



SRPROTEC - SPACE RADIATION SHIELDING IN MICROELECTRONICS THROUGH THE DEVELOPMENT OF ADVANCED COMPOSITE MATERIALS



ALTER | **ACCEDE | ESCCON**
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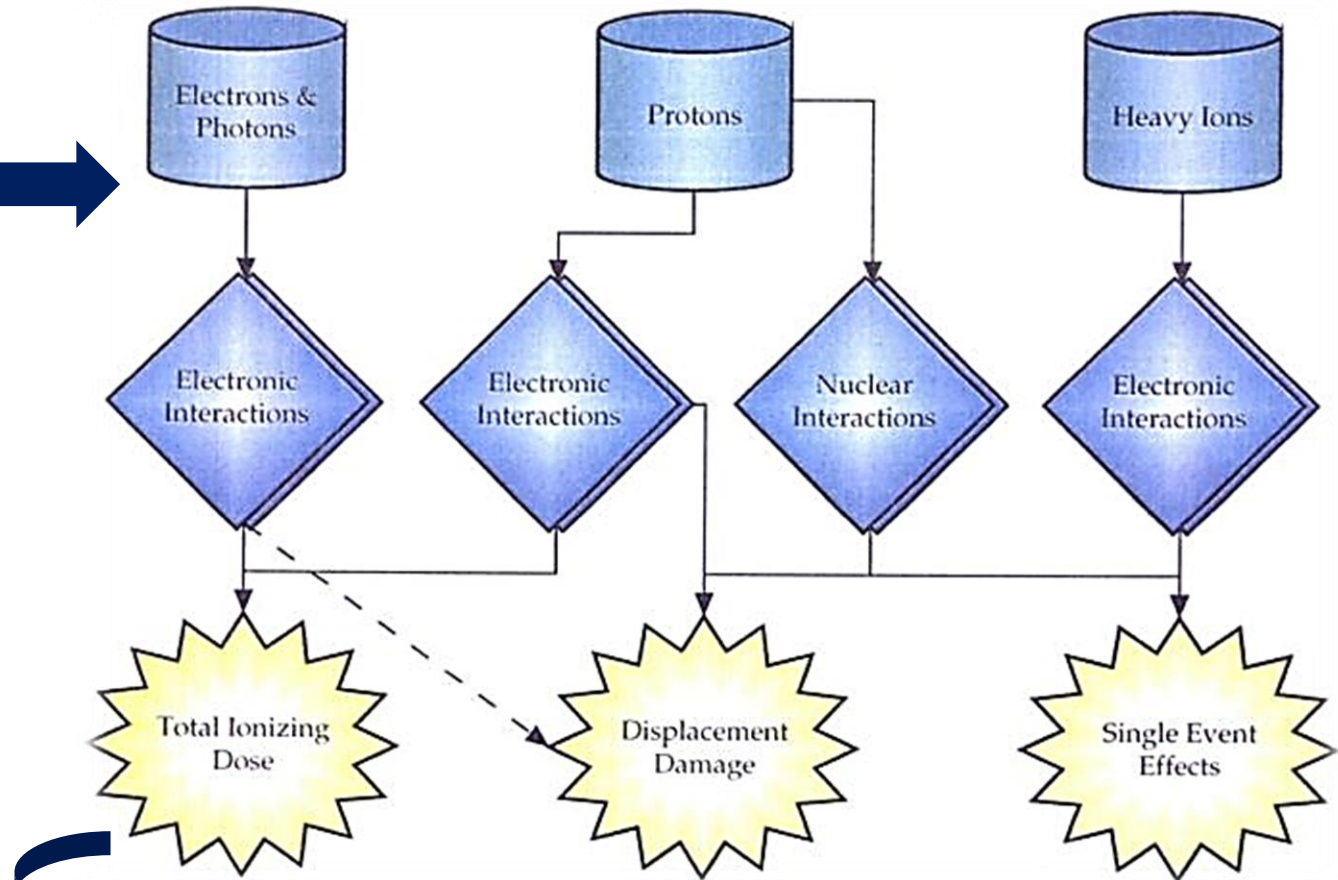
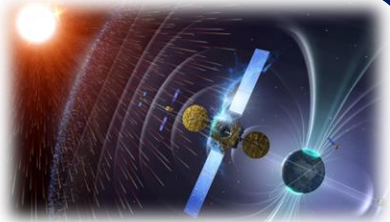
Proyectos en Colaboración Público-Privada
CPP2021-008565

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1. Introduction and objective

- ✓ GCRs
- ✓ SEPs
- ✓ Trapped particles in Van Allen belts



- Failures: Parametric and functional
- ✓ Leakage current
 - ✓ Threshold voltage
 - ✓ ...

J.W. Howard, Jr. And D.M. Hardage.
Spacecraft Environments Interactions:
Space Radiation and Its Effects on
Electronic Systems

1. Introduction and objective

Protection of microelectronics



Passive shielding



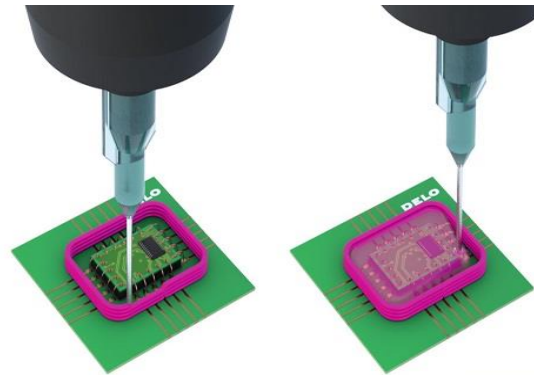
✓ Development
✓ Validation



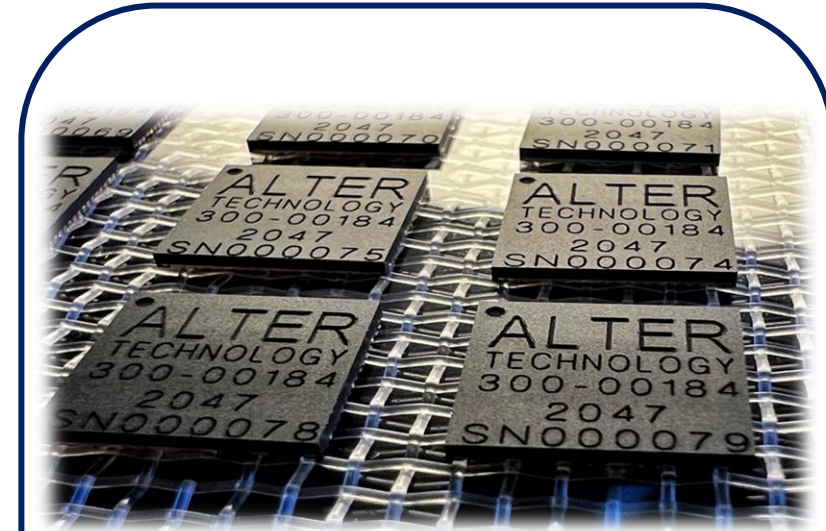
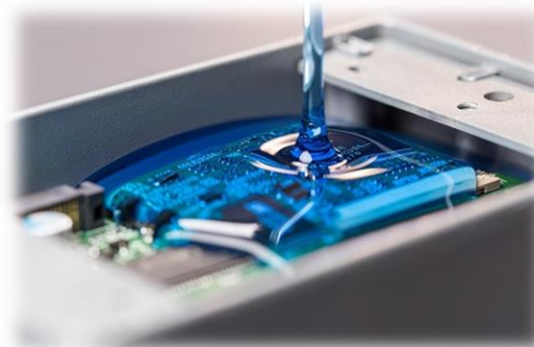
- New composite materials with space radiation shielding performance



Dam & Fill



Potting process



Process integrated light weight solution for space radiation

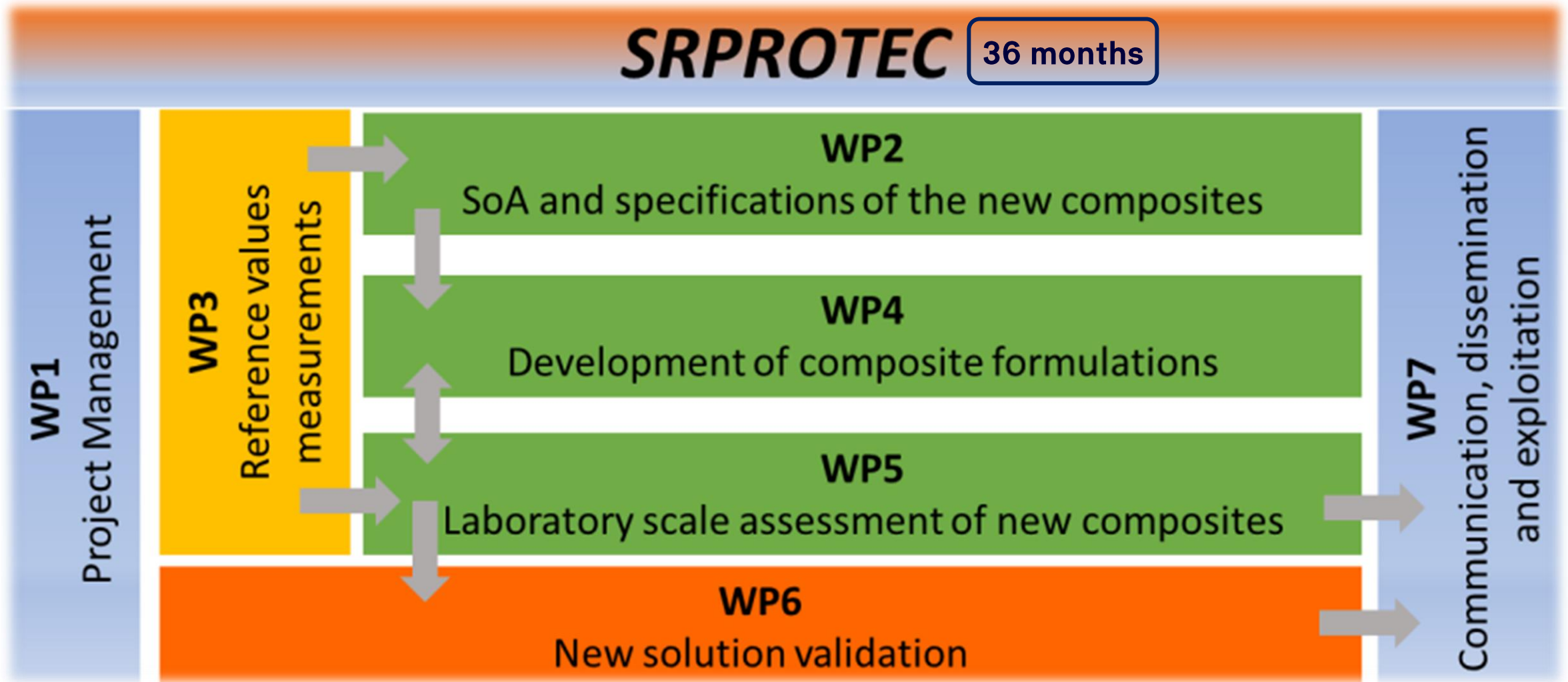
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Keeping desired MC properties

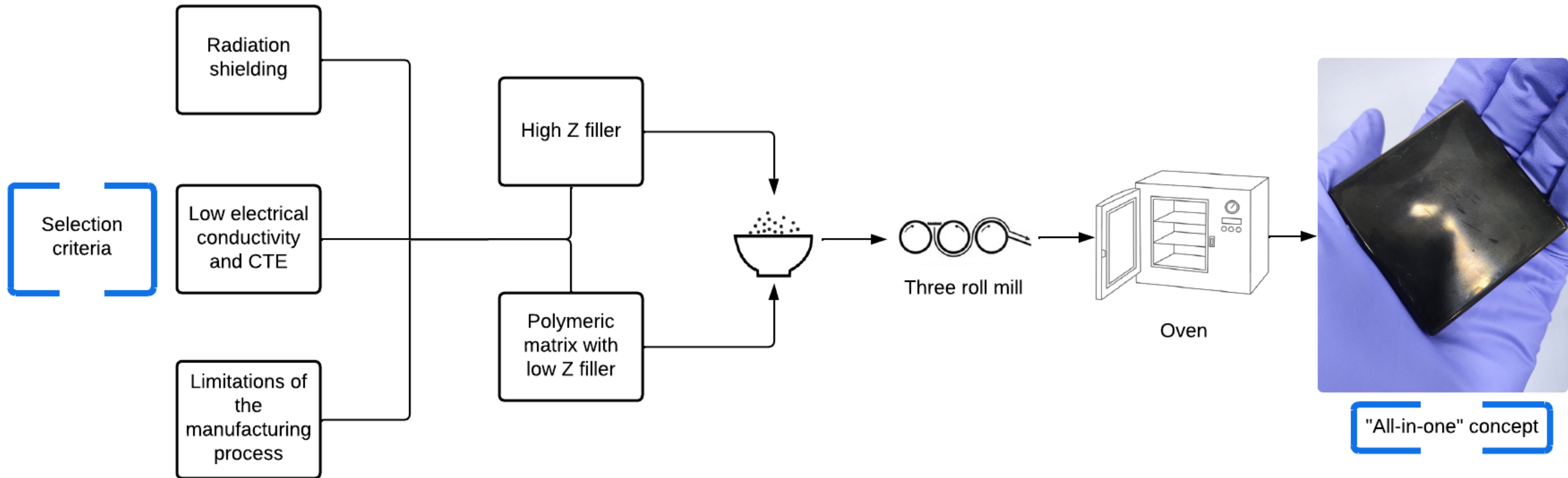
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Applicable as potting compound

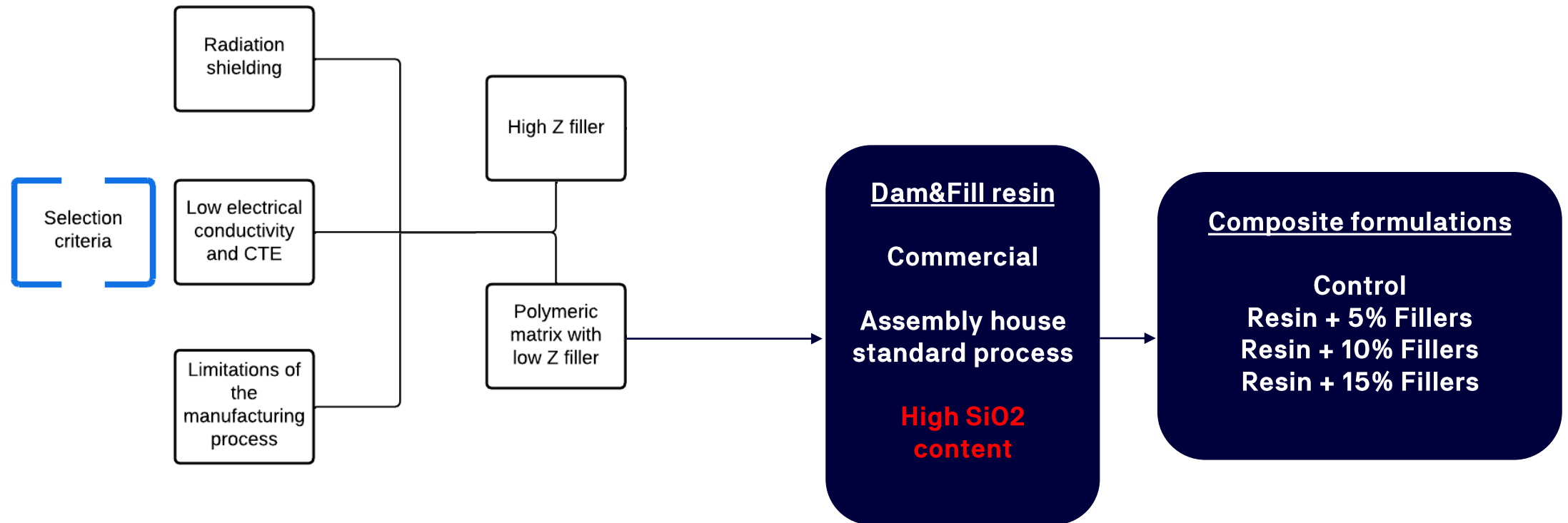
2. Project structure



3. New composite materials development (WP2-WP4)



3. New composite materials development (WP2-WP4)

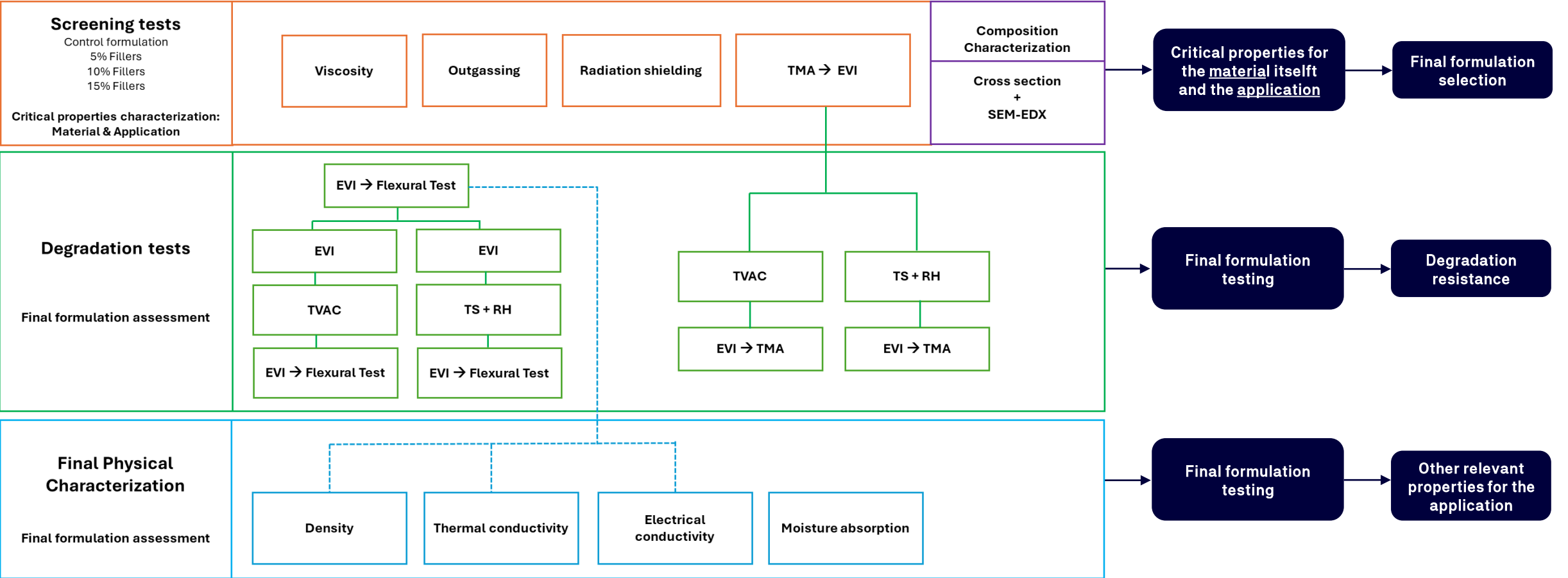


4. Materials characterization (WP5)

WP5
Laboratory scale assessment of new composites

*TMA: Thermomechanical Analysis
*EVI: External visual inspection
*TVAC: Thermal Vacuum Cycling
*TS: Thermal Shock
*RH: Relative humidity (humidity exposure)

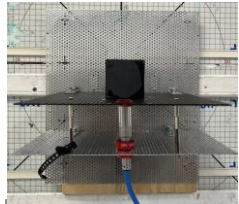
Material test plan elaborated considering
ECSS-Q-ST-70C Rev. 2: Materials, mechanical parts and processes



5. Radiation shielding characterization (WP5)

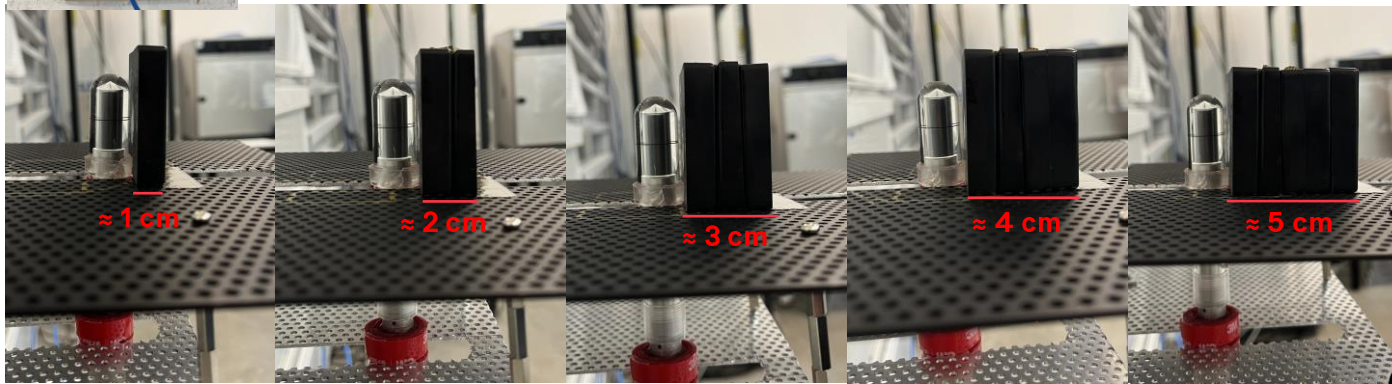
Direct measurements set up

WP5
Laboratory scale assessment of new composites



Gamma radiation shielding test set up development

Goal: To measure a property to quantify radiation shielding for each formulation



Radiation source (Co60)



RadLab
ALTER-CNA
⁶⁰Co radiation source



Electrometer: Live monitoring of charge

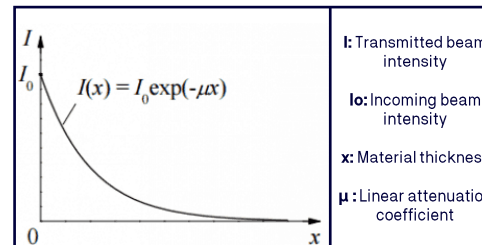
Data treatment

Nk conversion factor
P-T correction factors

Radiation dose obtained for each measurement

Data treatment

Beer-Lambert Law

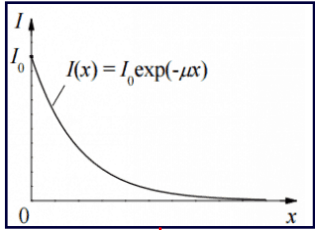


Linear Attenuation coefficient for each material
 μ [cm⁻¹]

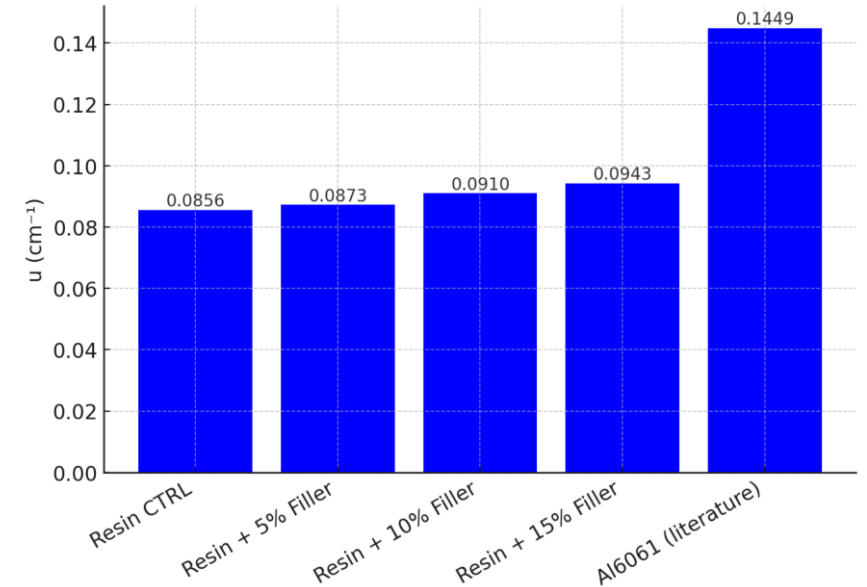
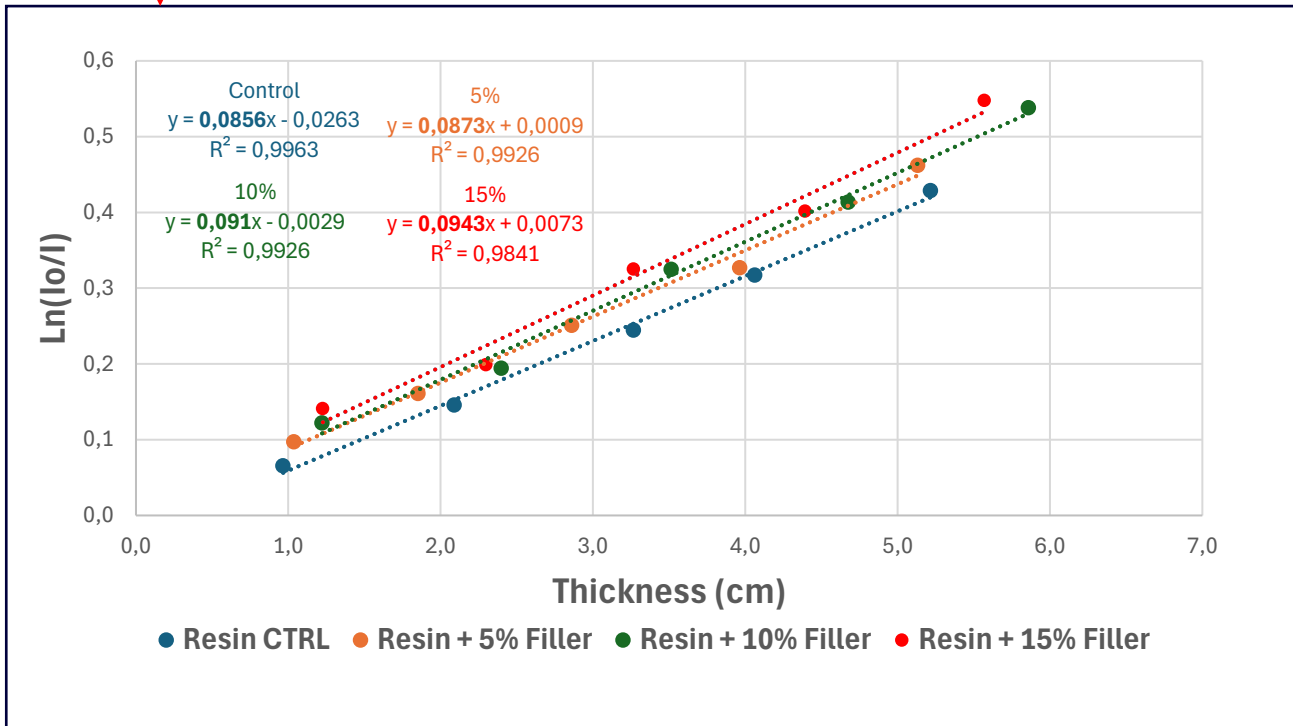
ALTER

6. Materials characterization results (WP5)

Radiation shielding



Ln(I₀/I) vs x (thickness) → μ (slope)



Filler content (%)	μ (cm ⁻¹)	μ Formulations / μ Al6061 (%)
0%	0,0856	59,06
5%	0,0873	60,24
10%	0,0910	62,79
15%	0,0943	65,07

6. Materials characterization results (WP5)

TMA

Average values			
Formulation	CTE 1 (T<T _g) [μm/°C]	T _g (°C)	CTE 1 (T>T _g) [μm/°C]
Resin (control)	19,27	135,46	51,60
Resin + 5% Filler	18,01	135,08	47,82
Resin + 10% Filler	17,84	130,91	44,88
Resin + 15% Filler	17,53	144,67	46,50

CTE decreases for formulations with filler content when compared with control formulation

CTE mismatches decreased

Si ≈ 2,6 μm/°C
Cu ≈ 17 μm/°C

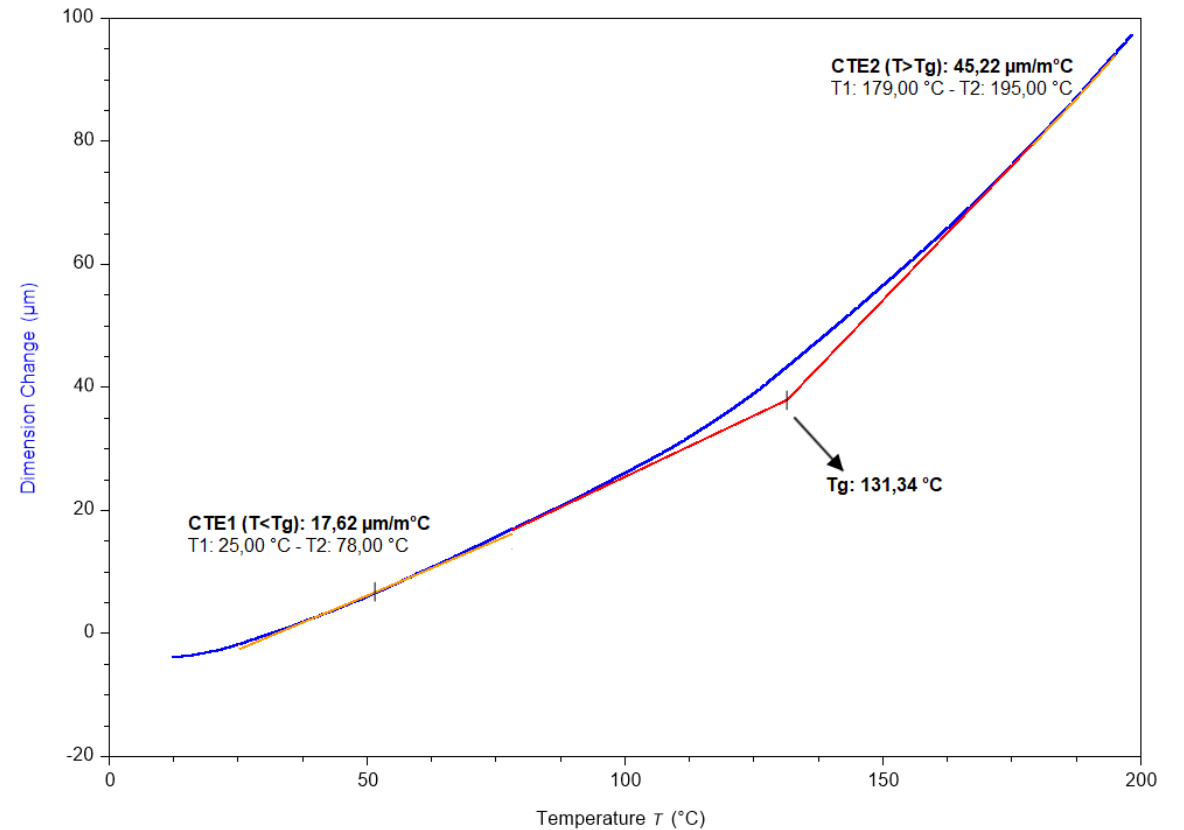
DS T_g (DMTA) = 187 °C

Measured T_g (TMA) < DS T_g (DMTA)

DSC comparison



Resin + 10% (SN 10)

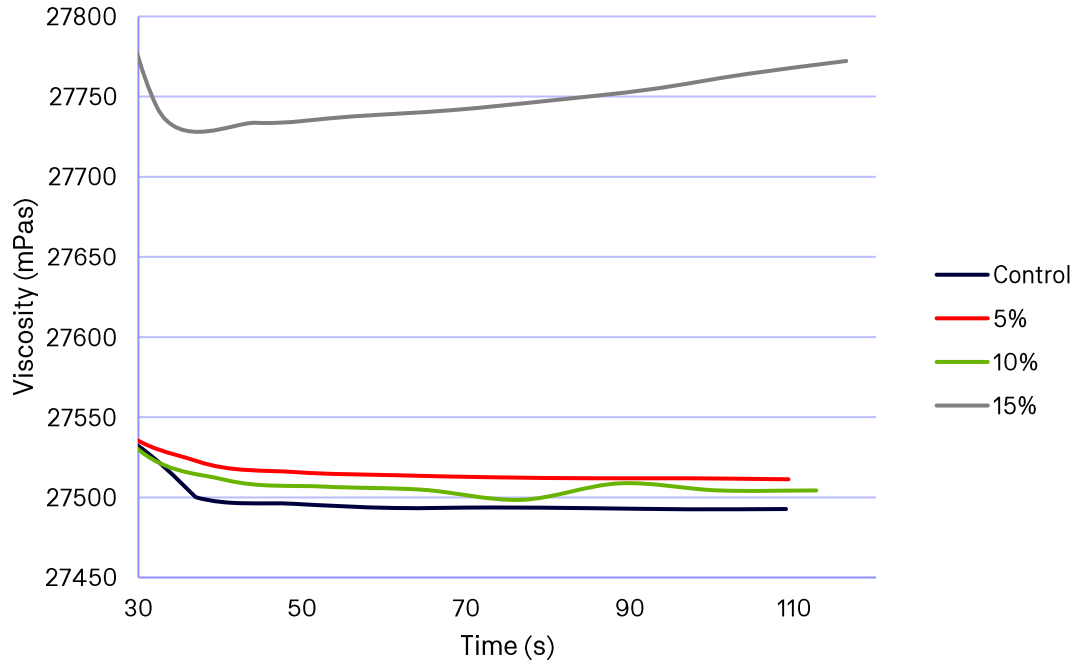


6. Materials characterization results (WP5)



The samples tested are **compliant** with general limits of acceptance for material selection according to **ECSS-Q-ST-70-02C**: RML < 1,00%, CVCM < 0,10%.

Viscosity



Test conditions	
Equipment	Rotational Viscometer Fungilab™
Speed (RPM)	5
Time (s)	120

Outgassing

Sample	TML (%)	SD (%)	RML (%)	SD (%)	CVCM (%)	SD (%)
Control	0,412	±0,004	0,295	±0,003	0,003	±0,002
5%	0,349	±0,021	0,205	±0,019	0,003	±0,003
10%	0,323	±0,010	0,171	±0,025	0,005	±0,001
15%	0,277	±0,013	0,170	±0,008	0,003	±0,001

TEST CONDITIONS-STANDARD	
Material sample temperature:	125 °C
Condensable collectors (CVCM) temperature:	25 °C
Test vacuum:	range 10 ⁻⁶ - 10 ⁻⁷ mbar
RT airconditioning of samples before test:	22 ±3 °C, 55 ±10 %HR, 24 h
RT airconditioning of sample after test:	

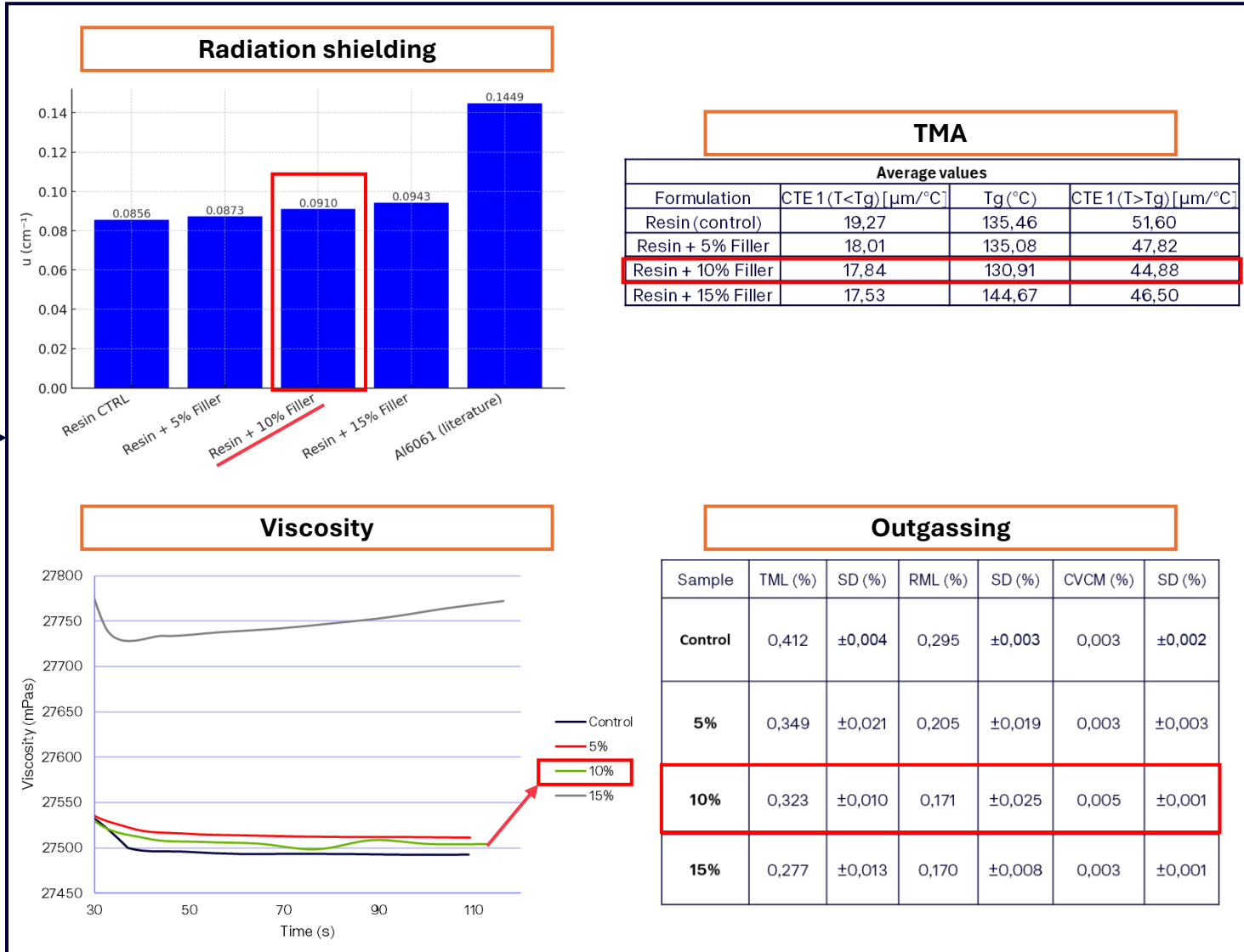


7. Final formulation selection (WP5)

Screening tests approach

Critical properties characterization:

- Material & Application
- Approach valid for other combinations of materials



Resin + 10% Fillers vs Control

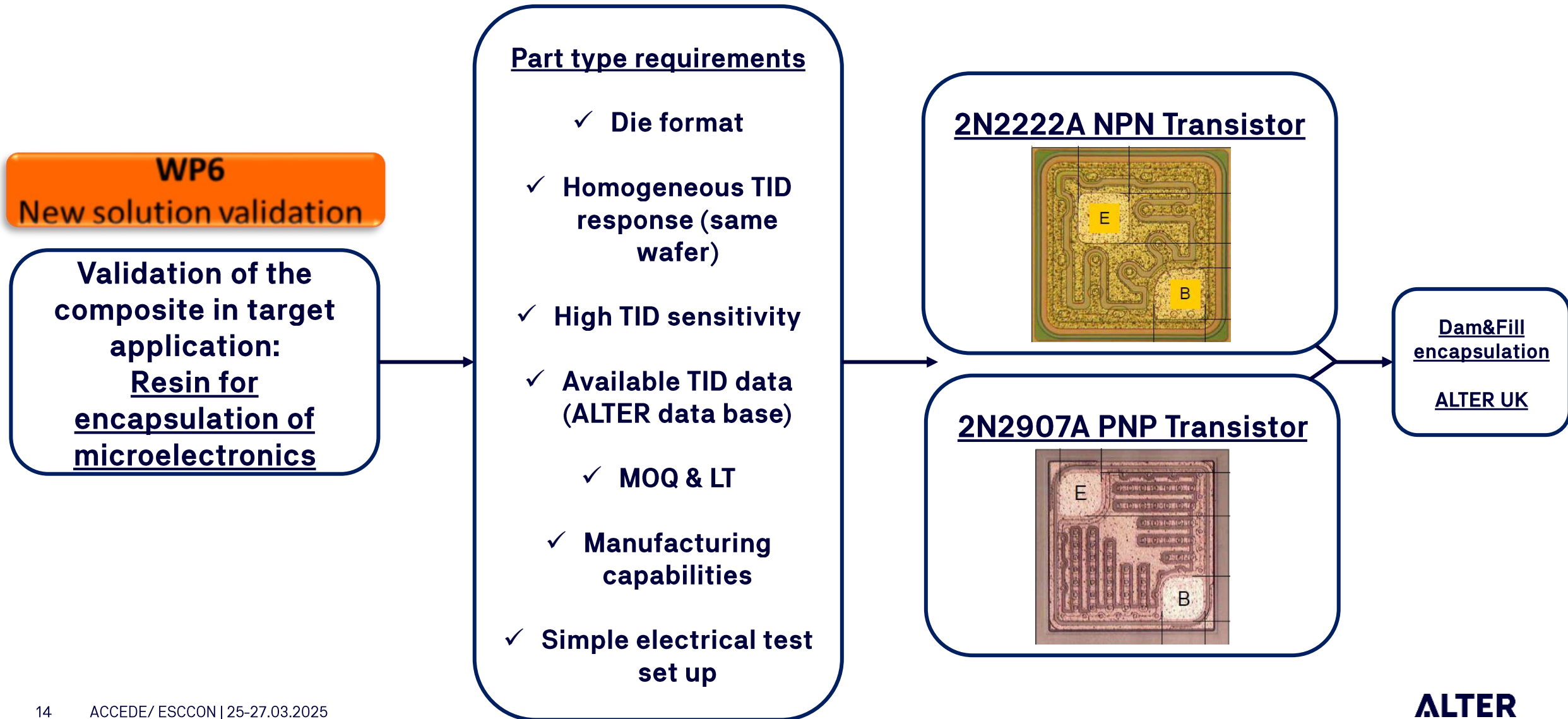
Slight radiation shielding improvement

Thermomechanical properties enhanced (lower CTE)

Compliant with ECSS outgassing limits

Not excessively high viscosity

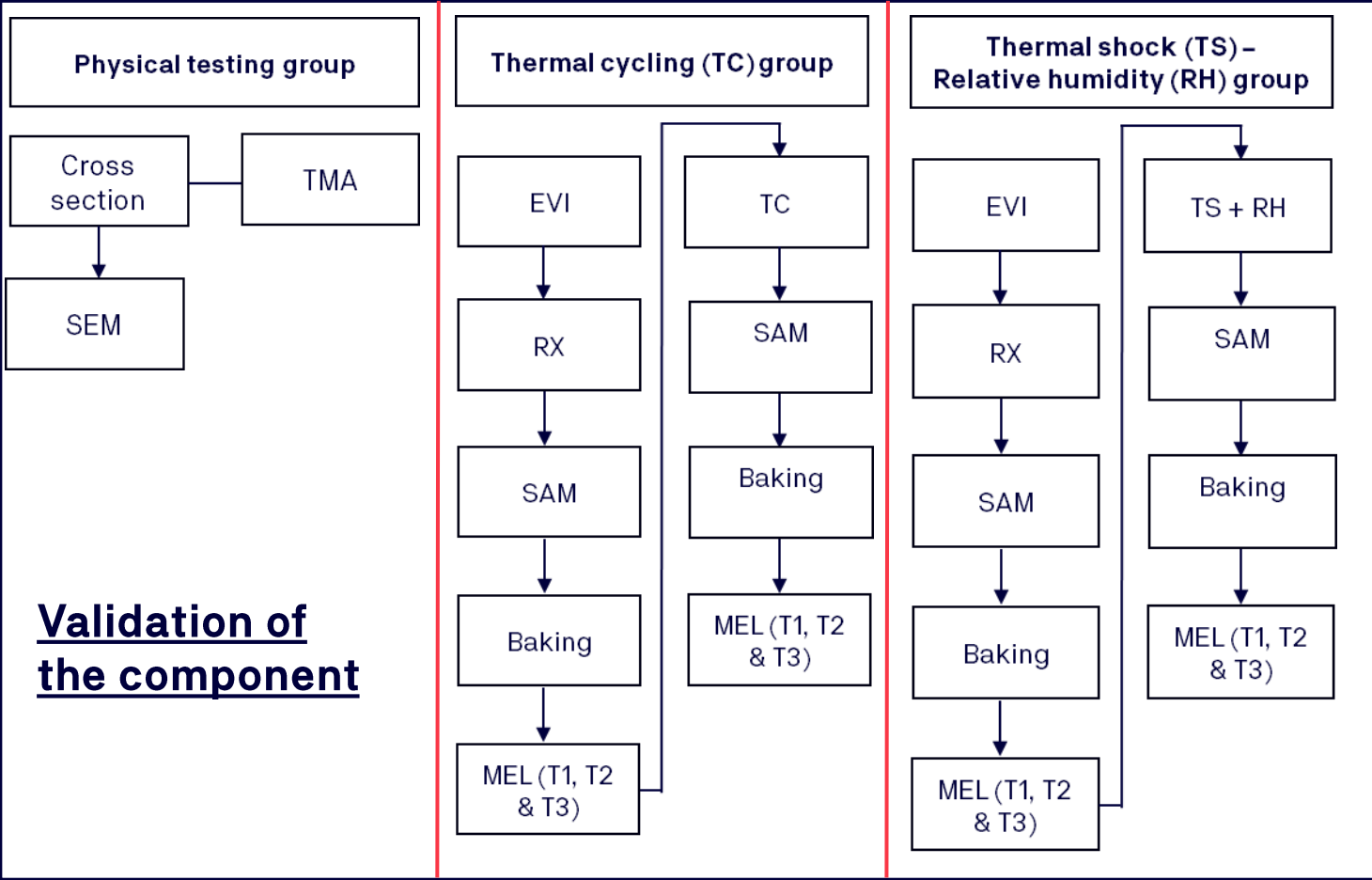
8. Test vehicles definition: Part types selection (WP6)



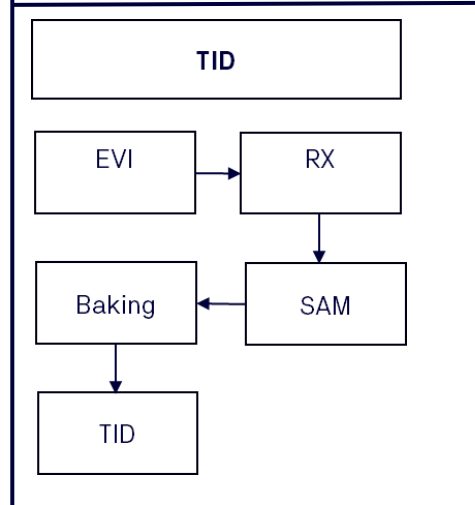
9. Test vehicles characterization (WP6)

Test plan: 2N2222A & 2N2907A

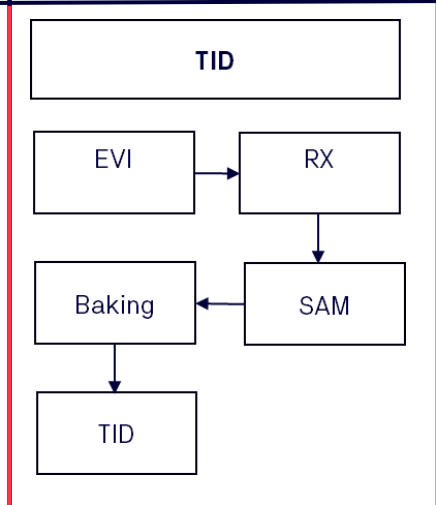
Encapsulated transistors: Resin + 10% Fillers



Encapsulated transistors: Resin + 10% Fillers

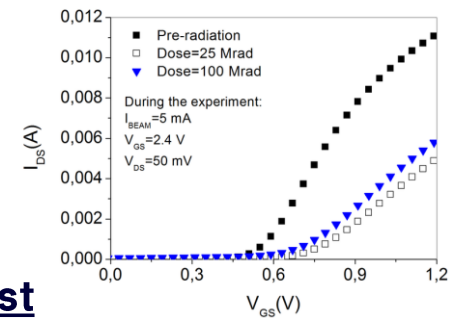


Encapsulated transistors: Resin (Control)



Radiation shielding performance through electrical response:

Composite vs Control test vehicles



A. Marcilei, M. Nilberto, R. Valéria and N. Added. Brazilian facilities to study radiation effects in electronic devices.



10. Potting compound approach (WP6)



Advantages

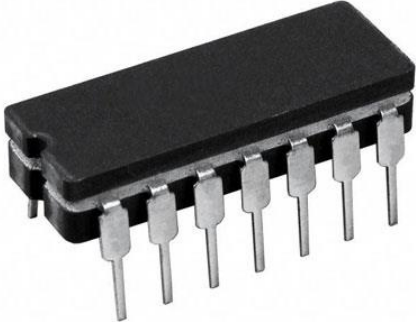
Customized protection: For the whole assembly or selective

Thickness can be increased, improving radiation shielding properties

Part type selection

- ✓ Encapsulated: Real assembly scenario
- ✓ High TID sensitivity
- ✓ Available TID data (ALTER data base)
 - ✓ MOQ & LT
- ✓ Simple electrical test set up

OP470AY/883C: Operational amplifier

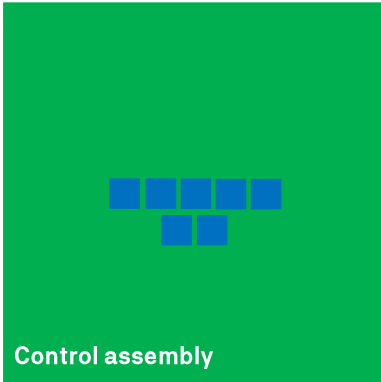


Potting compound approach testing

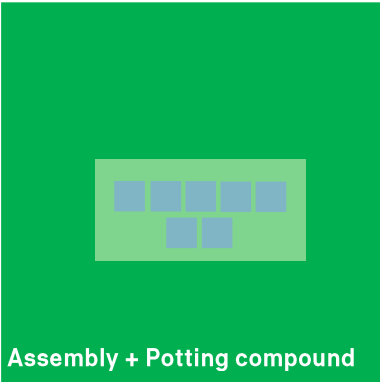
Control assembly:
TID

Assembly + Potting compound:
EVI + SAM + Baking + TID +
Cross section

Detection of radiation shielding improvement through TID response



Control assembly

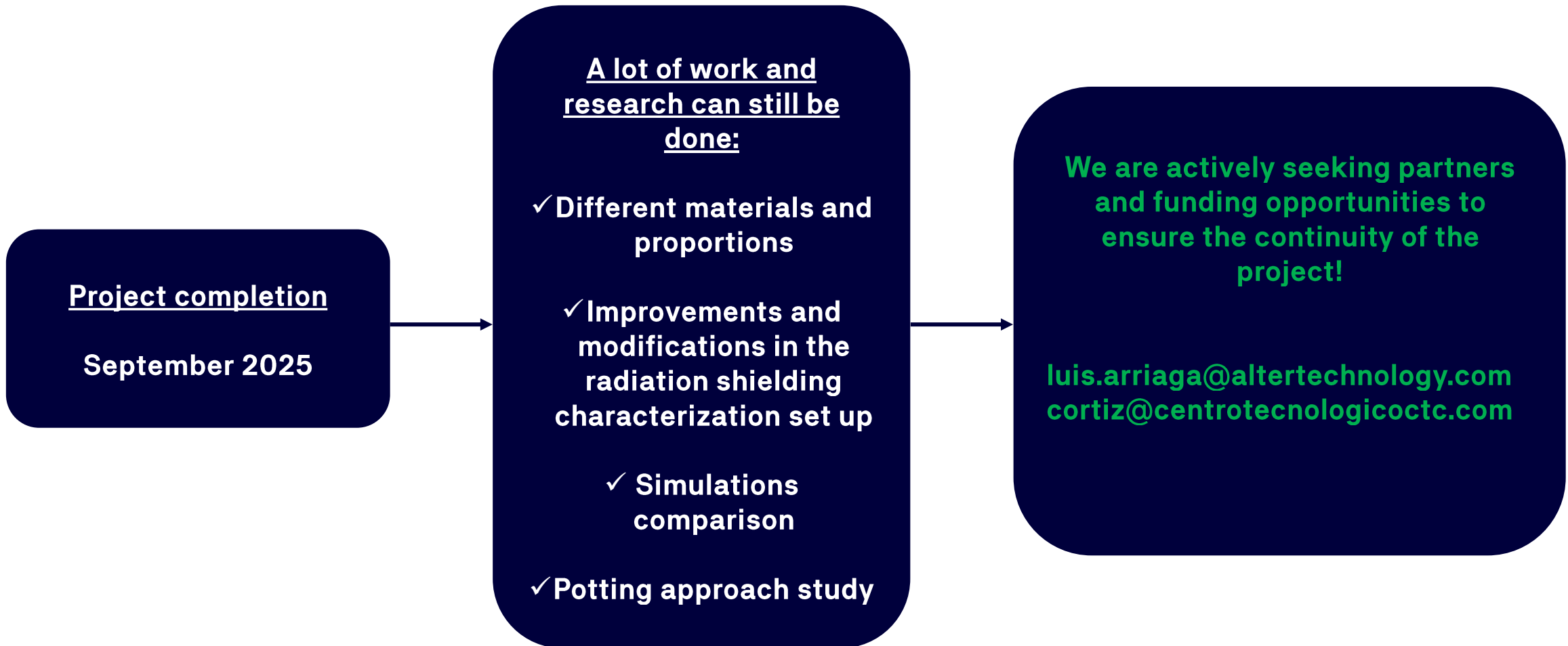


Assembly + Potting compound

11. Project status & Future developments

- ✓ **Materials, fillers, proportions and manufacturing processes: Defined**
- ✓ **Composites manufactured and final formulation selected.**
- ✓ **Part types for test vehicles manufacturing: Procured.**
- ✓ **Adaptable methodology for material selection and characterization successfully developed and implemented: Valid for other material combinations.**
- ✓ **Adaptable gamma radiation shielding characterization setup successfully implemented: Valid for other materials.**
- ✓ **Extrapolation to other manufacturing processes at assembly level (potting): On going**
- ✓ **Radiation shielding simulations will be perform and will be compared with experimental results. Simulations can be extended to other combinations of materials to widen the scope of the study in the future.**
- ✓ **Expertise acquired during this project could lead to light weight solutions that allow to solve other radiation related issues in space like crew protection, structural protection, etc.**

11. Project status and future developments



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Thanks for your attention

Questions and collaborations:

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