

Space Qualification Approach for MOEMS Technologies

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- I. Context and challenges
- II. M(O)EMS Δ Qualification study
- III. Results and lessons learnt
- IV. M(O)EMS new standard for ECSS or ESCC ?
- V. Conclusion

Micro (Opto) Electro Mechanical structures, as identified in the 5 years MNT Strategic dossier published by CTB WG, are MEMS with the ability to alter or modulate a light beam by reflection, diffraction or refraction.

They combine optical, mechanical and electrical functions.

MOEMS is an enabling technology for some applications, such as optical communication and projection display.

MOEMS devices are promising for the implementation of on-board optical communication systems with high bandwidth (several GBit/s), thanks to their low consumption, low volume and low mass. They can be used as:

- **Optical switches** for redundancy switching or low-port-count switching
- **Switching matrices** of optical cross connects for medium and high-port-count signal routing
- **Variable Optical Attenuators (VOA)** to balance power levels among channels
- **Tunable optical filters** (such as Fabry-Perot) to select different bands of wavelength for WDM
- **Tilting micro-mirrors** for the fine optical beam steering and tracking to support free space optical links.

Optical payload applications as:

- Telecommunications,
- On-board data-handling,
- Data communications

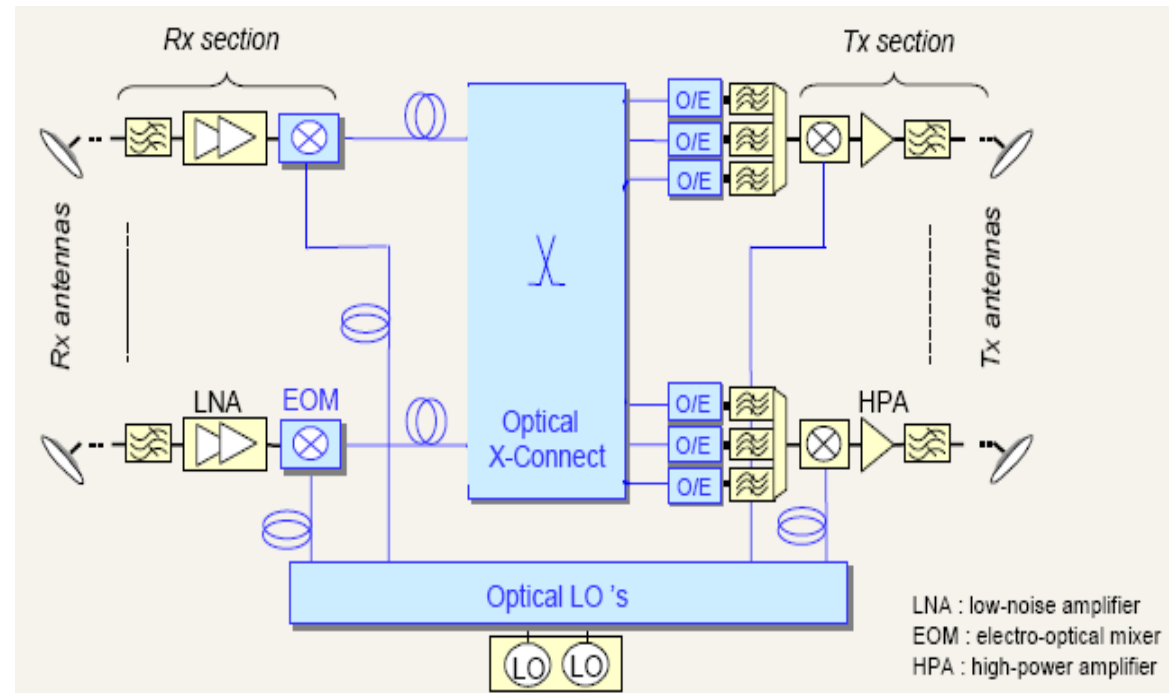
might be incorporating MOEMS devices.

Generic architectural concept

- optical distribution of centralised LO's
- optical frequency down-conversion
- optical cross-connection of μ -wave channels

Merits

- flexible beam-to-beam connectivity
- broadband, frequency-independent design
- scalable to large sizes
- low mass & volume, power consumption

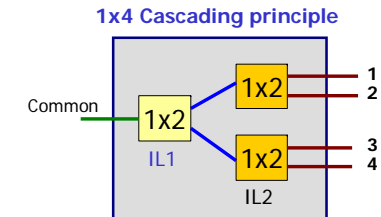
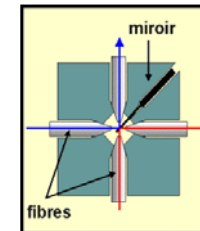


Application example: a schematic of *opto-microwave transparent repeater* (after SAT 'N LIGHT project)

Using MOEMS switch architectures available on the market 1/2

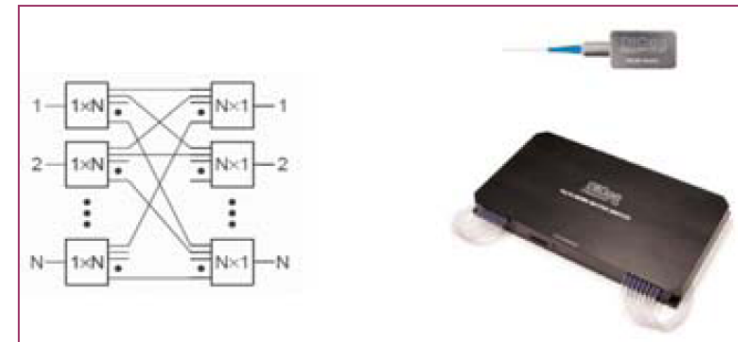
□ Elementary 1x2 and 2x2 switches

- ✓ SERCALO, DICON, NORTHROP GRUMMAN, ...



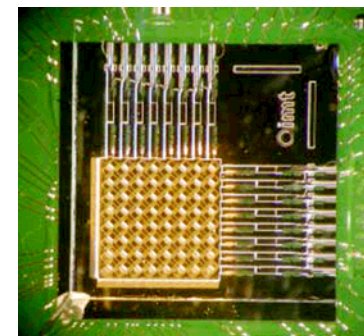
□ Hybrid assemblies of discrete 1x2 and 1xN switches

- ✓ Any connectivity,
- ✓ Matrices as Spanke network
- ✓ SERCALO, DICON (up to 24x24)



□ Integrated 2D crossbar switches

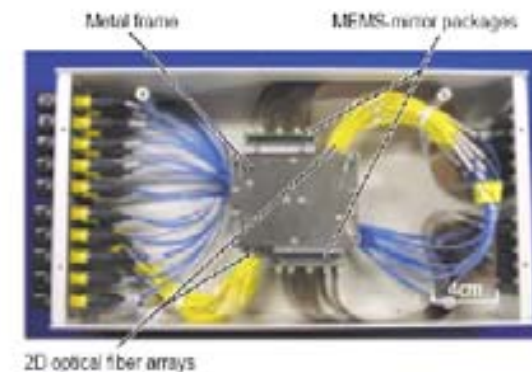
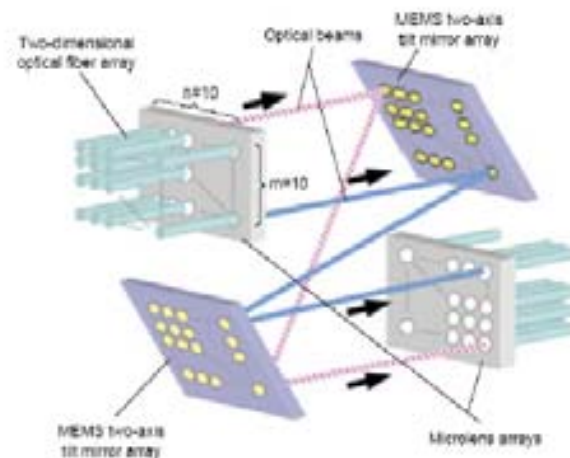
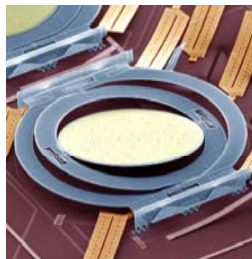
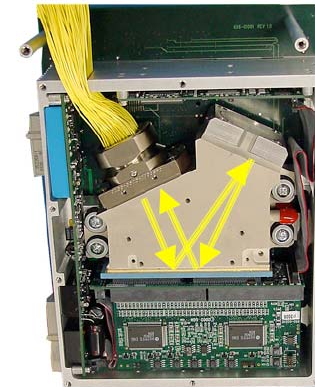
- ✓ Planar matrices with digital control,
- ✓ Complexity grows as N^2
- ✓ Single-sided or double sided
- ✓ AT&T, OMM, SERCALO



Using MOEMS switch architectures available on the market 2/2

□ Analog, free-space MOEMS switches

- ✓ linear mirror arrays, with analog 2D steering, ...
- ✓ mirror matrices with analog 3D steering
- ✓ **complexity grows as 2.N**
- ✓ **Folded, unfolded architectures**
- ✓ **LUCENT, Glimmerglass, NTT, FUJITSU**



- Early demonstrators of MOEMS based microwave photonic sub-systems have confirmed feasibility and proved excellent RF performance.
- Preparing the future, is now to consolidate (and industrialize) these technologies able to be used under space constraints.

Δ Space Qualification objective study : to define and evaluate qualifications tests sequences defined as close as possible to these constraints (and above), in order to assess MOEMS product limits when placed under these application domains.



(“ Δ ” wording means based on existing TELCORDIA and MIL-STD qualification normative documents reconsidered.)

Phase 1

- Survey of Micro Optical Electro Mechanical Systems (MOEMS).
- MOEMS Suitability for Space Applications and Technology Maturity Level.
- Space Qualification Definition, Effort for Δ Qualification, MOEMS selection

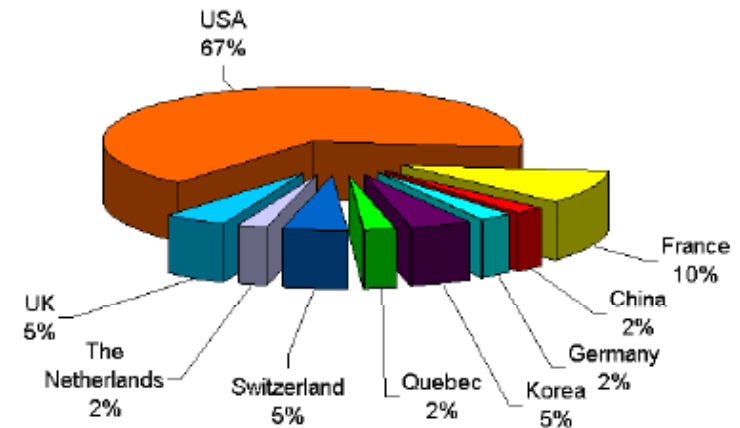
Phase 2

- Consolidation of Δ Space Qualification Definition, Procurement of selected MOEMS.
- Δ Space Qualification Testing and Reporting.

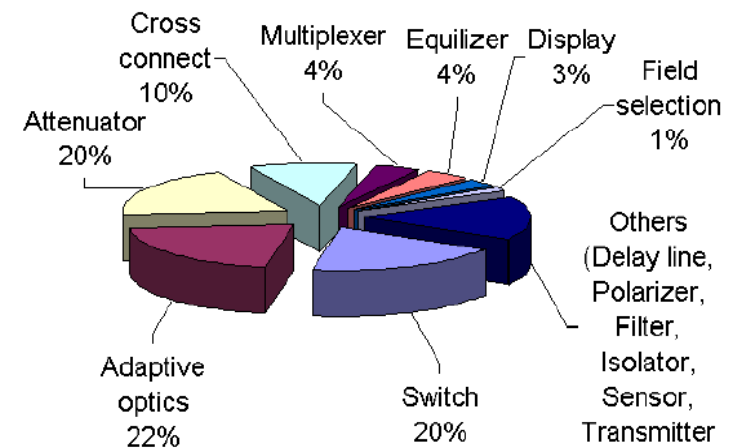
Data base construction

- Up to 100 companies contacted
- Identify companies which are real market players, which have their own MOEMS products
- Companies proposing commercial MOEMS products
- database completed
 - more than 100 datasheets downloaded
 - more than 150 papers
- Data base completed at the end of 2006 (phase 1).

MOEMS Market players (2005)



Family panorama



A full comparison between existing standards (Test methods from ESA, MIL and TELCORDIA): ECSS-Q-60-05 for hybrids, ESCC9010 for ICs, ESCC3702 for switches, ESCC3602 for relays, Telcordia GR-1209 Generic requirements for passive optical components, Telcordia GR-1221 Reliability assurance for passive optical components, Telcordia GR-1073 Generic requirements for single mode fiber optic switches.

A strong effort and fruitful exchange of background between all partners help to identify for and against test methods applicable to MOEMS.

MIL-STD-883 & 750: Advantages, drawbacks, implementation ability, effectiveness, availability, have been ranked for each of them:

MECHANICAL TESTS			
Insulation resistance MIL-STD-883, method 1003	NR		+ Required for high control voltage + ESD test seems more relevant for MEMS/MOEM
Moisture resistance MIL-STD-883, method 1004.7	M		Main failure cause
PERFORMANCE CHARACTERIZATION TESTS			
Thermal, electrical and optical characteristics MIL-STD-883, method 1012.1 and other for electrical and optical	M		Relevant for space application.
ENVIRONMENTAL OPERATING TESTS			
Damp heat MIL-STD-202. Method 103 or IEC 68-2-3. 85_/85%RH /1000hrs	M		Results used for FIT rate estimates
Burn-in test MIL-STD-883, method 1015.9	M		Burn-in test is specific for IC circuits and packages. Should be optimized and adapted for MEMS/MOEMS
OTHER ENVIRONMENTAL TESTS			
Constant acceleration MIL-STD-883, method 2001.2	NR		Not necessary, shocks are more representative
Mechanical shocks MIL-STD-883, method 2002.4	M		ESA standard compared to Telcordia: shorter and more powerful
Solderability MIL-STD-883, method 2003.8	M		The materials that will replace tin lead should be qualified for space.

INITIALLY 25 DEVICES + 4 Bare dice			
SG 1 / Thermo-integrity 4 devices + CD	SG 2 / Thermo-Electrical 8 devices + CD	SG 3 / Mechanical & endurance 6 devices + CD	SG 4 / Miscellaneous 6 devices + CD
Temperature cycling (m1010.8) cond. A (-55°C, + 85°C) 50Cy. cond. A (-55°C, +100°C) 50Cy. →100 cy. pass/fail cond. B (-55°C, +125°C) #Cy. Up to 50% lot fail	Electrical / Optical characteristics at room, high and low temperature (m1012.1)	Vibration Fatigue (m2007.3)	Radiation tests (m1019.6)
IMEC Alternative test for Sealing integrity	Moisture resistance 5 cy. -40°C to +75°C, RH 90% only controlled at 75°C	Mechanical shocks (m2002.4)	RGA (m1018.4) 2 devices
	Thermal Shocks 5 cy., -55, +125°C (m 1011.9)	Endurance test at high Temp & Under Vacuum (m1033)	Out gassing test 4 devices
	3 Devices		CONSTRUCTIONAL ANALYSIS 2 devices + 4 Bare dice including Lead Integrity & Bending test
	ESD testing (m3015.7)		Rapid depressurization (m1001)
	5 Devices		
	Moisture resistance 10 cy. -40°C to +75°C, RH 90% only controlled at 75°C (m1004.7)		
	IMEC Alternative test for Sealing integrity		
	Thermal Shocks 5 cy., -55, +125°C (m 1011.9)		

Do While:
lot failure rate <=40%
or 5 loops

The test exercise was conducted on **2 MOEMS manufacturers (Optical switch and VOA)**: 25 samples (+10 bare dice) from each manufacturer.

Each lot was characterized, screened and tested according to the Δ Qualification flow including :

Electro-Mechanical,
Environmental,
and Miscellaneous test groups (Radiation, CA, Out gassing...).

THERMO-INTEGRITY TESTS GROUP

The goal is to assess the internal package assembly robustness and temperature cycling.

As the products are new and have internal mechanical parts it was decided to start with 3 step of stress from a low temperature range to an increased range up to the final usually implemented for space (i.e. 100 cy at -55°C to $+125^{\circ}\text{C}$).

THERMO-ELECTRICAL TESTS GROUP

The goal is to characterize the devices in the temperature range compatible to space application in order to confirm the maximum rating temperature values with the corresponding set of opto-electrical parameters.

The second part of the Thermo-Electrical sub-group is to assess the behavior of the devices when submitted to external humidity and temperature stress step sequences: this series of test with a loop has been defined to progressively identify the limits of the products in order to derive some recommendations for the final methodology.

ENVIRONMENTAL TESTS GROUP

The goal is related to mechanical stress applied to the products either to stress the package integrity or to stress the MOEMS mechanical parts included into the dice.

Vibration fatigue (supposed to be the most representative equivalent stress seen during the launch of the satellite) and the mechanical shocks are defined in series with standard space level of strains.

At the end an endurance test at high temperature and under vacuum is proposed in order to simulate the space environment and to verify if vacuum (including rapid depressurization stress) may have an impact on the reliability of the products, especially when it is not clear if the MOEMS package can be considered as hermetic.

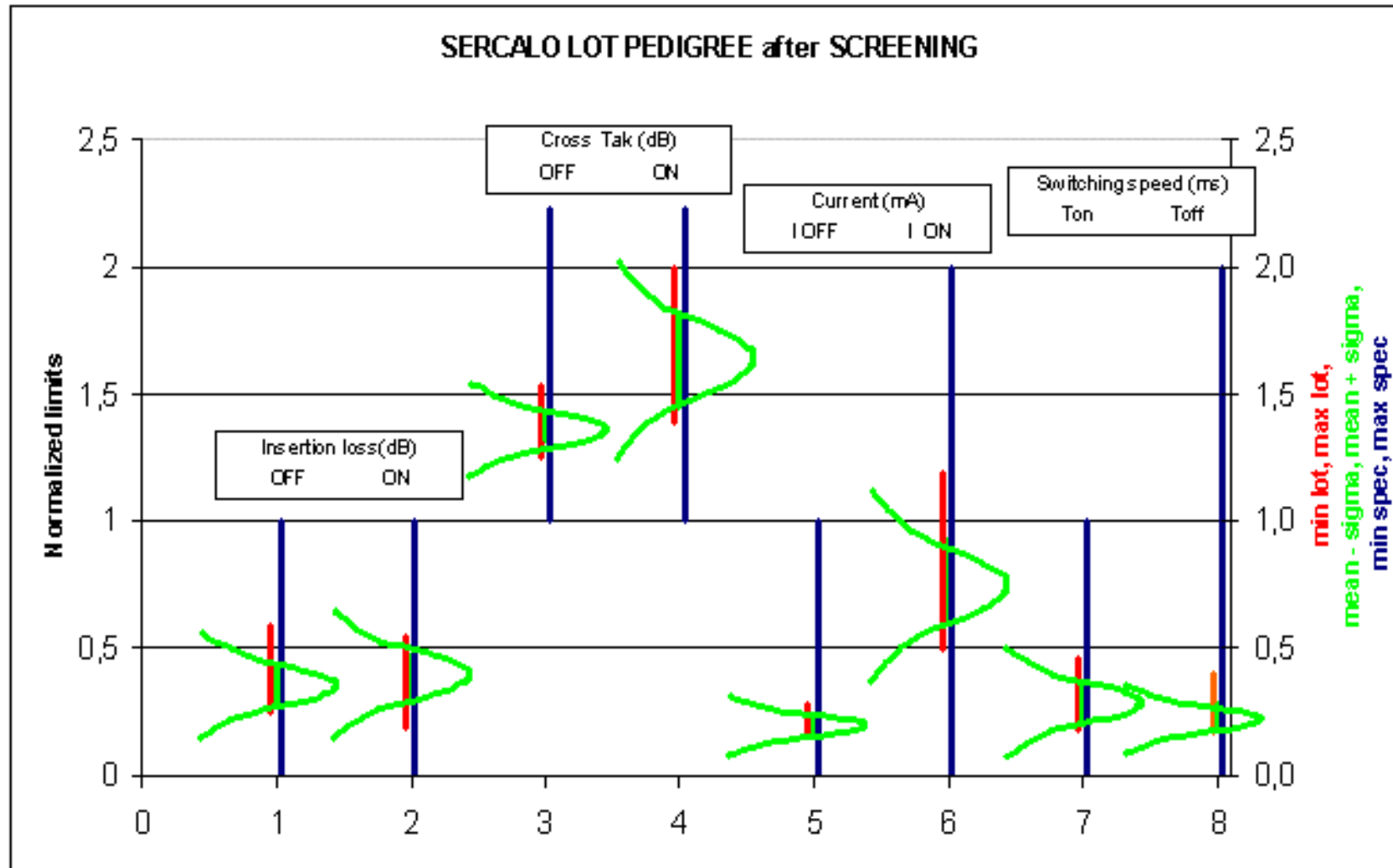
MISCELLANEOUS TESTS GROUP

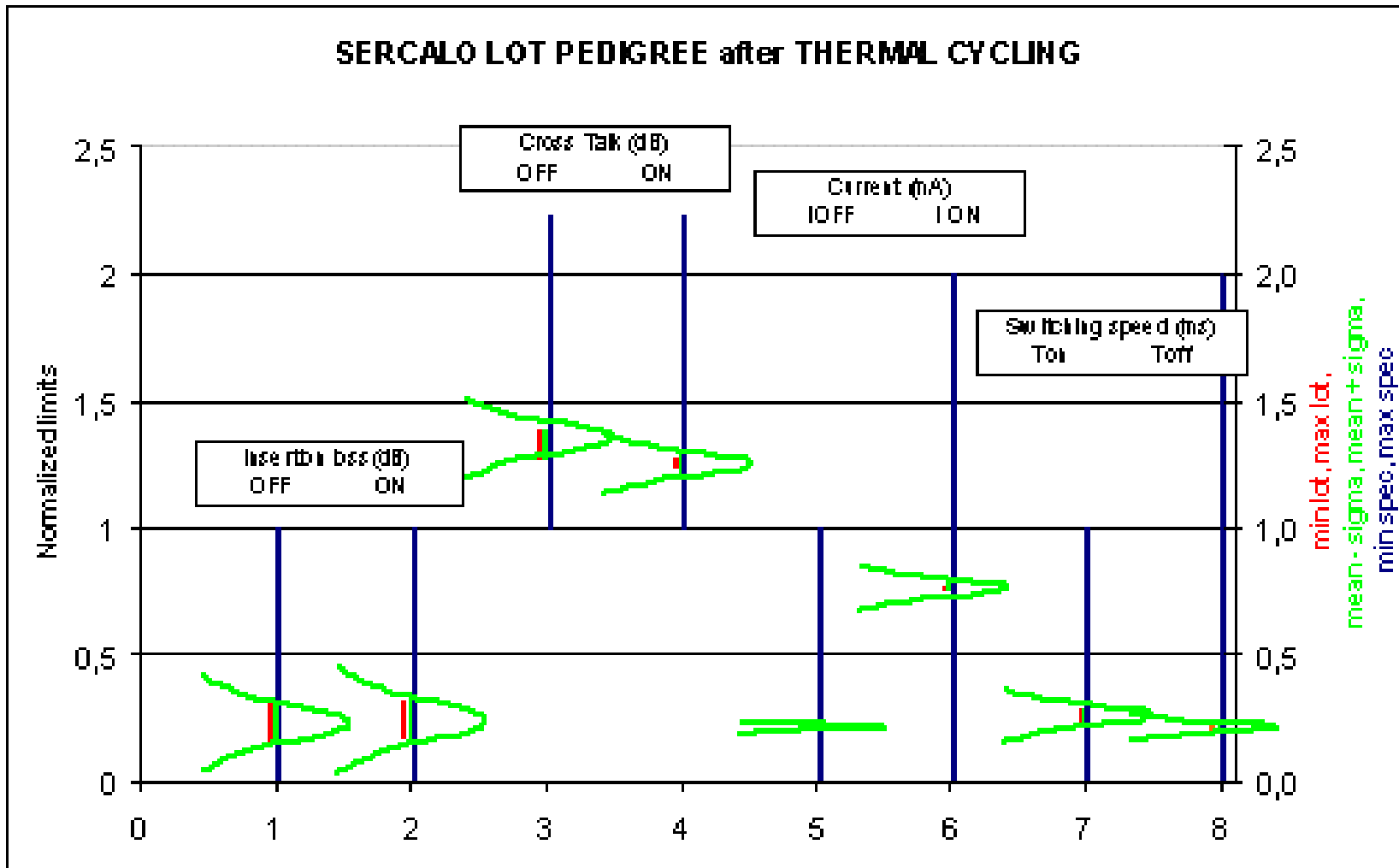
The goal is to address product manufacturing and industrialization suitable for space application and environment.

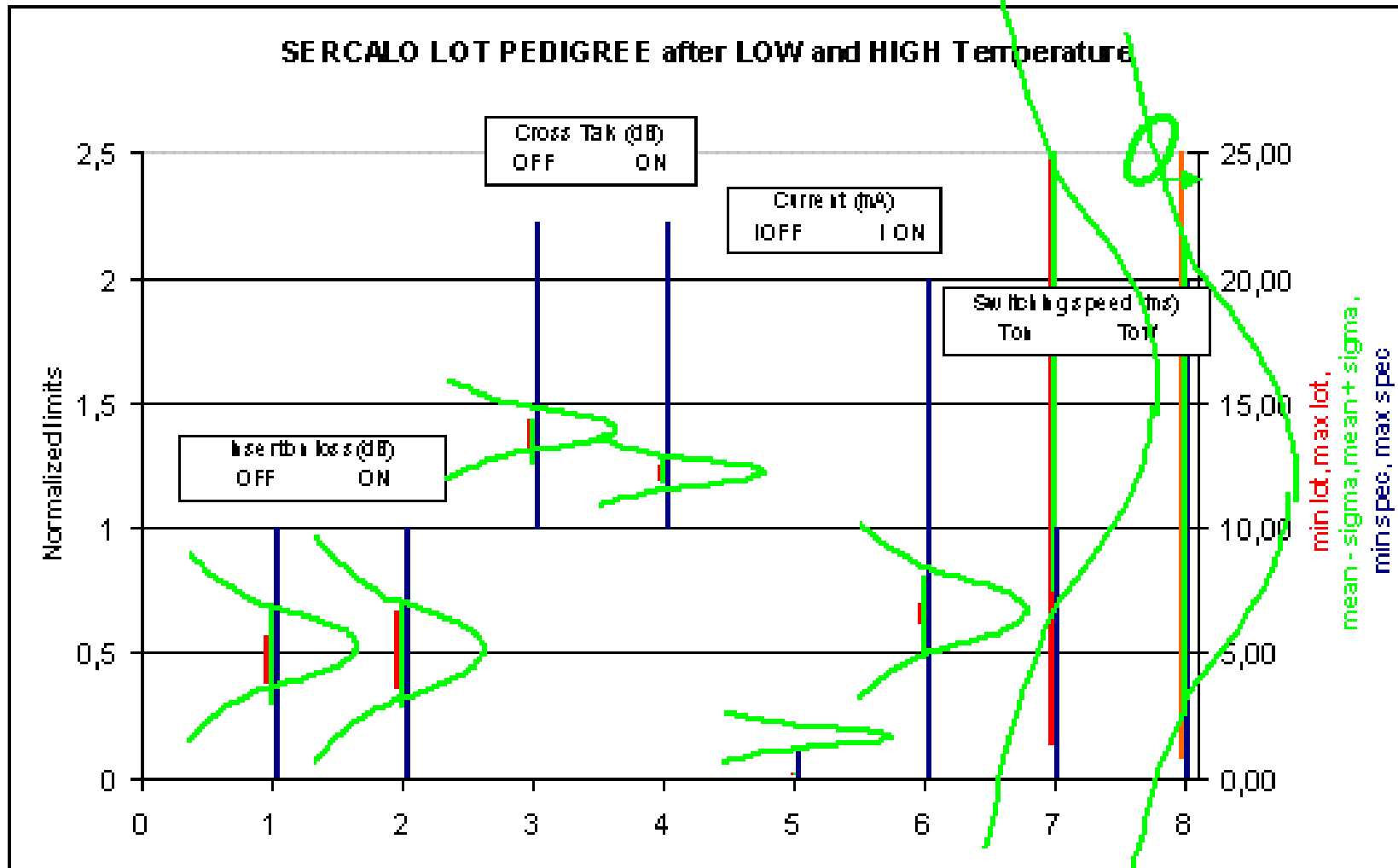
Sequence details radiation testing together with constructional analyses and material behavior under vacuum including RGA and out gassing tests (for contamination risk, if any, on optical structures) or rapid depressurization to demonstrate that the package integrity may be safe during rocket launch.

Thermo-Mechanical Test Group		Environmental Test Group	Miscellaneous Test Group	
SG 1: Thermo-integrity 4 devices	SG 2: Performance Characterization 8 devices	SG 3: Mechanical & endurance 6 devices	SG 4: Miscellaneous 6 devices	
+ 1 Control Device for all Test Groups				
Temperature cycling (m1010.8) ESA	Electrical / Optical characteristics at room, high and low temperature (m1012.1) SERCALO	Vibration Fatigue(m2005.2) ESA under SERCALO responsibility	Radiation tests (m1019.6) ESA	
DAMP HEAT SERCALO	Moisture resistance (m1004.7) SERCALO	Mechanical shocks (m2002.4) ESA under SERCALO responsibility	2 Devices ***NOT applicable*** RGA (m1018.4)	4 Devices Out gassing test ESA
<u>SERCALO</u>	Thermal Shocks (m 1010.8 with dual chamber) SERCALO	Endurance test at high Temp & Under Vacuum (m1033) Already done by SERCALO	CONSTRUCTIONAL ANALYSIS ESA	Rapid depressurization (m1001) IMEC
	3 Devices ESD testing(m3015.7) IMEC	5 Devices Moisture resistance (m1004.7) SERCALO		
		DAMP HEAT SERCALO		

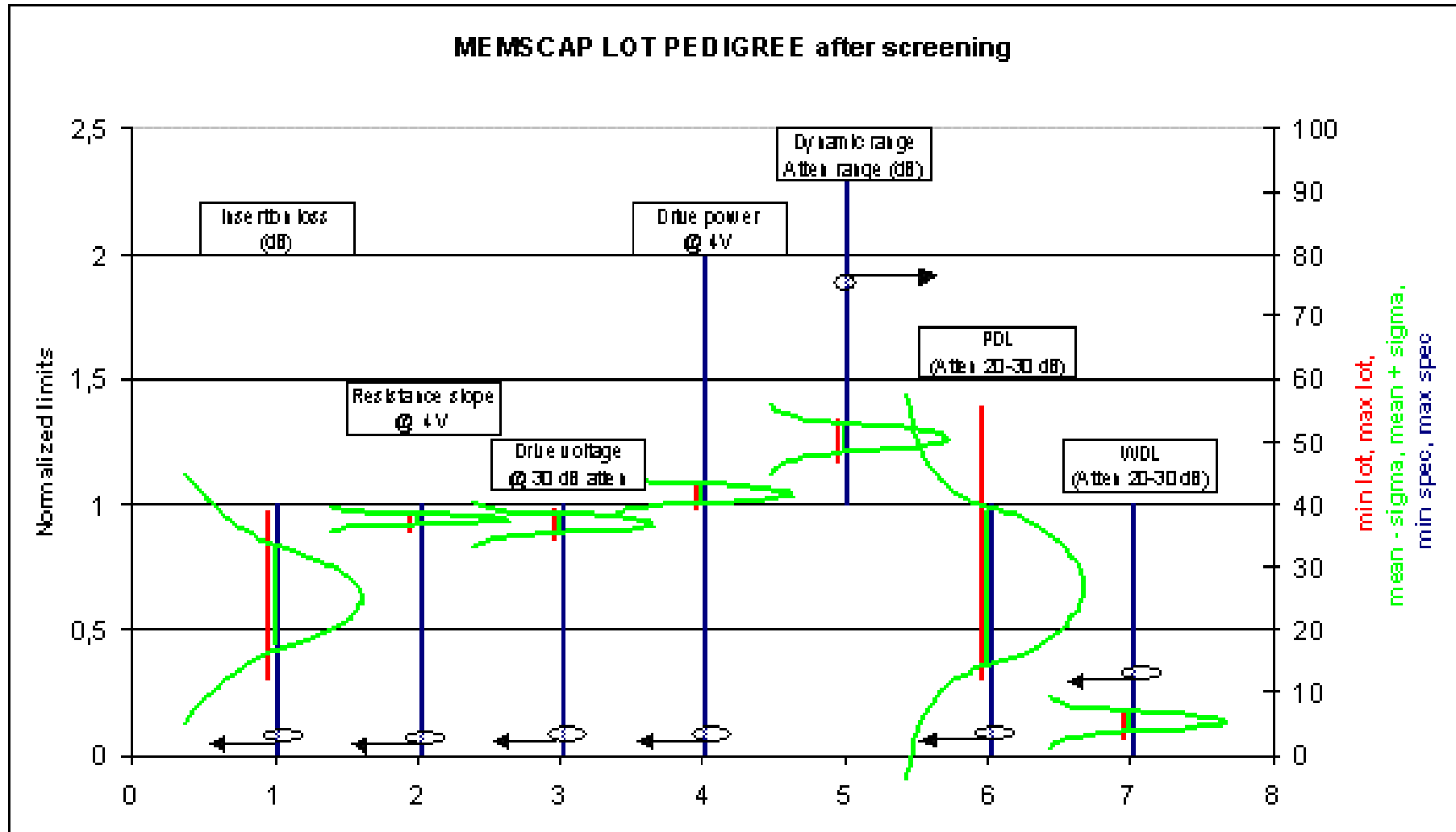
Do While:
lot failure rate
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Thermo-Mechanical Test Group		Environmental Test Group		Miscellaneous Test Group		
SG 1: Thermo-integrity 4 devices		SG2: Performance Characterization 8 devices		SG 3: Mechanical & endurance 6 devices		
+ 1 Control Device for all Test Groups						
Temperature cycling (m1010.8) MEMSCAP	Electrical / Optical characteristics at room, high and low temperature (m1012.1) IMEC		Vibration Fatigue (m2005.2) IMEC		Radiation tests (m1019.6) ESA	
	2 Devices		4 Devices		2 Devices	
	ESD testing(m3015.7) IMEC		Moisture resistance (m1004.7) MEMSCAP		RGA (m1018.4) ORS	
MEMSCAP		Endurance test at high Temp & Under Vacuum (m1033) THALES		CONSTRUCTIONAL ANALYSIS ESA		
				4 Devices Out gassing test ESA		
				Rapid depressurization(m1001) IMEC		



➤ Screening data

- Screening process didn't reveal infant mortality behaviour.
- Insertion loss, PDL and WDL are parameters complicated to be characterized (reproducibility, large dispersion, precision not guaranteed). Measurement techniques to be demonstrated or even improved for reliability assessment.
- Main failures observed at fiber input and output connectors due to handling and interim characterization.

➤ Thermo-electrical group

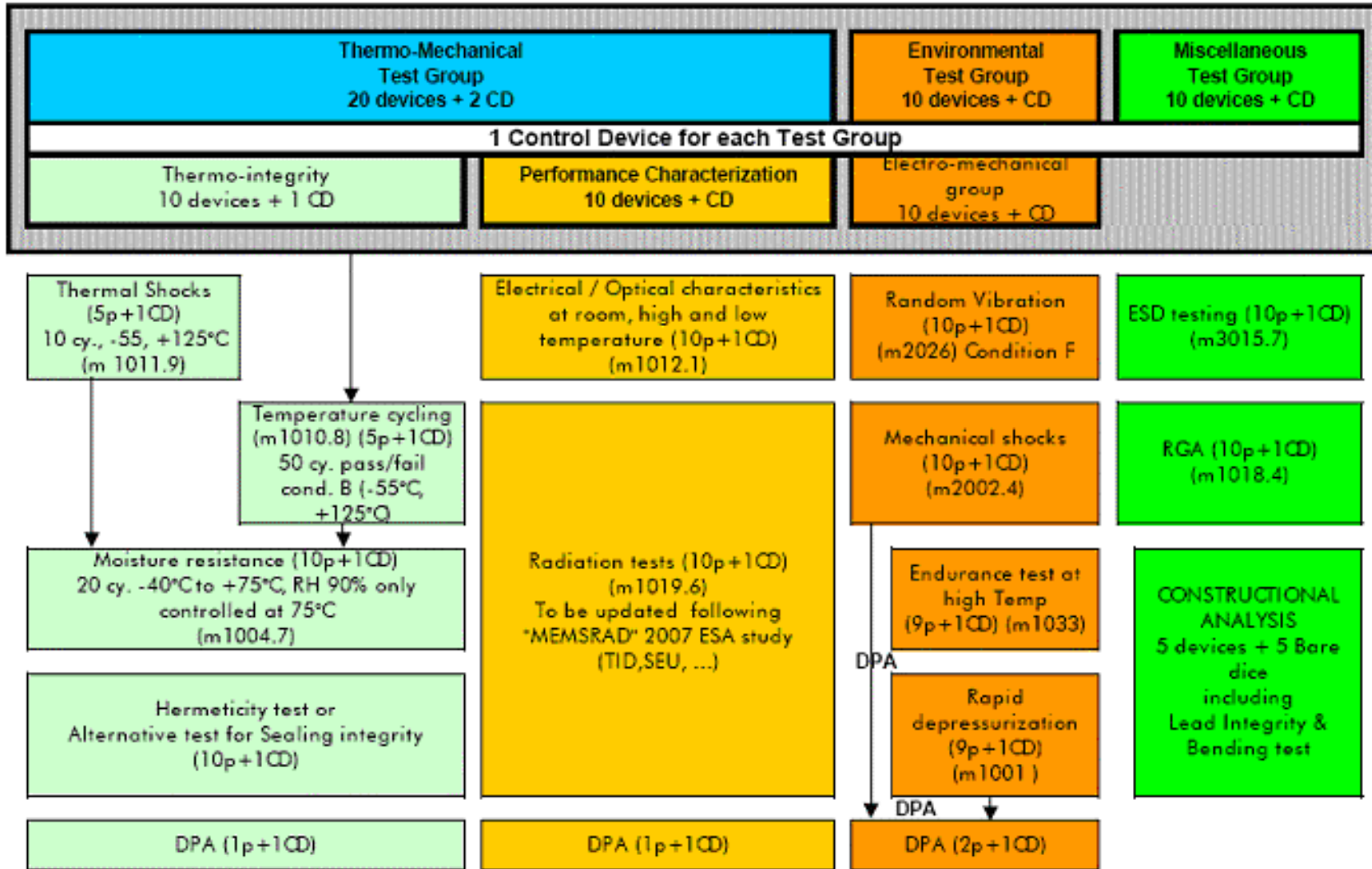
- Thermal cycling and damp heat are conclusive (but we observed dispersions on insertion loss parameter)
- Temperature cycles conclusive (no failure)
- ESD stress test with no failure identified.
- Moisture Resistance and Thermal Shocks are successful (no failure observed)

➤ Environmental group

- Only failures at fibers level: handling or mechanical failure on PCB. Need to secure connections handling during the test.
- Vibration and mechanical shocks are successful (no failure)
- Endurance high temp and vacuum : conclusive but vacuum is not necessary implementing as no change in optical characteristics.

➤ Miscellaneous group

- Radiation tests: Failures observed on IC COST electronic command. MOEMS not affected, Switch recovers and still working after electronic part replacement.



- MOEMS devices are TML3 to TML4 maturity for space application.
- TELCORDIA standard is not enough to cover space requirements and Delta Qualification program added or more stringent conditions:
 - Radiations tests,
 - ESD, RGA, Out gassing and Rapid depressurization,
 - Thermal shocks and Moisture resistance (test in series and loop proposed),
 - Mechanical shocks, Vibration variable frequency and Endurance life test under vacuum,
 - Thermal cycling (-55, +125°C) and alternative test method for hermeticity.

- Test conducted at various facilities including: SERCALO, MEMSCAP, IMEC, Thales Alenia Space.
- Opto-electrical tests at nominal room temperature are critical and need to be carefully implemented: some opto-electrical parameters are sensitive to fiber bent and require attaching fibers to the test equipment set (to reduce mechanical vibration).
- Measurement reproducibility is tough to achieve and needs safe (ESD) and dedicated test room.
- Radiation tests need to be performed with a care test jig design in order to avoid EMC or oscillation.
- ESD test are difficult to implement as some devices are with (and other without) electronic parts inside the module.

- Environmental endurance test under vacuum performed successfully. Not needed to have vacuum. Identification of critical parameters for reliability assessment.
- Endurance under vacuum was also intended to assess the effect of the rapid depressurization test (around $5 \cdot 10^{-5}$ bars after 9 mn at each interim electrical measurement sequence).
- PDL and WDL parameters are not confirmed to be used for quality and reliability verification: they are too much sensitive (measurement dispersion) for environmental test conditions or we need to define the test set up with extreme care.
- Generic qualification flow chart identified and adapted for each manufactured. Global methodology proposed and to be generalized.

**Several benefits and know-how achieved.
Good European collaboration and fruitful contributions.
Space Qualification methodology drawn and tested with important
and major lessons learned.**

Nevertheless, MOEMS in space will require maturity at multiple levels:

- **Products designed, industrialized and manufactured for space harsh environment,**
- **Equipment designed for new enabling architectures demonstrating the benefits of MOEMS instead of standard products (cost and technical benefits must be demonstrated),**
- **User manufacturing and testing in mature environments (learning curve to be initiated and maintained) : role of Agencies to promote these new technologies and help to built a new age of techniques (opto and nanosystems)**

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n°18669/04/NL/CP – “*M(O)EMS Delta Qualification Methodology*”



Thanks to Laurent MARCHAND, Technical Officer at ESTEC, and to
our Partners in this project for the fruitful cooperation.

Annexes

