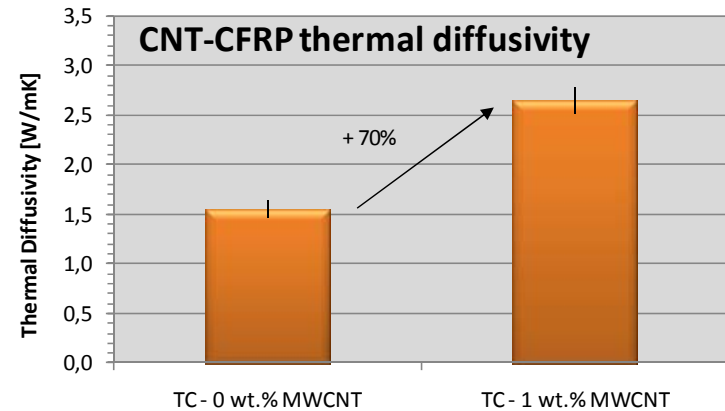
Specimens for resistance measurements



Normalised specific conductance of 1 exemplary CNT-CFRP material combination (UHM & epoxy) and its reference

Results – Thermal Conductivity

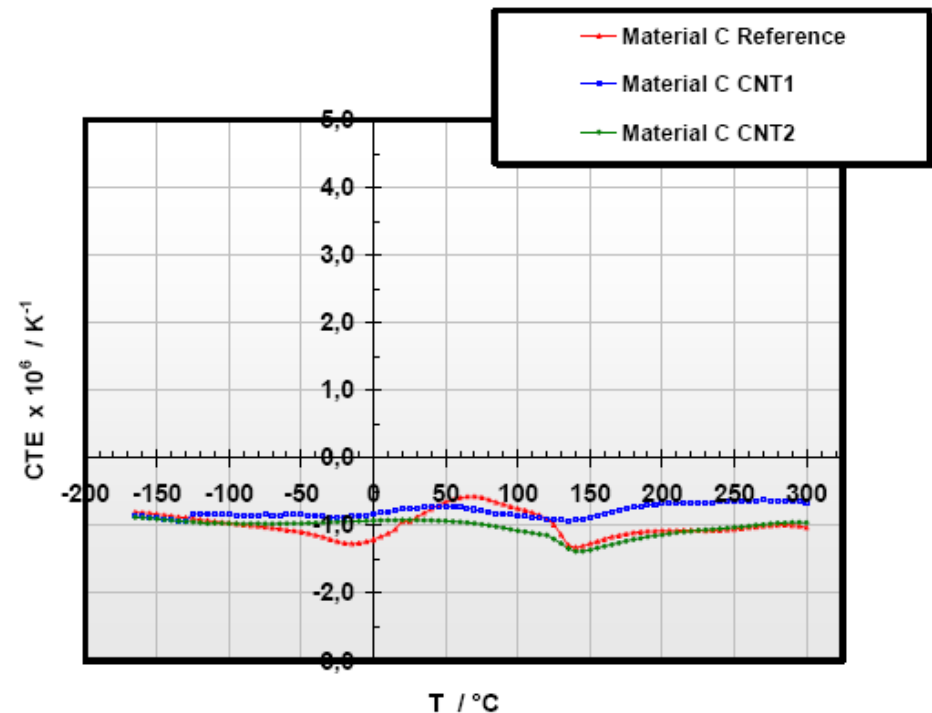
- CNT-CFRP fibre in-plane thermal conductivity not determined due to low importance (C-fibre dominated property).
- Transverse thermal conductivity could be improved by up to approx. 100%
- Also degradation in thermal conductivity observed, potential reason: decreased FVF.



Thermal conductivity of different CNT-CFRP materials

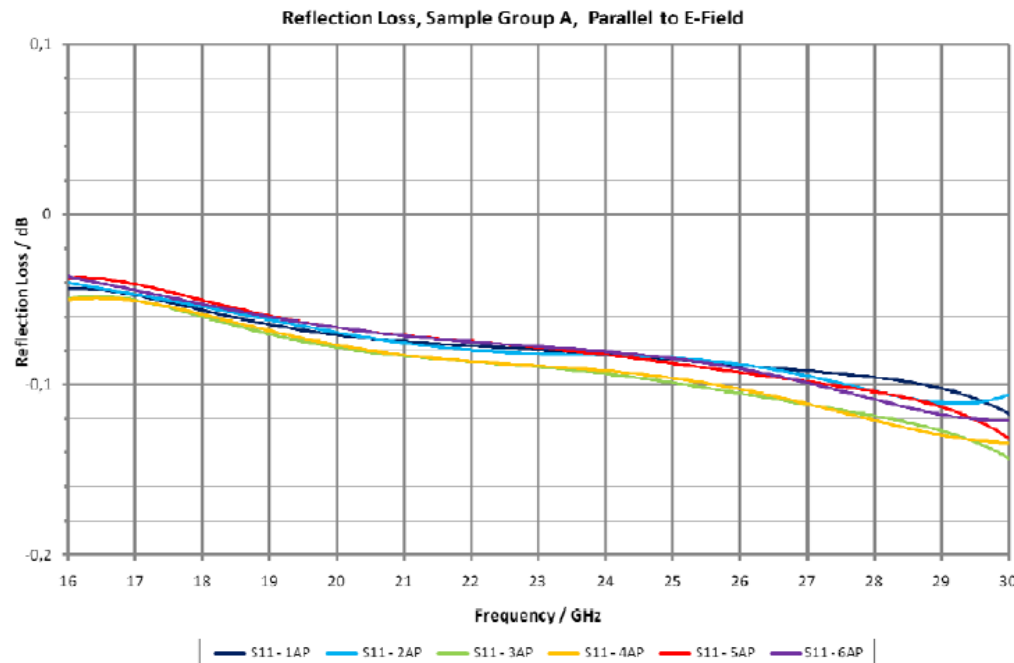
Results – Coefficient of Thermal Expansion (CTE)

- CTE of tested materials yield the following results:
 - slight increase
 - slight decrease
 - no significant change (see graph, material CNT2 and its reference)
- CTE of tested CNT materials show no apparent trend



CTE measurements at AIT for one specific material and its CNT variations

Results – Radio Frequency Testing



Representative for RF properties, i.e. reflection loss

scope of testing:

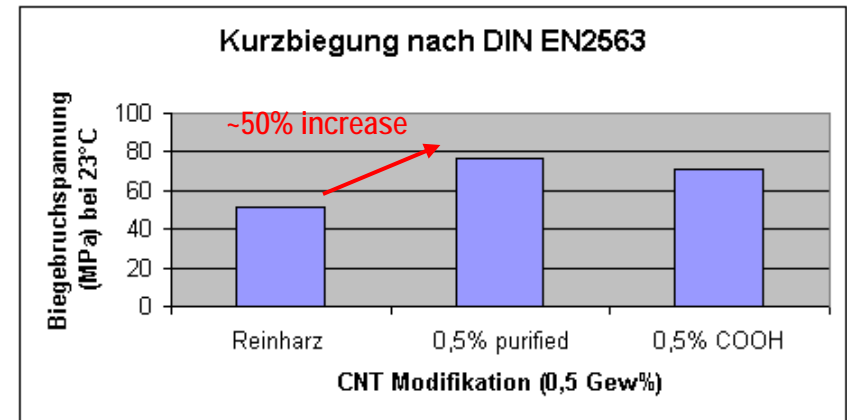
- Ku-band up to Ka-band (16-30GHz)
- Reflection Loss (S11)
- Amplitude Variation ΔA for orthogonal polarizations
- Phase Variation ΔP for orthogonal polarizations
- Depolarization
- Transmission Loss (S21)

No major impact on RF properties due to CNT doping observed.

Results – Mechanical Strength

Reference test for mechanical strength and laminate quality:

- ILSS acc. to EN 2563
- High scattering of results, i.e. some materials show improvement (see right side) some show decline in ILSS
- Results appear to correspond to manufacturing process
- Good results for pultrusion and winding (see figure on right side)



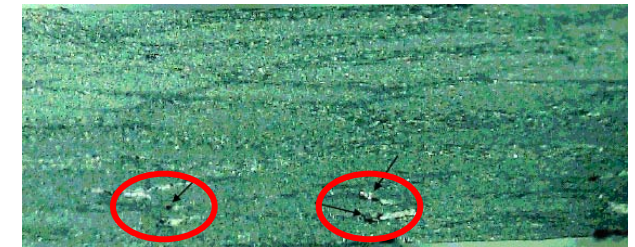
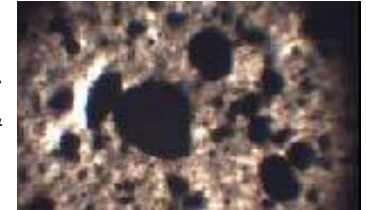
*Representative for mechanical properties, i.e. ILSS
- Results for Invent made "fibre wound pre-prepregs"*



Encountered Problems

- High variety in ingredients
 - CNT's (e.g. as-produced, purified, processing, number of walls, aspect ratio, etc.)
 - Fibres (HT, IM, HM, UHM etc.)
 - Resins (epoxies, cyanate esters etc.)
- Variation CNT dispersion quality in resin systems (agglomerations etc.)
- Increase of viscosity prohibits some manufacturing techniques
- Filter effects prohibits the use of RTM manufacturing at reasonable fibre volume fractions, i.e. approx. 50% and higher
- For the moment unexplainable variations in test results → reproducibility problem
- Health aspects require special handling

CNT-resin microscopic image of Invent



CNT-CFRP cross-section polish with distributed pores



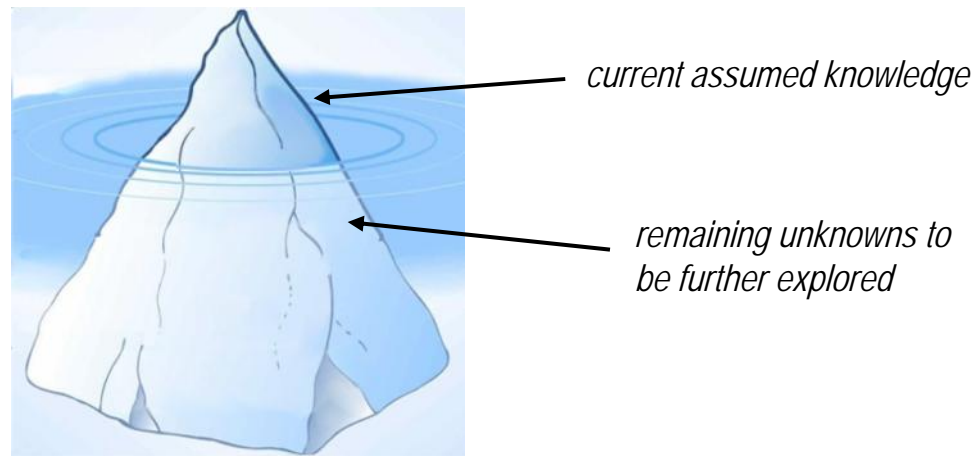
Sample plate produced at TU Hamburg-Harburg

Conclusions

- On first sight some results are somewhat “disappointing” since either only slight improvements were observed or sometimes even a degradation was observed
→ Too high expectations?
- However also good and pleasant improvement results are found – future activities should aim for their further enhancement
- Manufacturing techniques are of high importance and determine the final CNT-CFRP characteristics
- There are favourable combinations of the ingredients (CNT/resin/fibre) and favourable manufacturing processes: pultrusion & winding
- Reproducibility trouble

Conclusions – cont'd

- First steps of upscaling of CNT-CFRP manufacturing accomplished (prepreg)
- Mechanisms of CNT's inside the CFRP have to be investigated fundamentally



- Currently, we can only see the top of the iceberg → Target: further research!

Future Prospects

- Fundamental research of mechanisms required
- Only “primary” characteristics were examined. However, “secondary” characteristics are also of high importance such as:
 - Machinability
 - Adhesion of coatings and adhesives
 - Ageing resistance
 - Dynamical performance
 - Micro fracture resistance, already observed: less micro crack propagation & brittleness
 - Etc.
- Determination of appropriate material combinations (e.g. 5 fibre types, 5 matrices, 5 CNT's, 3 different filler volume fractions already yield a tremendous amount of testing)
- Establishment of qualified manufacturing processes and their enhancement

Acknowledgments

Advanced Composite Group (UK)

Astrium (D)

Austrian Institute of Technology (A)

DLR - *German Aerospace Center* (D)

European Space Agency

FutureCarbon (D)

German Ministry of Economics & Technology

Invent (D)

Inegi (P)

PTJ (D)

University of Patras (GR)

TU Hamburg-Harburg (D)

Main Office:

HPS GmbH
Christian-Pommer-Str. 34
38112 Braunschweig
Germany

Munich Office:

HPS GmbH
Perchtinger Straße 5
81379 München
Germany

HPS Portugal:

HPS Lda.
Rua Roberto Frias 378
4200-465 Porto
Portugal

www.hps-gmbh.com



High Performance Space
Structure Systems GmbH